

# Meshing or not meshing?

Iso/sub/super-geometric analysis  
(adaptive unfitted methods for real-time simulations)

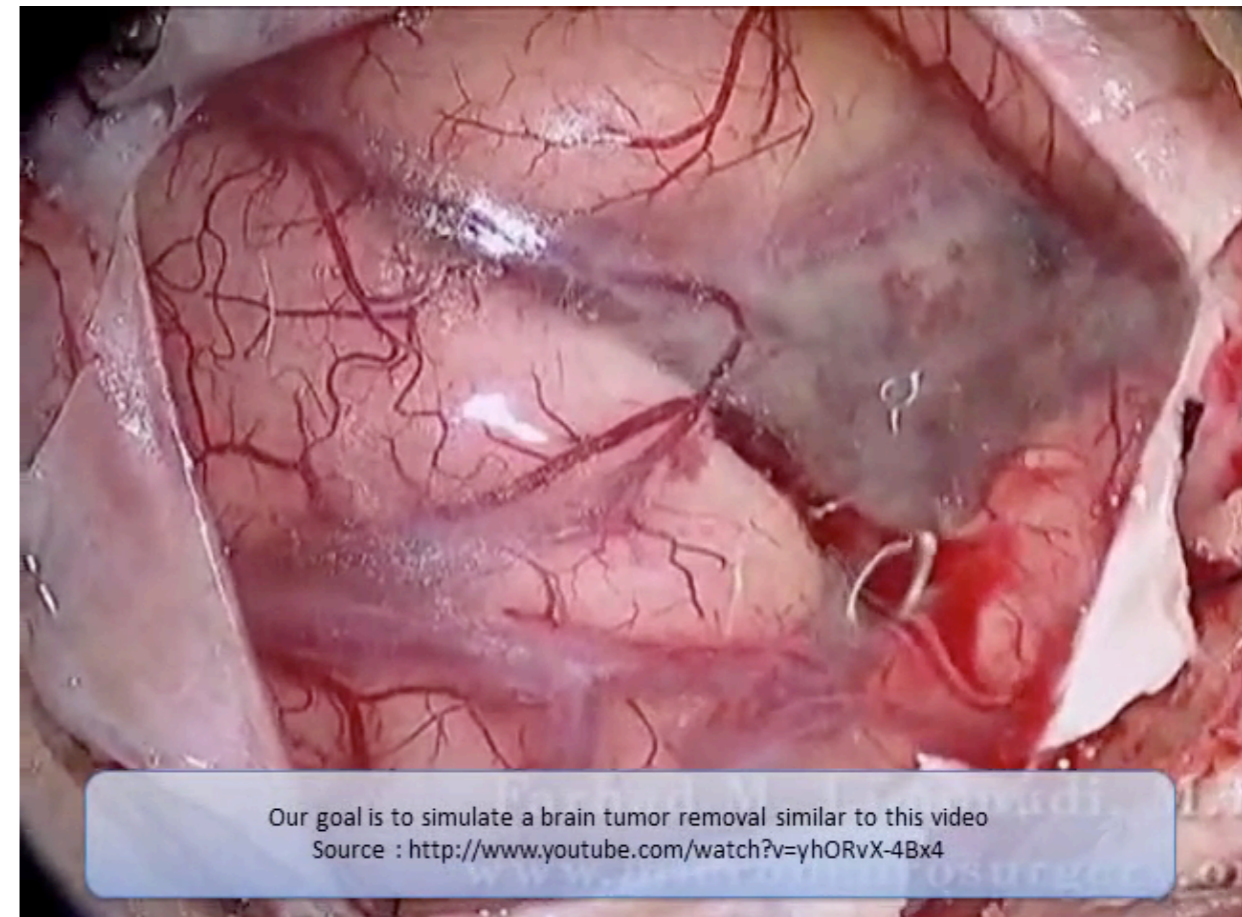
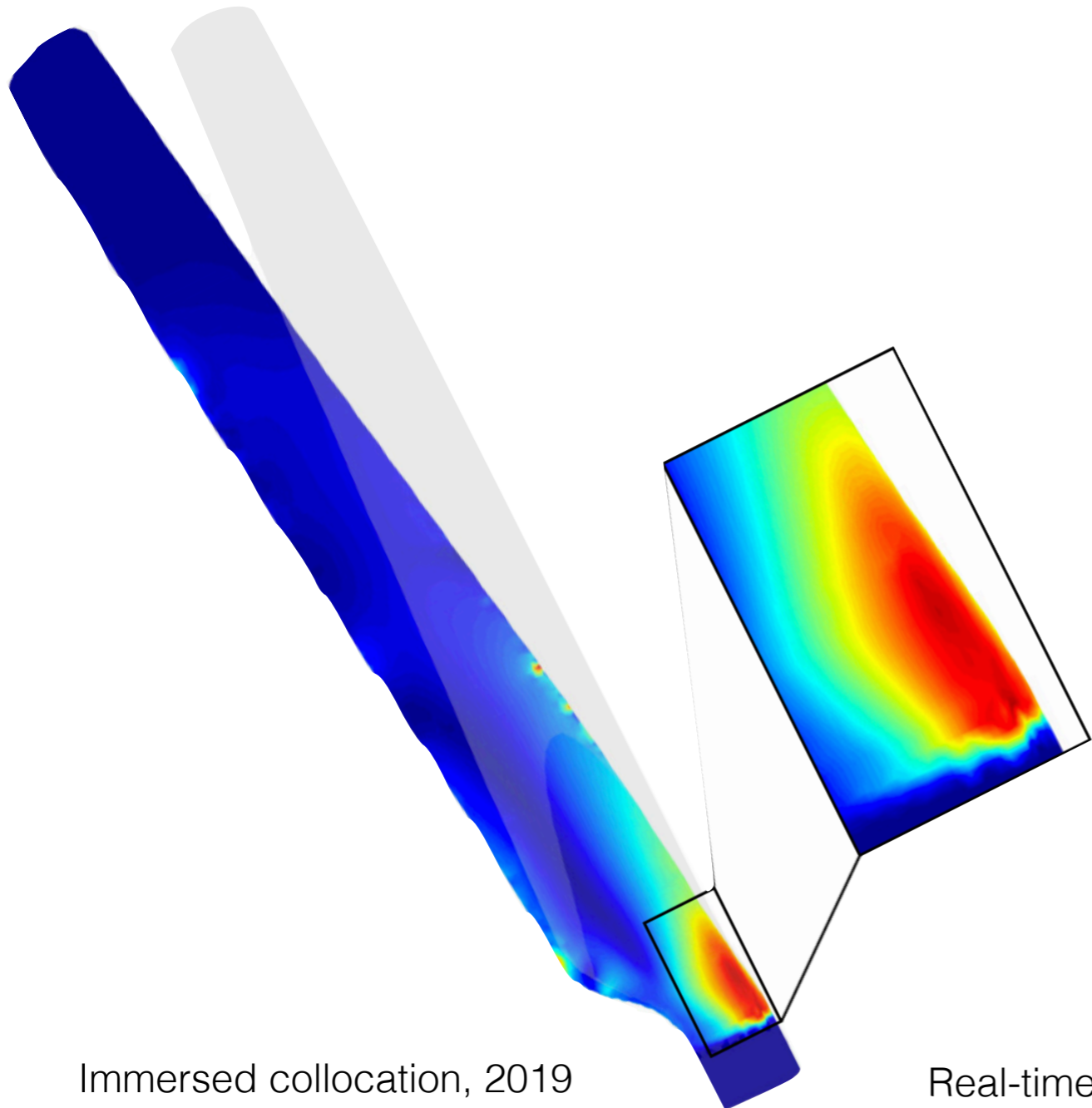
Immersed collocation methods

# Computational Mechanics of Interfaces

with Engineering and Medical Applications

*Stéphane P.A. Bordas*

*stephane.bordas@alum.northwestern.edu*



Immersed collocation, 2019

Real-time cutting, MEDIA2014, IEEE2017, IJNMBE2018, CMAME2018

# One machine, one minute, three billion tetrahedra

Célestin Marot\* | Jeanne Pellerin | Jean-François Remacle

<sup>1</sup>Université catholique de Louvain, iMMC,  
Avenue Georges Lemaitre 4, bte L4.05.02,  
1348 Louvain-la-Neuve, Belgium

## Correspondence

\*Corresponding author: Email:  
celestin.marot@uclouvain.be

## Summary

This paper presents a new scalable parallelization scheme to generate the 3D Delaunay triangulation of a given set of points. Our first contribution is an efficient serial implementation of the incremental Delaunay insertion algorithm. A simple dedicated data structure, an efficient sorting of the points and the optimization of the insertion algorithm have permitted to accelerate reference implementations by a factor three. Our second contribution is a multi-threaded version of the Delaunay kernel that is able to concurrently insert vertices. Moore curve coordinates are used to partition the point set, avoiding heavy synchronization overheads. Conflicts are managed by modifying the partitions with a simple rescaling of the space-filling curve. The performances of our implementation have been measured on three different processors, an Intel core-i7, an Intel Xeon Phi and an AMD EPYC, on which we have been able to compute 3 billion tetrahedra in 53 seconds. This corresponds to a generation rate of over 55 million tetrahedra per second. We finally show how this very efficient parallel Delaunay triangulation can be integrated in a Delaunay refinement mesh generator which takes as input the triangulated surface boundary of the volume to mesh.

# One machine, one minute, three billion tetrahedra

Célestin Marot\* | Jeanne Pellerin | Jean-François Remacle



## Truck tire

# threads	# tetrahedra	Timings (s)		
		BR	Refine	Total
1	123 640 429	75.9	259.7	364.7
2	123 593 913	74.5	166.8	267.1
4	123 625 696	74.2	107.4	203.6
8	123 452 318	74.2	95.5	190.0

# One machine, one minute, three billion tetrahedra

Célestin Marot\* | Jeanne Pellerin | Jean-François Remacle

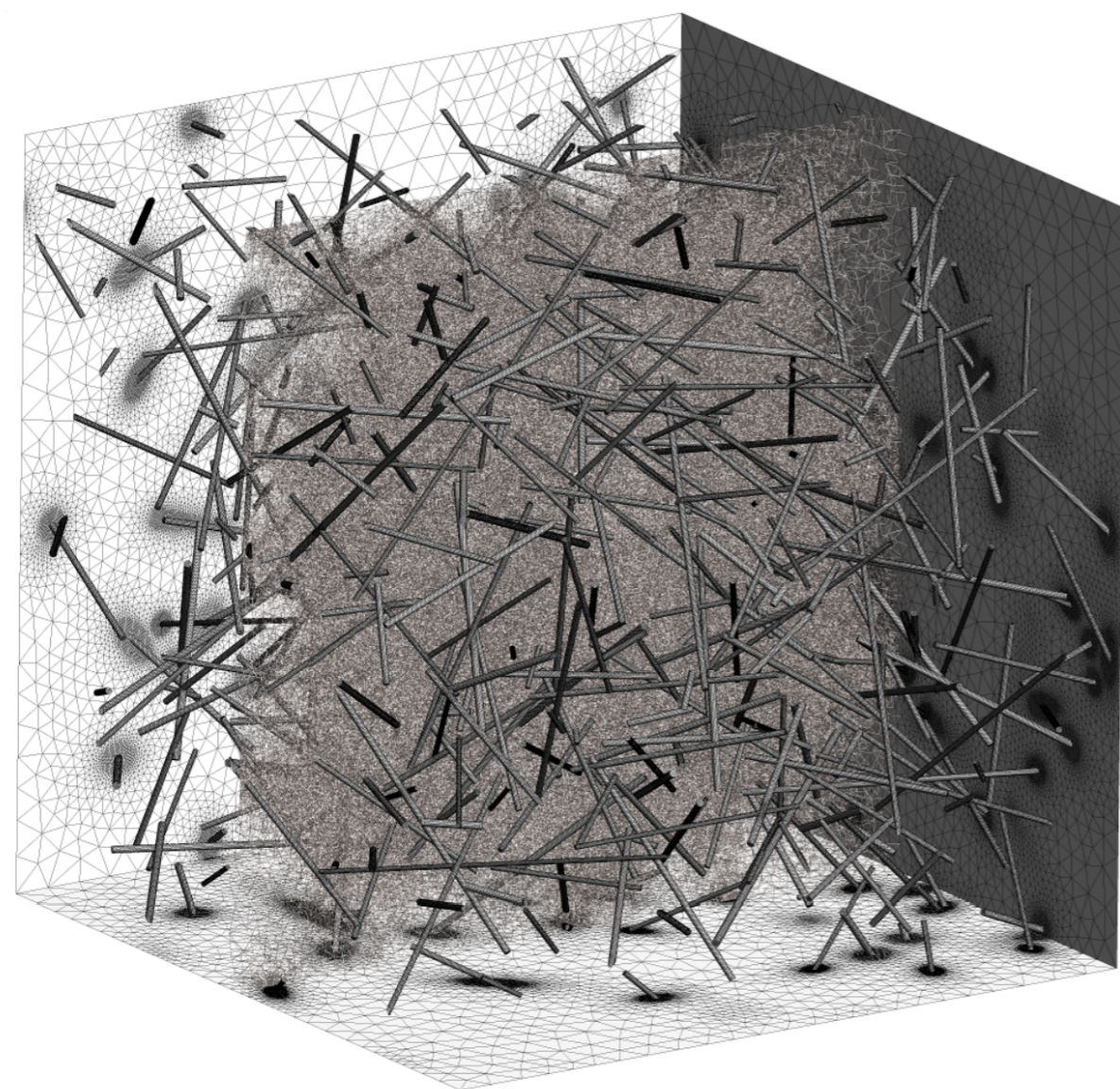


## Truck tire

# threads	# tetrahedra	Timings (s)		
		BR	Refine	Total
1	123 640 429	75.9	259.7	364.7
2	123 593 913	74.5	166.8	267.1
4	123 625 696	74.2	107.4	203.6
8	123 452 318	74.2	95.5	190.0

# One machine, one minute, three billion tetrahedra

Célestin Marot\* | Jeanne Pellerin | Jean-François Remacle



## 100 thin fibers

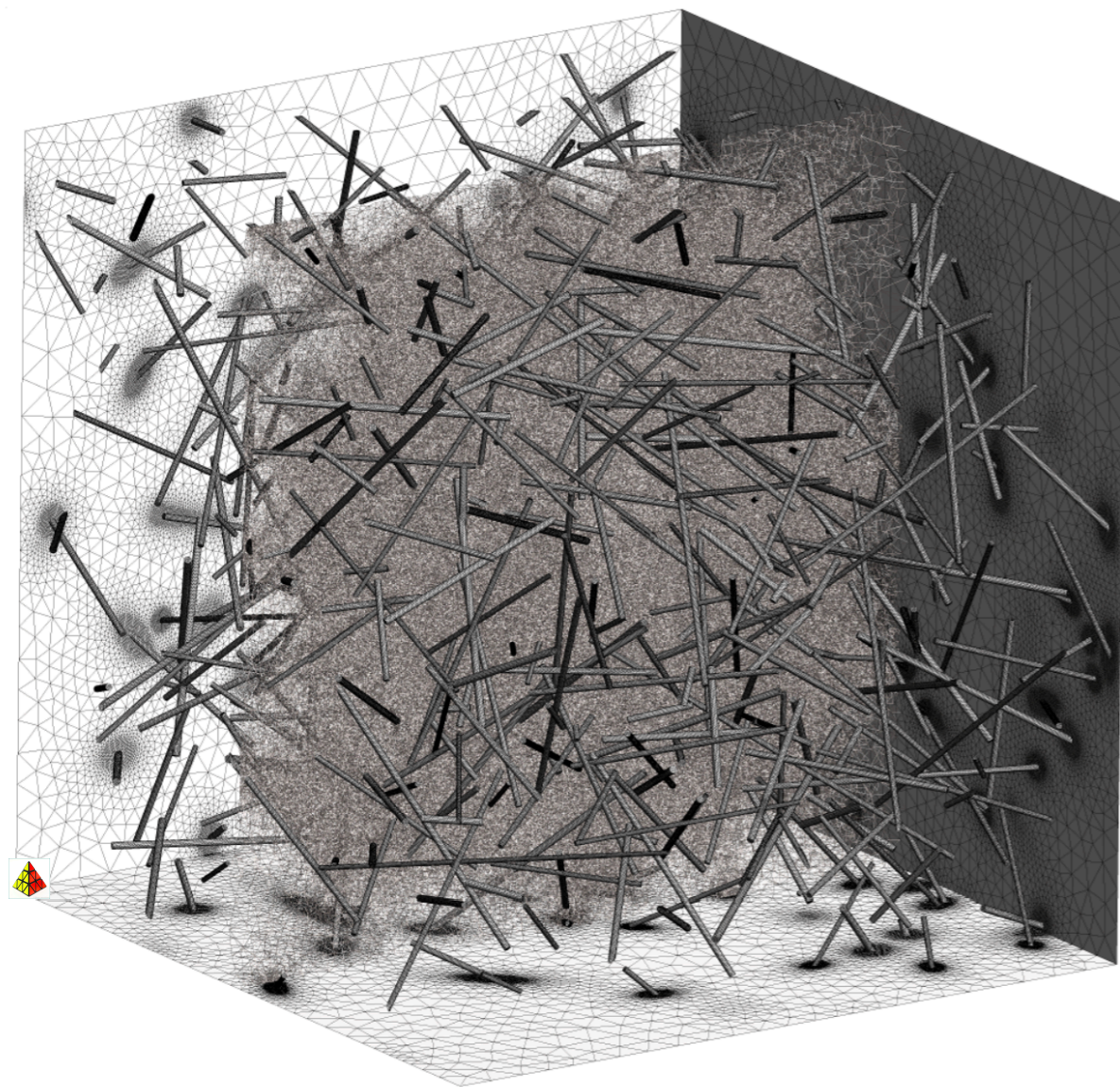
# threads	# tetrahedra	Timings (s)		
		BR	Refine	Total
1	325 611 841	3.1	492.1	497.2
2	325 786 170	2.9	329.7	334.3
4	325 691 796	2.8	229.5	233.9
8	325 211 989	2.7	154.6	158.7
16	324 897 471	2.8	96.8	100.9
32	325 221 244	2.7	71.7	75.8
64	324 701 883	2.8	55.8	60.1
127	324 190 447	2.9	47.6	52.0

## 500 thin fibers

# threads	# tetrahedra	Timings (s)		
		BR	Refine	Total
1	723 208 595	18.9	1205.8	1234.4
2	723 098 577	16.0	780.3	804.8
4	722 664 991	86.6	567.1	659.8
8	722 329 174	15.8	349.1	370.1
16	723 093 143	15.6	216.2	236.5
32	722 013 476	15.6	149.7	169.8
64	721 572 235	15.9	119.7	140.4
127	721 591 846	15.9	114.2	135.2

# One machine, one minute, three billion tetrahedra

Célestin Marot\* | Jeanne Pellerin | Jean-François Remacle

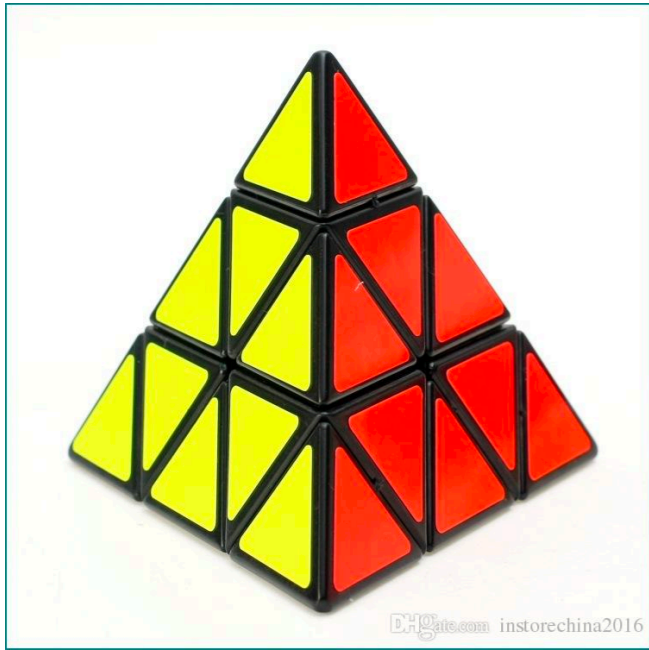


## 100 thin fibers

# threads	# tetrahedra	Timings (s)		
		BR	Refine	Total
1	325 611 841	3.1	492.1	497.2
2	325 786 170	2.9	329.7	334.3
4	325 691 796	2.8	229.5	233.9
8	325 211 989	2.7	154.6	158.7
16	324 897 471	2.8	96.8	100.9
32	325 221 244	2.7	71.7	75.8
64	324 701 883	2.8	55.8	60.1
127	324 190 447	2.9	47.6	52.0

## 500 thin fibers

# threads	# tetrahedra	Timings (s)		
		BR	Refine	Total
1	723 208 595	18.9	1205.8	1234.4
2	723 098 577	16.0	780.3	804.8
4	722 664 991	86.6	567.1	659.8
8	722 329 174	15.8	349.1	370.1
16	723 093 143	15.6	216.2	236.5
32	722 013 476	15.6	149.7	169.8
64	721 572 235	15.9	119.7	140.4
127	721 591 846	15.9	114.2	135.2

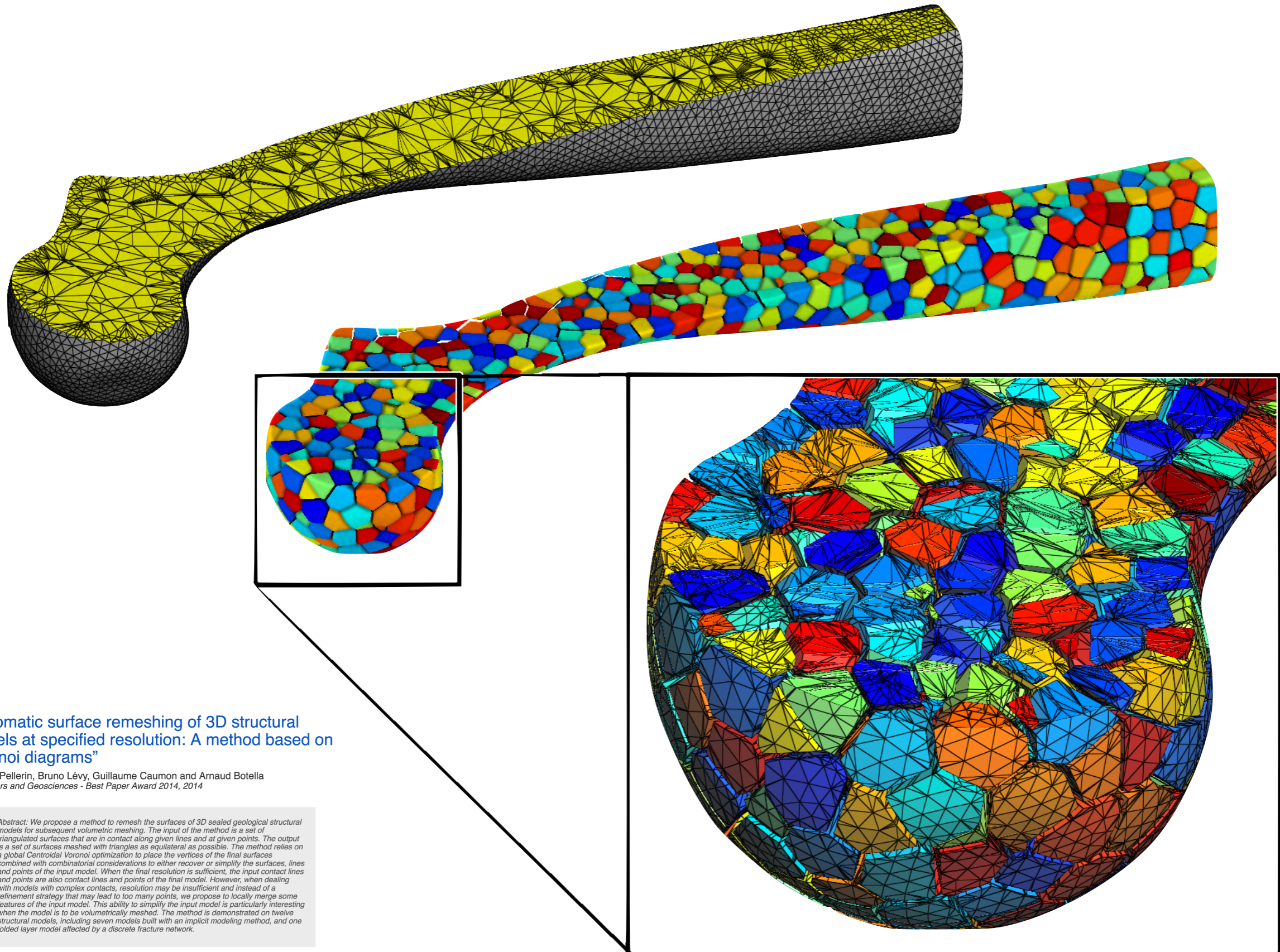


Linear tetrahedral elements are limited

- Stiff
- Locking
- ...



# Alternative elements - polyhedral - virtual elements, HHO, SBFEM, smoothed FEM...

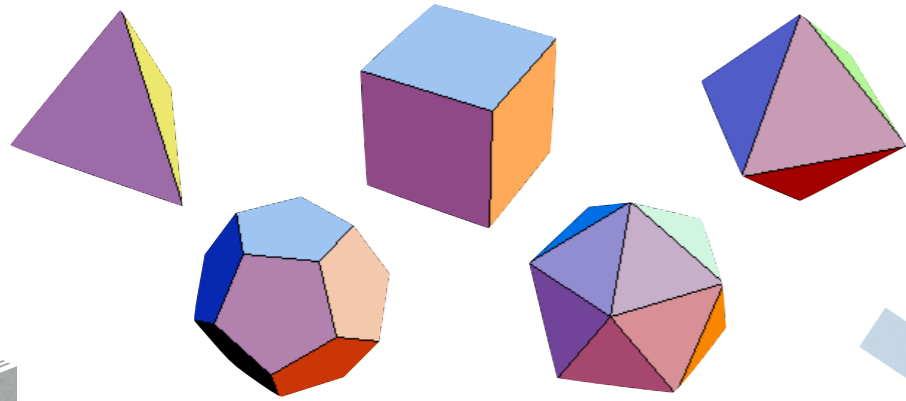


“Automatic surface remeshing of 3D structural models at specified resolution: A method based on Voronoi diagrams”

Jeanne Pellerin, Bruno Lévy, Guillaume Caumon and Arnaud Botella  
Computers and Geosciences - Best Paper Award 2014, 2014

*Abstract: We propose a method to remesh the surfaces of 3D sealed geological structural models for subsequent volumetric meshing. The input of the method is a set of triangulated surfaces that are in contact along given lines and at given points. The output is a set of surfaces meshed with triangles as equilateral as possible. The method relies on a global Centroidal Voronoi optimization to place the vertices of the final surfaces combined with combinatorial considerations to either recover or simplify the surfaces, lines and points of the input model. When the final resolution is sufficient, the input contact lines and points are also contact lines and points of the final model. However, when dealing with models with complex contacts, resolution may be insufficient and instead of a refinement strategy that may lead to too many points, we propose to locally merge some features of the input model. This ability to simplify the input model is particularly interesting when the model is to be volumetrically meshed. The method is demonstrated on twelve structural models, including seven models built with an implicit modeling method, and one folded layer model affected by a discrete fracture network.*

# Use polyhedra



## Virtual elements (cf. Silvia Bertoluzza)

[The hitchhiker's guide to the virtual element method](#)

[Virtual and smoothed finite elements: A connection and its application to polygonal/polyhedral finite element methods](#) (Natarajan, Ooi, Bordas)

## Smoothed FEM

A theoretical study on the **smoothed FEM (S-FEM)** models: Properties, accuracy and convergence rates (G.R. Liu, Nguyen et al)

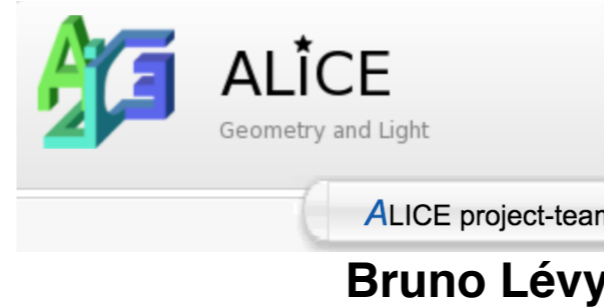
On the approximation in the **smoothed** finite element method (SFEM) (Natarajan, Bordas)

## Scaled boundary FEM

[The scaled boundary finite-element method—a primer: derivations](#) (Song, Wolf, 2000)

## HHO (cf. F. Chouly and G. Delay)

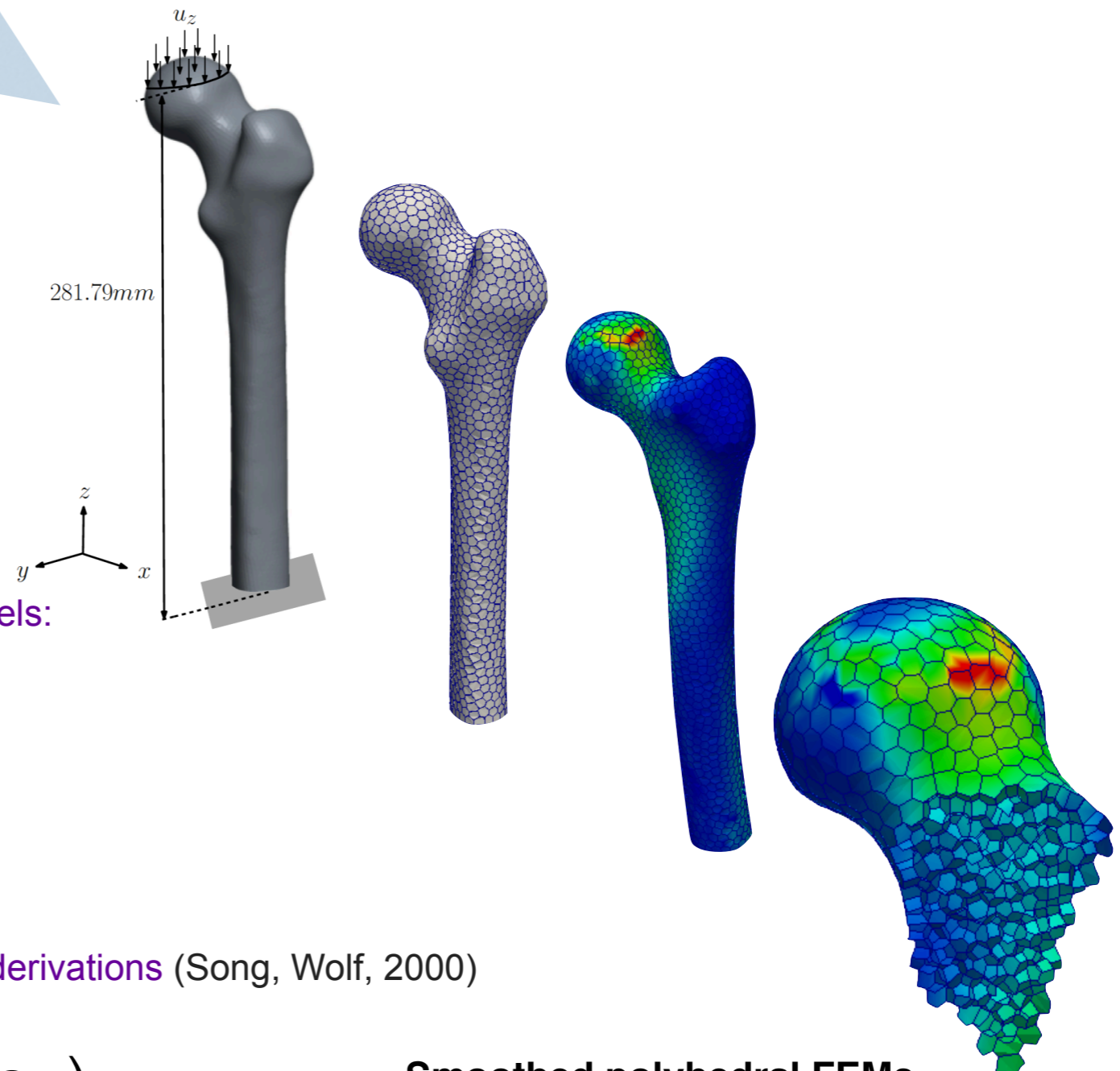
# Mesh generators...



## 3D NOffset mixed-element mesh generator approach

Claudio Lobos and Nancy Hitschfeld-Kahler  
Depto Ciencias de la Computación,  
FCFM, Universidad de Chile,  
Blanco Encalada 2120,  
Santiago - Chile, Zip code: 837-0459  
clobos@dcc.uchile.cl, nancy@dcc.uchile.cl

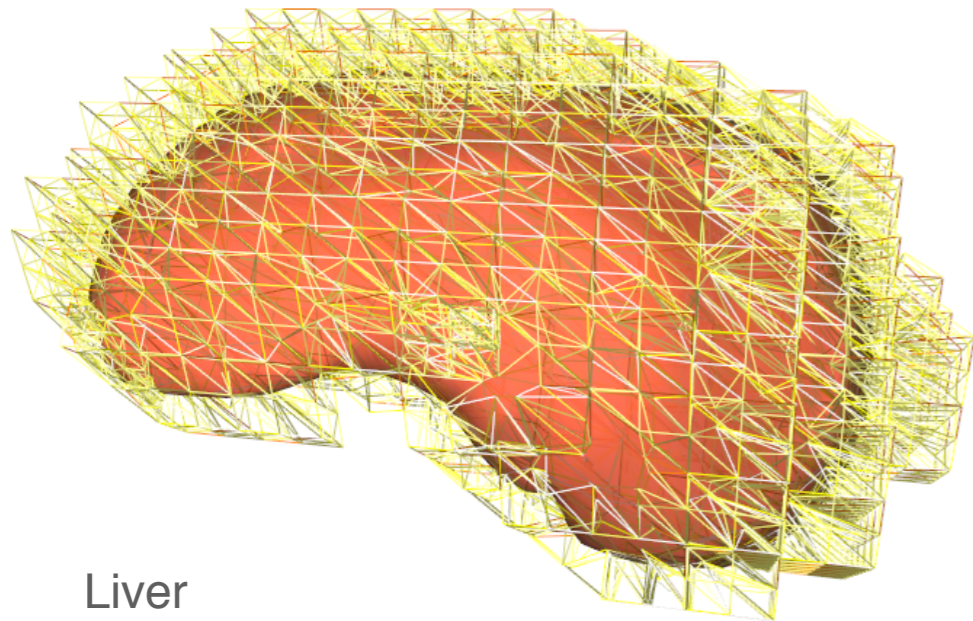
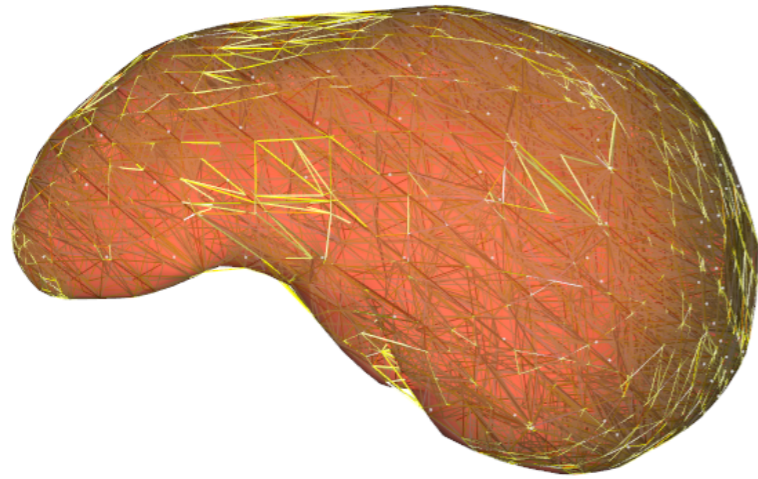
**Claudio Lobos**



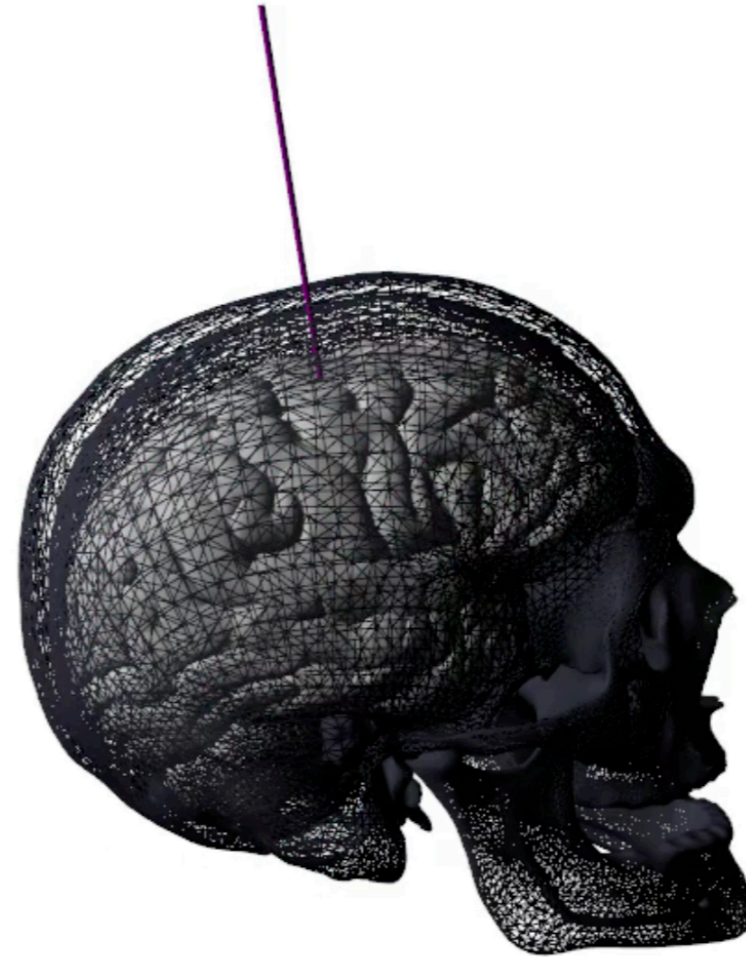
**Smoothed polyhedral FEMs**  
Francis, Natarajan, Lévy, Bordas, 2019

# Avoid meshing complex/evolving interfaces through unfitted methods

Implicit boundaries and error control for real time simulations



Liver



Deep brain stimulation simulation



Real-time Error Control for Surgical Simulation, HP Bui et al, **IEEE Trans. Biomed. Eng.**, 2016.

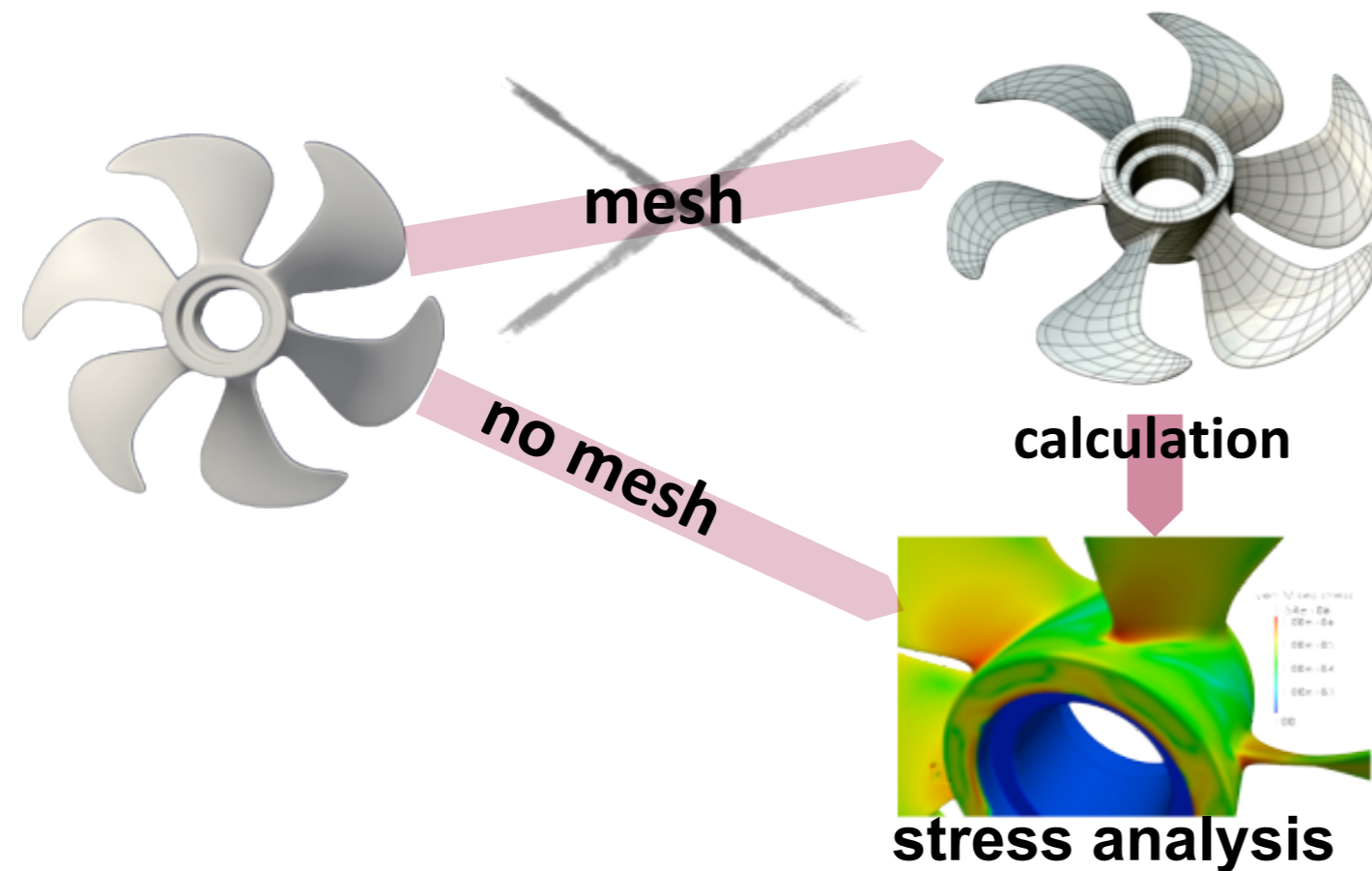
Controlling the Error on Target Motion through Real-time Mesh Adaptation: Applications to Deep Brain Stimulation, HP Bui et al, **Int J Numer Meth Bio**, 2017.

Corotational Cut Finite Element Method for real-time surgical simulation: application to needle insertion simulation, HP Bui et al, **arXiv:1712.03052[cs.CE]** 2018.



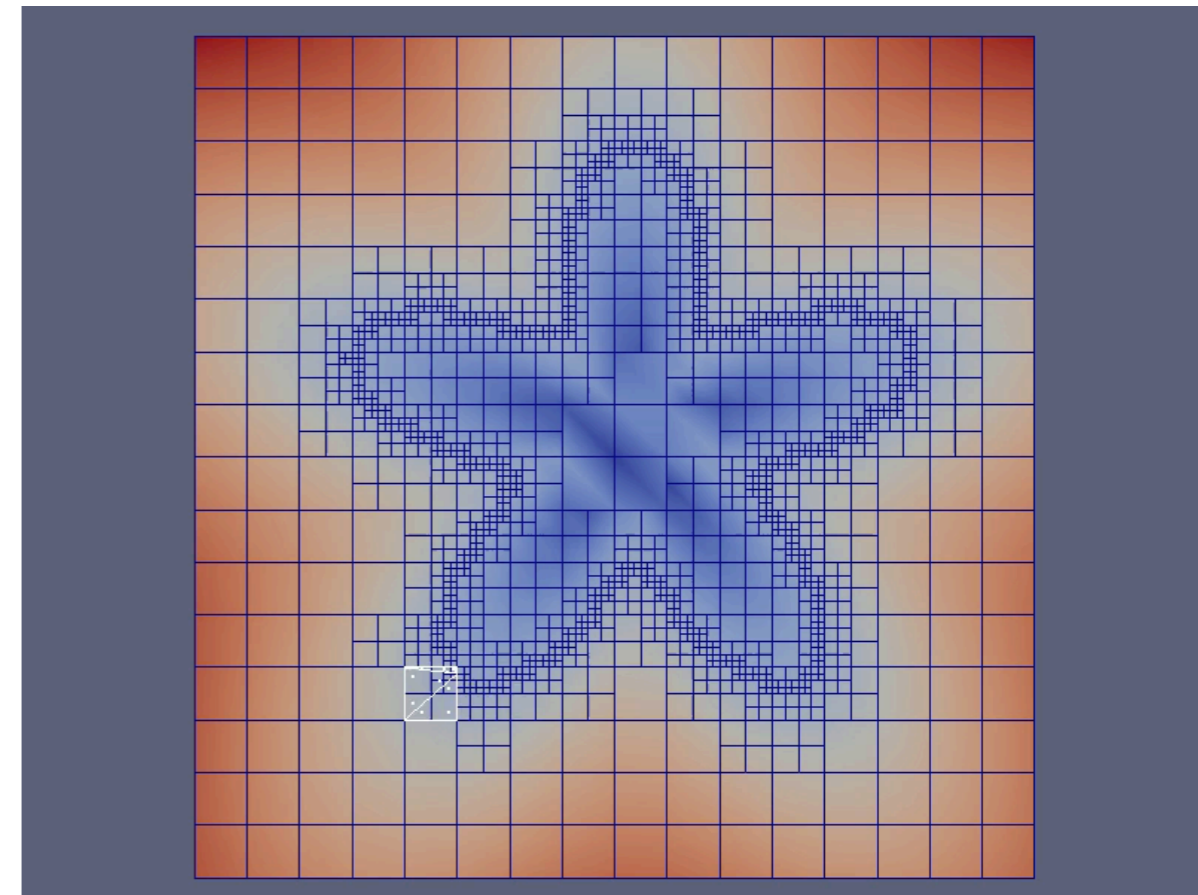
# Handling interfaces numerically

## Couple geometry & analysis



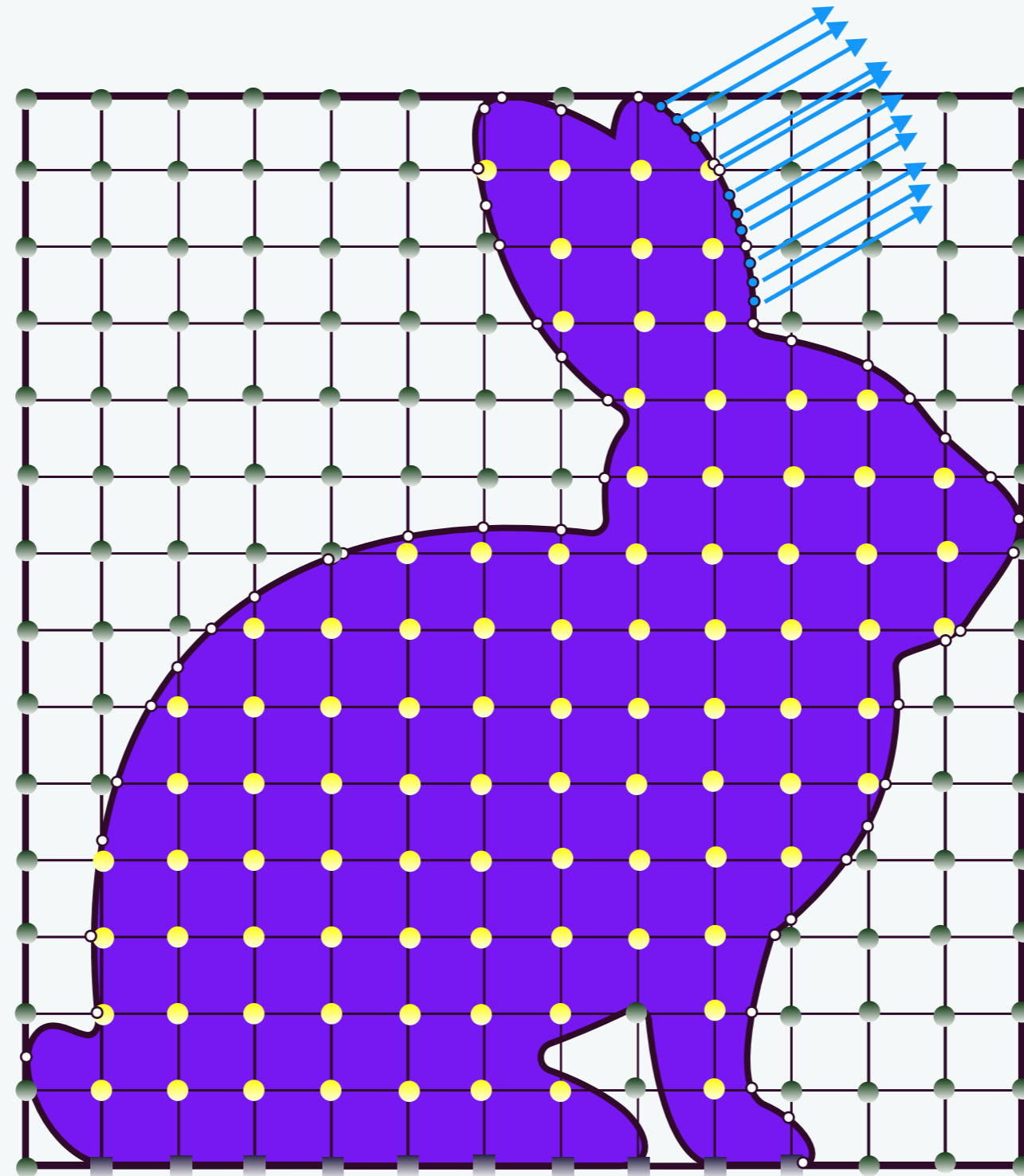
Isogeometric analysis

## Decouple geometry from analysis



Implicit interfaces/unfitted

# Immersed collocation generalized FD





+



Computed in Luxembourg

## Main collaborators

Kostas

Agathos

Tahsin

Khajah

Cosmin

Anitorescu

Alexei

Lozinski

Elena

Atroshchenko

Thibault

Jacquemin

Davide

Baroli

Haojie

Lian

Lars

Beex

Mohammed

Moumnassi

Phuoc

Bui

Xuan

Peng

Eleni

Chatzi

Timon

Rabczuk

Leilei

Chen

Hussein

Rappel

Franz

Chouly

Sundar

Natarajan

Susanne

Claus

Danas

Sutula

Sofia

Farina

Satyendra

Tomar

Jack

Hale

Javier

Videla

Paul

Hauseux

Gang

Xu

Qingyuan

Hu

Peng

Yu

Pierre

Kerfriden

Andreas

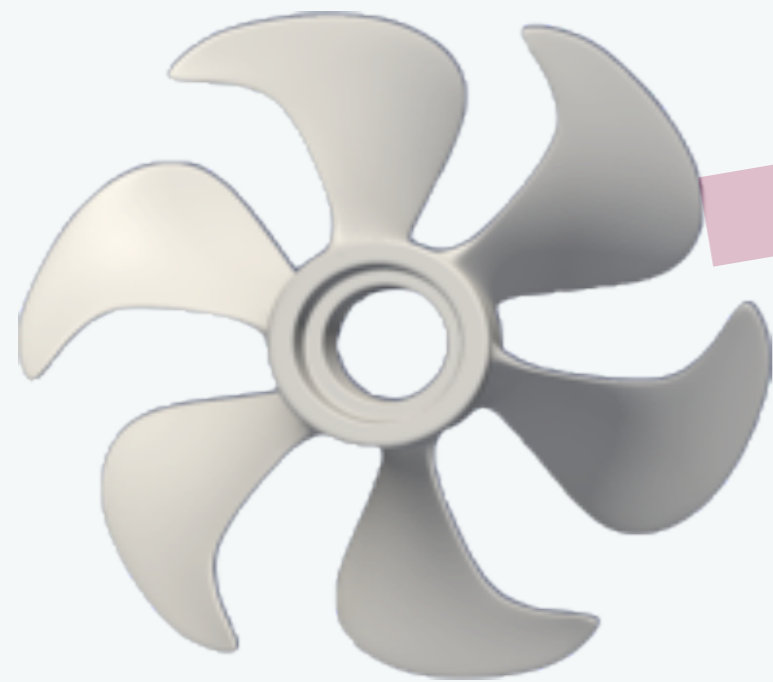
Zilian



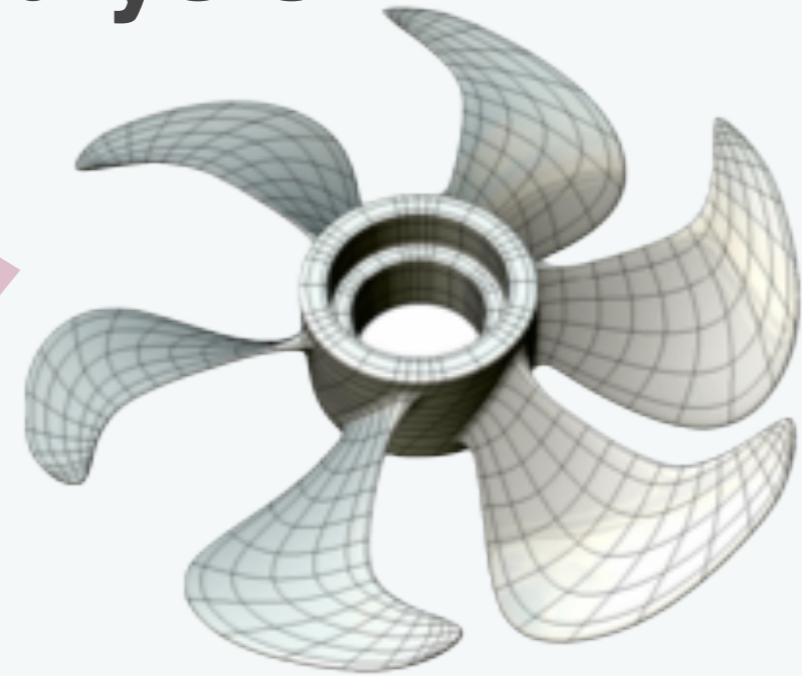
UNIVERSITÉ DU  
LUXEMBOURG

# Simplify CAD-Analysis

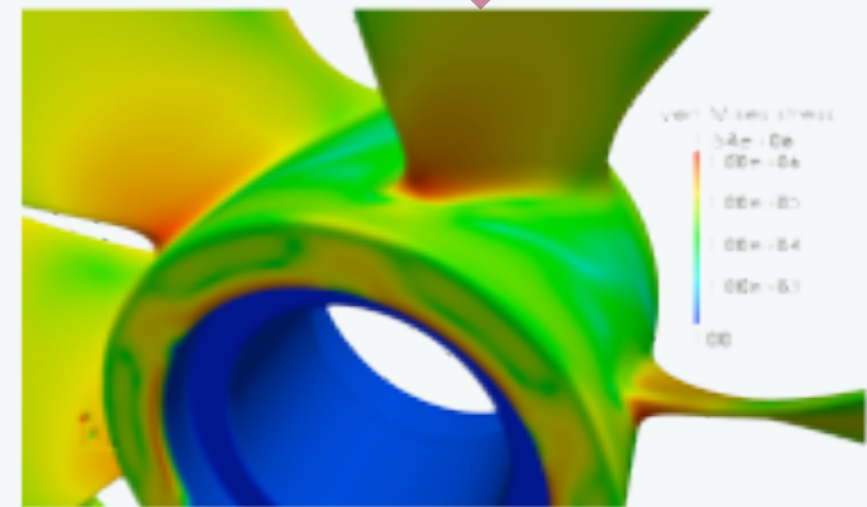
## Isogeometric analysis



mesh

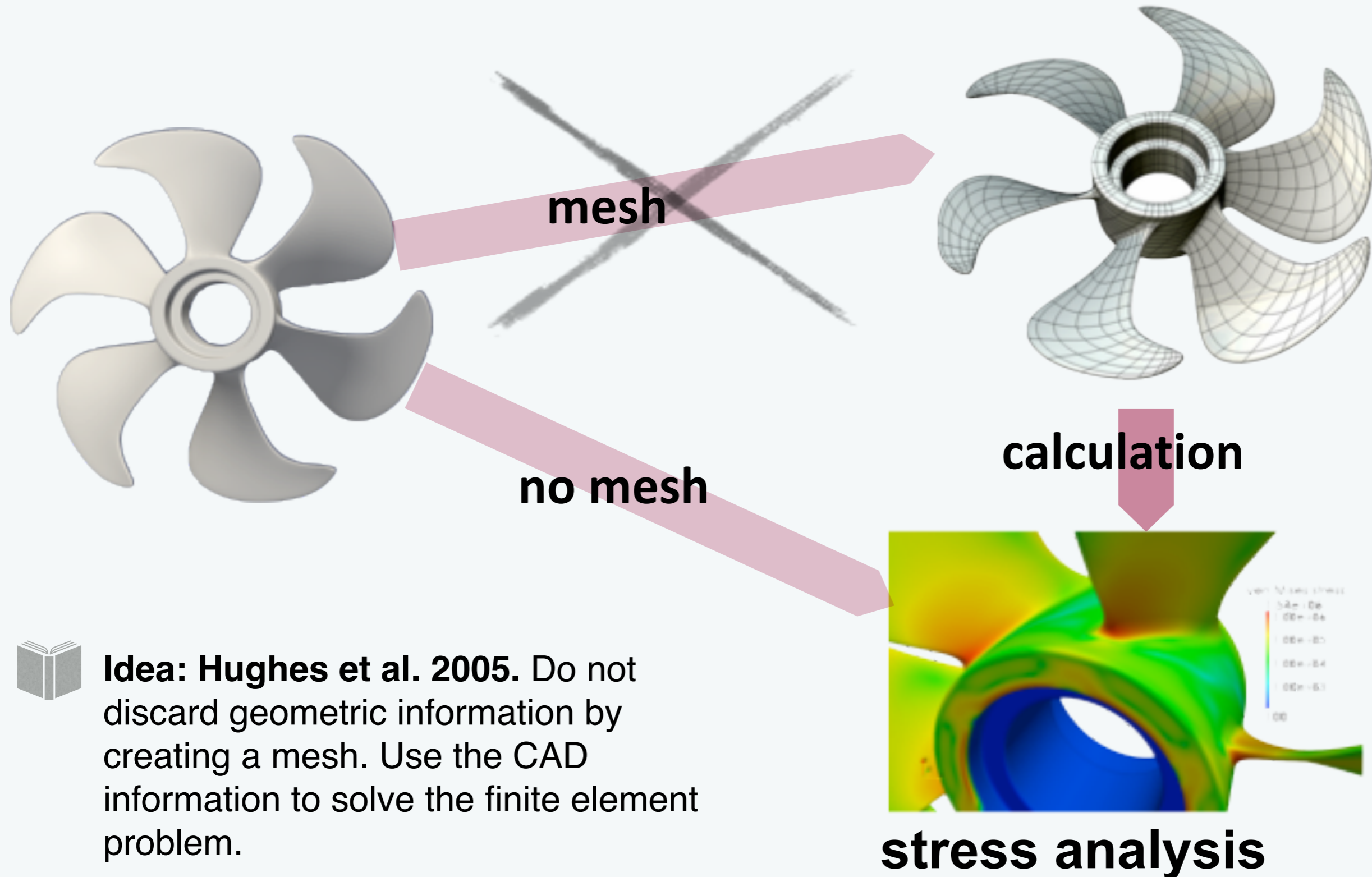


calculation



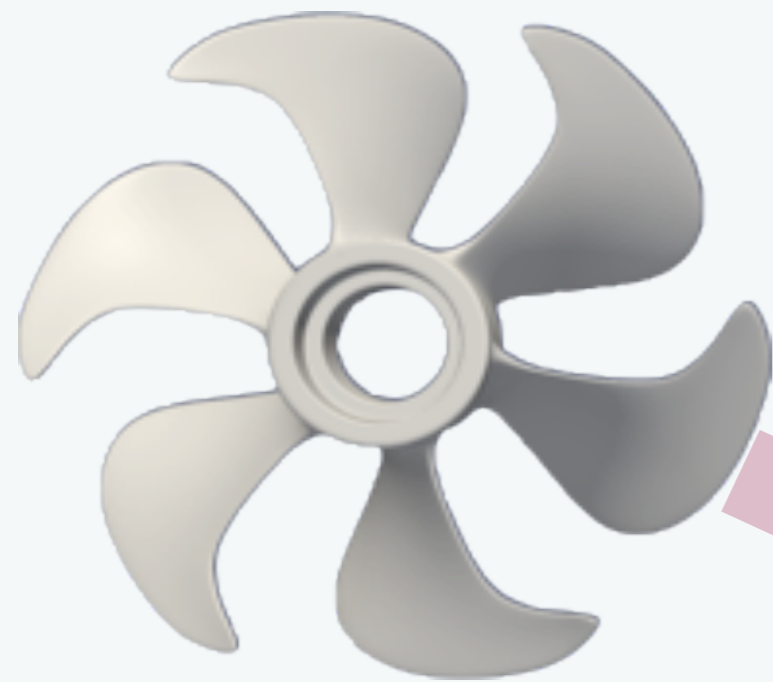
stress analysis

# Isogeometric analysis

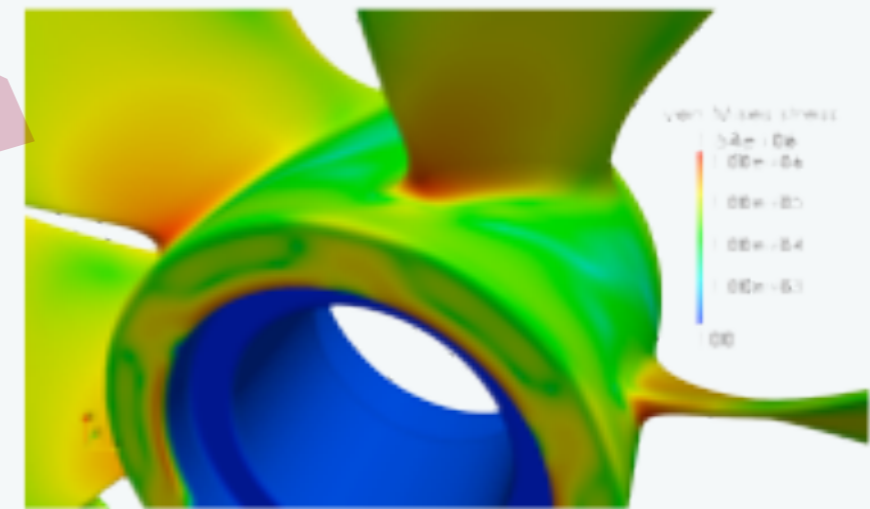


**Idea: Hughes et al. 2005.** Do not discard geometric information by creating a mesh. Use the CAD information to solve the finite element problem.





**direct calculation**



**stress analysis**

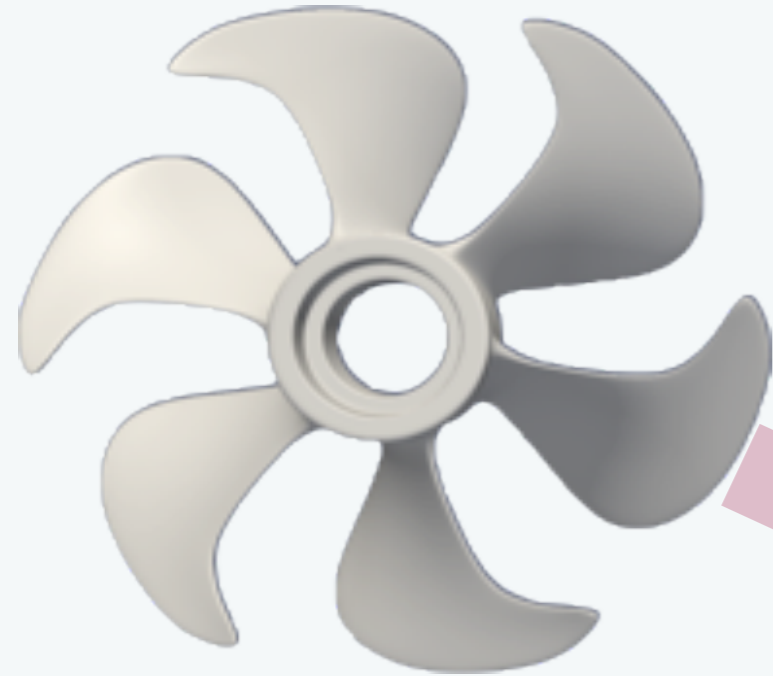


Isogeometric analysis: CAD, finite elements, NURBS, exact geometry and mesh refinement (CMAME05, T.J.R Hughes et al)

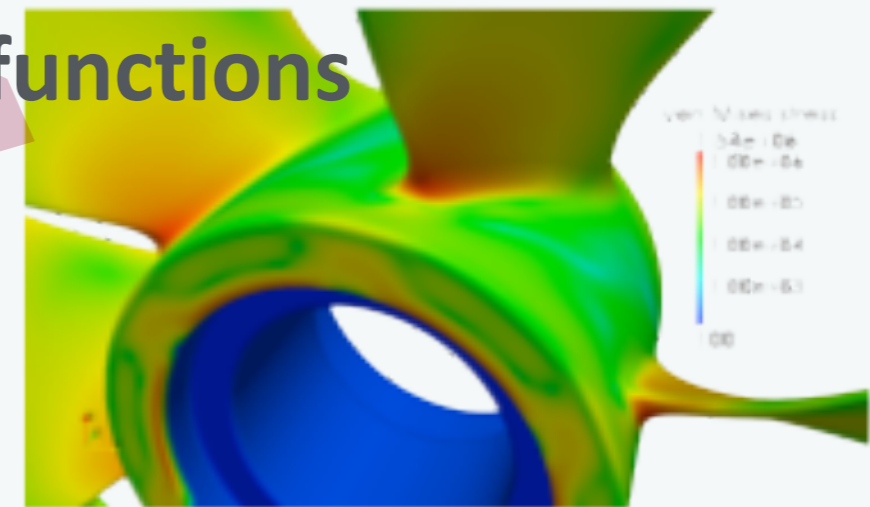


Isogeometric boundary element analysis using unstructured T-splines (CMAME13, M.A. Scott et al)

## CAD: described by NURBS



Use NURBS as basis functions



**stress analysis**



Isogeometric analysis: CAD, finite elements, NURBS, exact geometry and mesh refinement (CMAME05, T.J.R Hughes et al)



Isogeometric boundary element analysis using unstructured T-splines (CMAME13, M.A. Scott et al)

## NURBS

Universality.  
Industrial standard.  
Not watertight.  
No local refinement



Fig. 1: NURBS  
(source: Rhino3D website)

## T-splines

Water tight.  
Local refinement.  
No trimming operations.  
No linear independence.

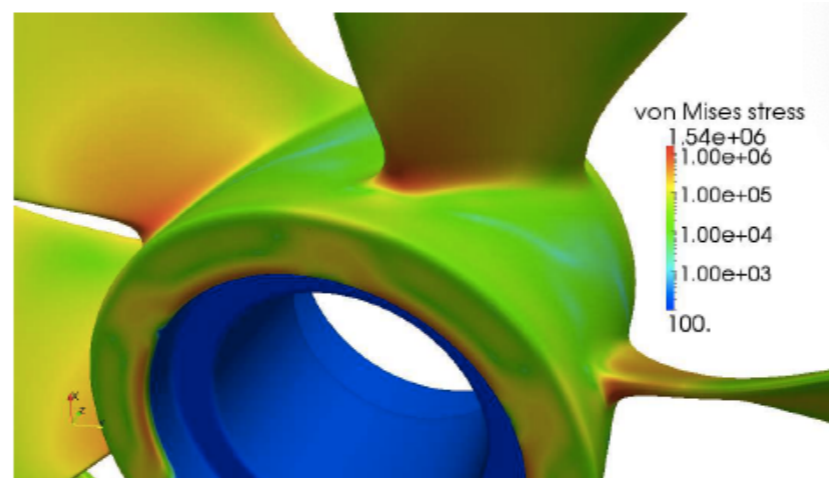


Fig. 2: T-splines

## Subdivision Surfaces

Watertight.  
Flexibility.  
3D printing.  
Low order smoothness.

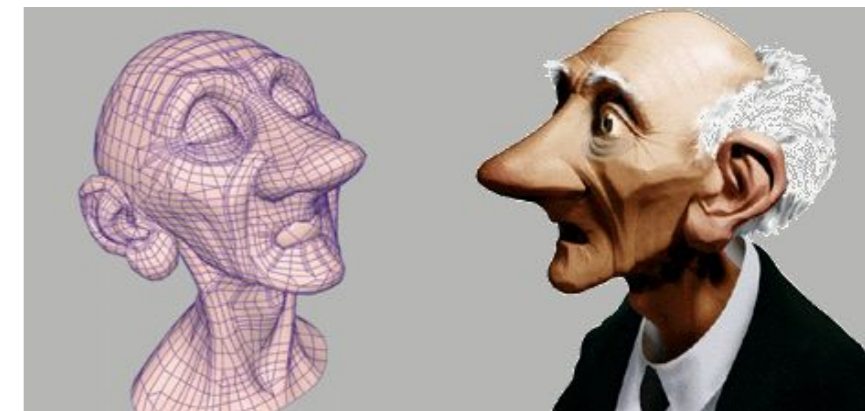


Fig. 3: Subdivision surfaces  
(source: Geri's game)

*All are boundary representations*

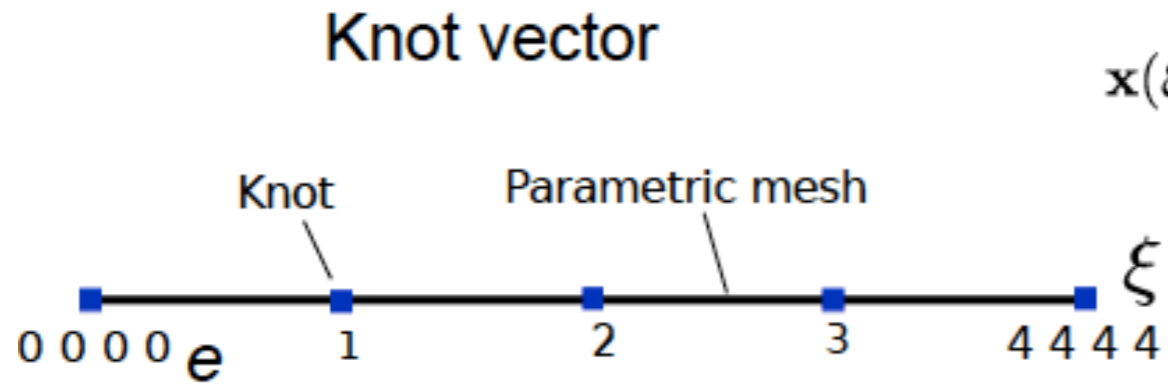


[Isogeometric analysis: an overview and computer implementation aspects](#)

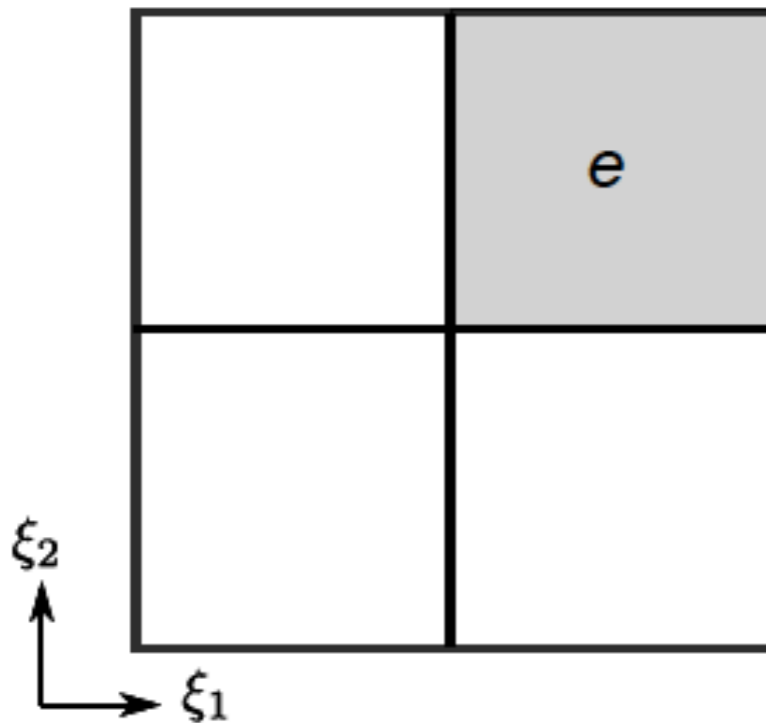
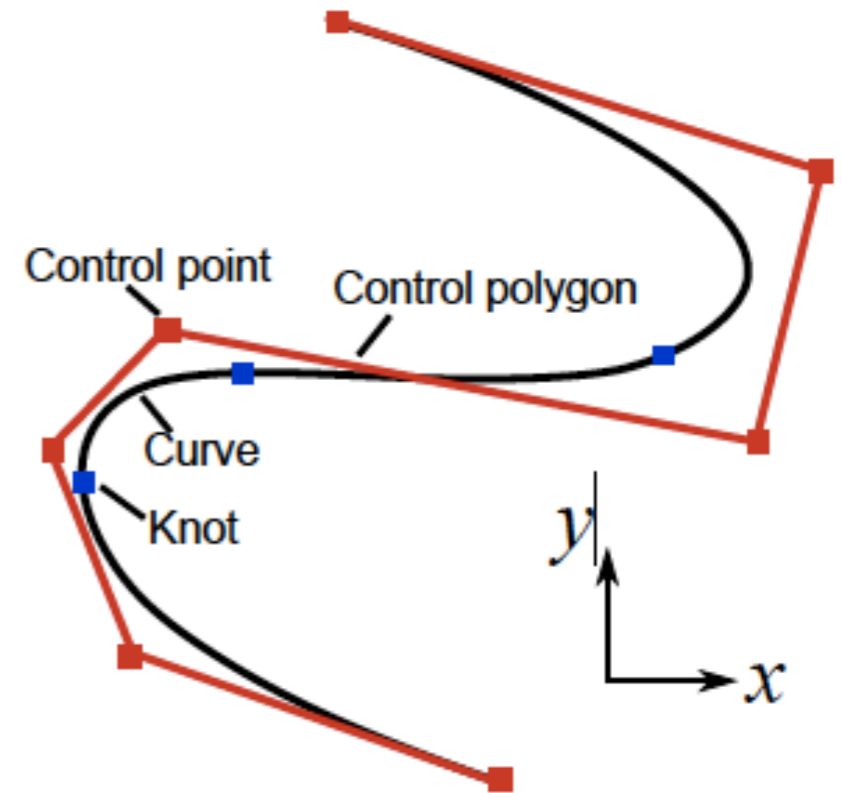
VP Nguyen, C Anitescu, SPA Bordas, T Rabczuk  
Mathematics and Computers in Simulation 117, 89-116



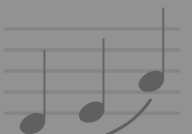
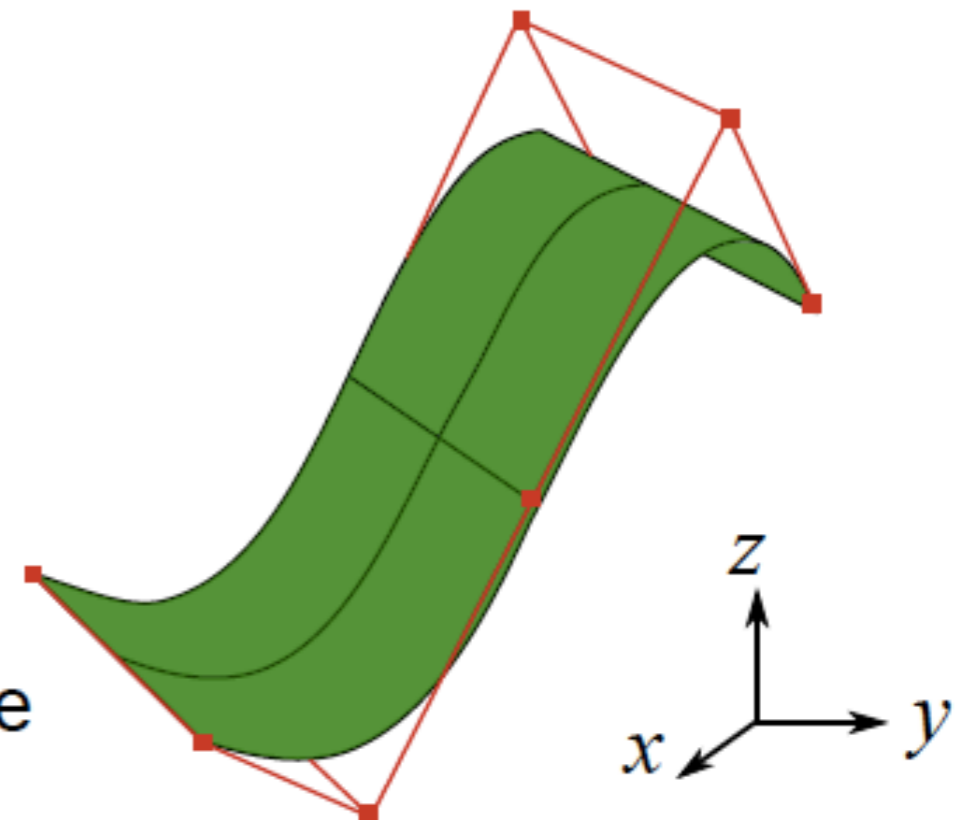
# NURBS



$$\mathbf{x}(\xi) = \sum_A N_A(\xi) \mathbf{P}_A$$



NURBS surface



# NURBS

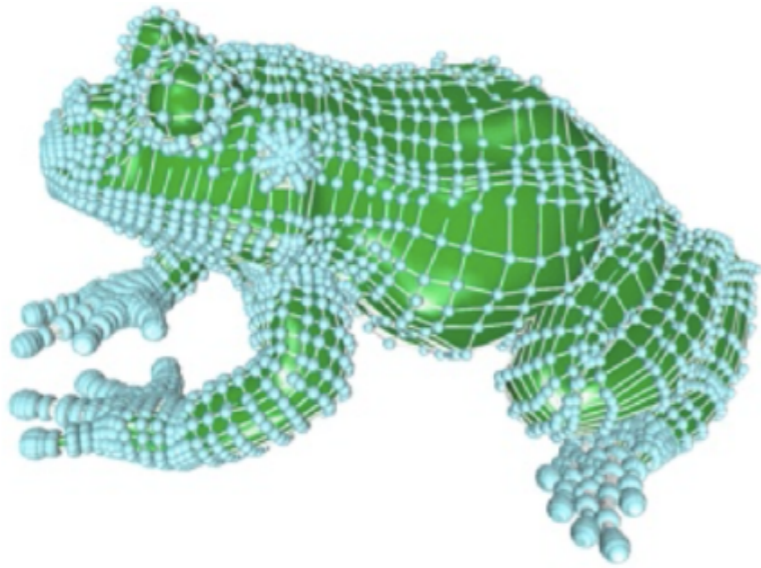


Fig. 1: NURBS(source: Rhino3D website)

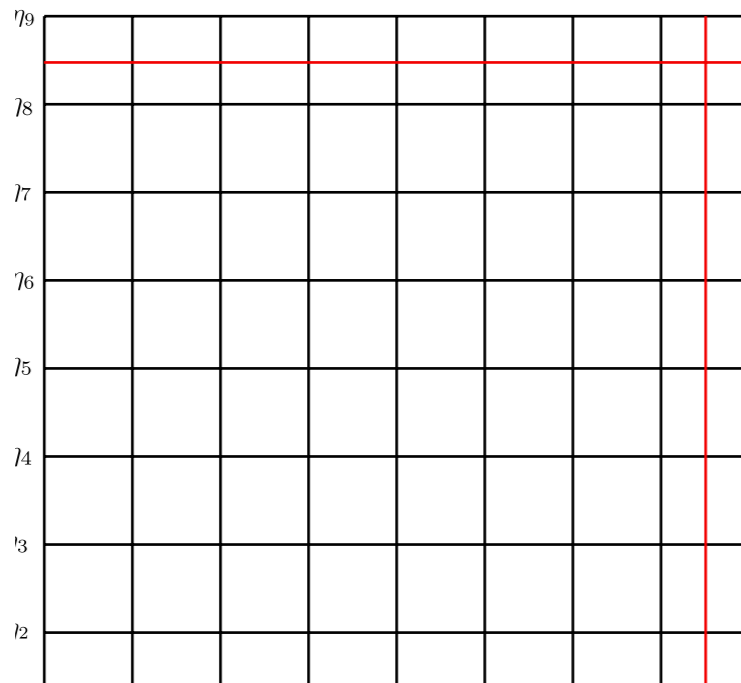
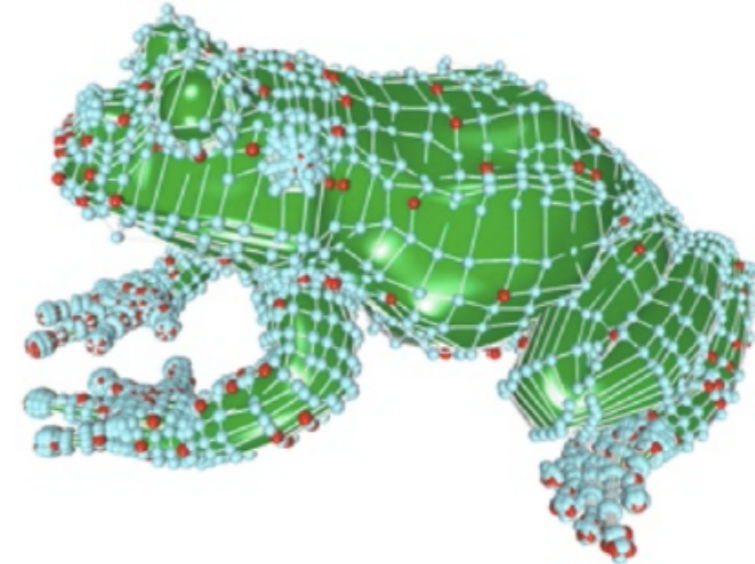


Fig. 3: NURBS mesh topology

# T-splines



Fewer control points for the same geometry

Fig. 2: T-splines(source: Rhino3D website)

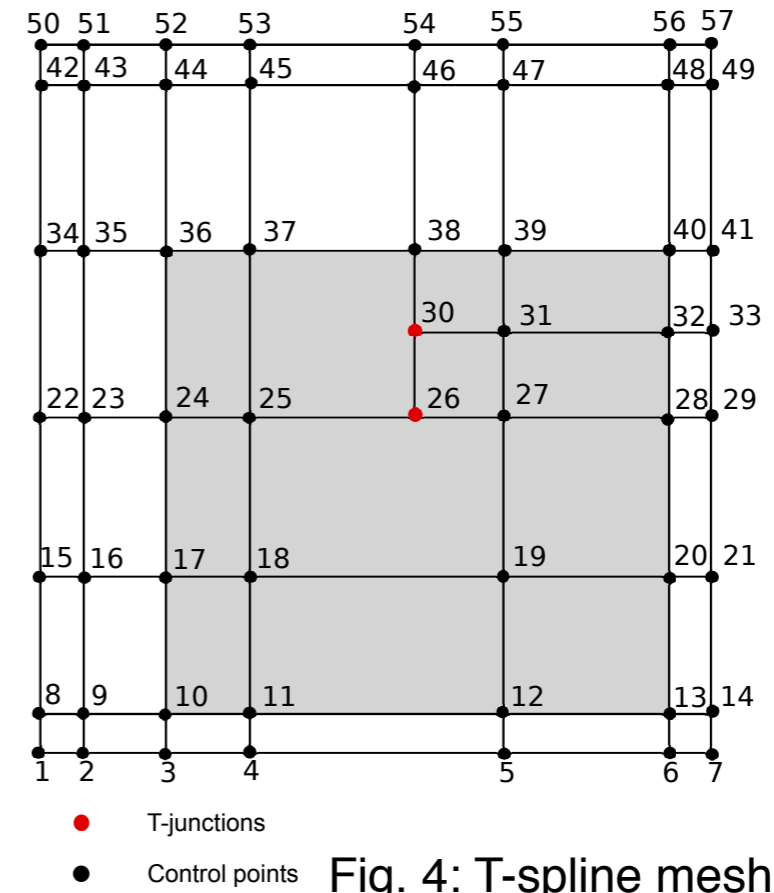
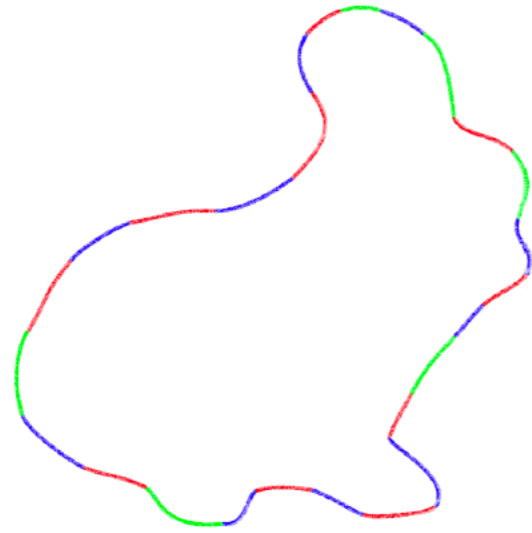
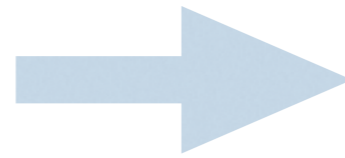


Fig. 4: T-spline mesh

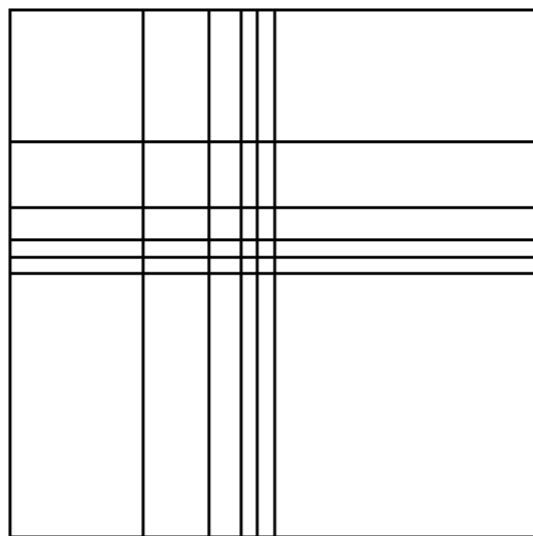
# Difficulties



**Given by CAD**



**Needed by IGAFEM**



**Local refinement**



**Patch coupling**

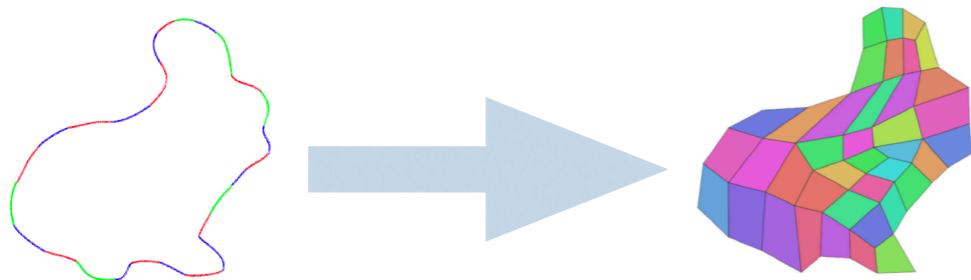
# IGABEM vs. IGAFEM

Question: How can we fully benefit from the “IGA” concept?

Suppress the mesh generation and regeneration completely

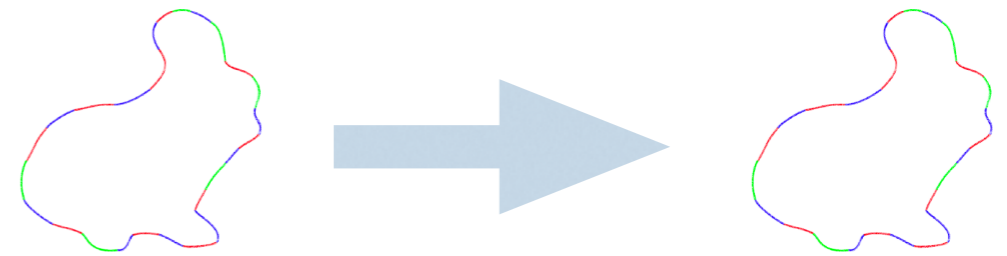
## Isogeometric FEM

- For shell-like domains
- For volumes (needs volume parameterisation, aka meshing)



## Isogeometric BEM

- For shell-like domains
- For volumes





### Stress analysis and shape optimisation directly from CAD

H. Lian et al. (2017). *CMAME*: 317: 1-41.  
H. Lian et al. (2015). *IJNME*  
H. Lian et al. (2013). *EACM*:166(2):88-99.  
M. Scott et al. (2013) *CMAME* 254: 197-221.  
R. N. Simpson et al. (2013) *CAS* 118: 2-12.  
R. N. Simpson et al. (2012) *CMAME* Feb 1;209:87-100.

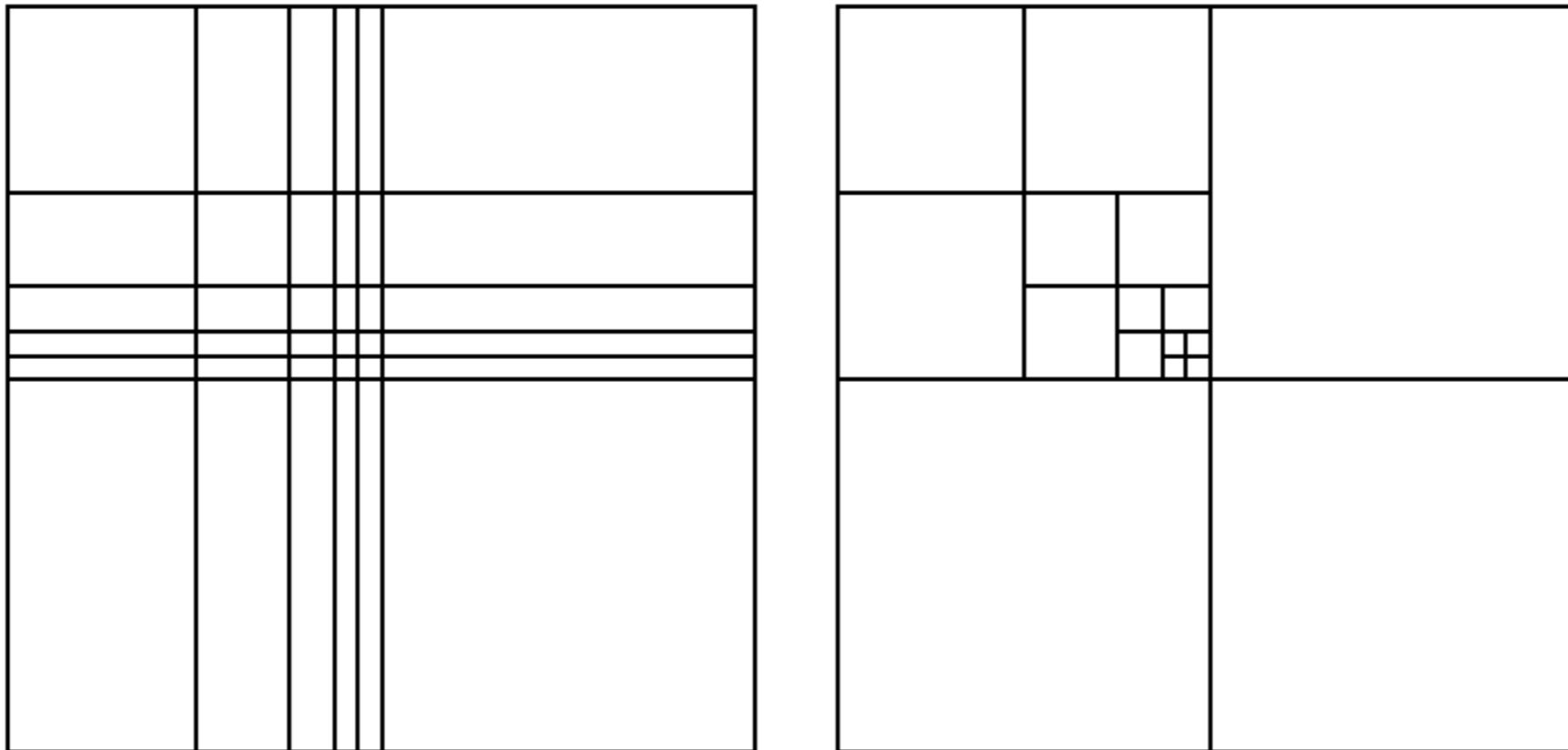
### Fracture mechanics directly from CAD

X. Peng, et al. (2017). *IJF*, 204(1), 55–78.  
X. Peng, et al. (2017). *CMAME*, 316, 151–185.

 [Nitsche's method for two and three dimensional NURBS patch coupling](#) (CMECH2014, Nguyen)

 [Skew-symmetric Nitsche's formulation in isogeometric analysis: Dirichlet and symmetry conditions, patch coupling and frictionless contact](#) (Qu, Chouly, Bordas...)

# Mesh refinement in NURBS-IGA



Global refinement (tensor-product mesh) vs local refinement (T-mesh)



# GIFT/sub/super-geometric approach

Question: How can we fully benefit from the “IGA” concept?

Refine the field approximation independently from the geometry

## Geometry Independent Field approximaTion (GIFT)

Super/Sub-geometric



[Boundary Element Analysis with trimmed NURBS and a generalized IGA approach](#)

G Beer, B Marussig, J Zechner, C Dünser, TP Fries (2014)  
arXiv preprint arXiv:1406.3499



[Fast isogeometric boundary element method based on independent field approximation](#)

B Marussig, J Zechner, G Beer, TP Fries  
Computer Methods in Applied Mechanics and Engineering 284, 458-488 (2015)

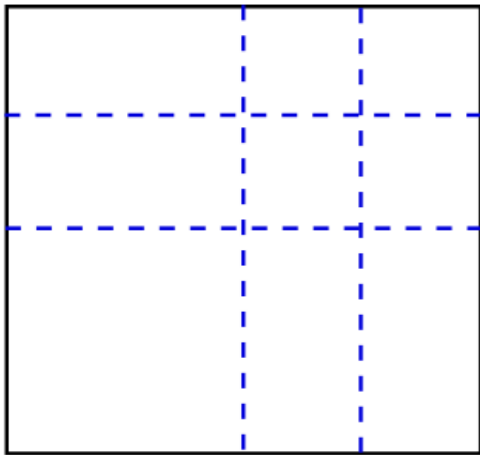


[Atroshchenko, E, et al. "Weakening the tight coupling between geometry and simulation in isogeometric analysis: From sub-and super-geometric analysis to Geometry-Independent Field approximaTion \(GIFT\)." International Journal for Numerical Methods in Engineering 114.10 \(2018\): 1131-1159.](#)

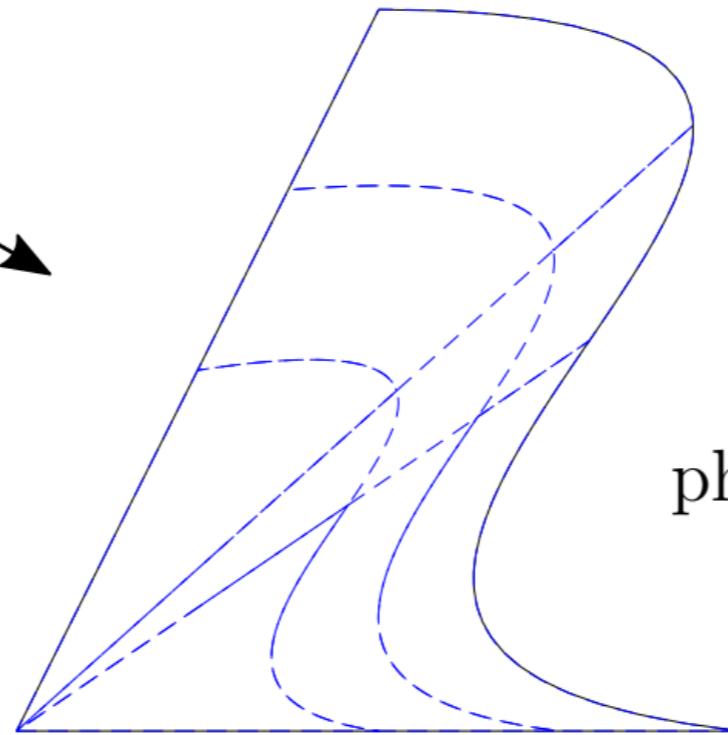
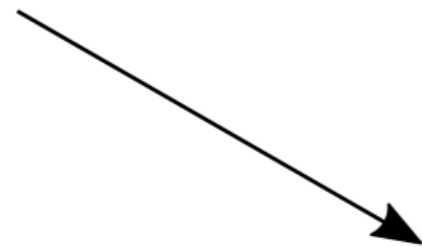
**Permalink:** <http://hdl.handle.net/10993/31469>

# Isogeometric analysis

$\mathcal{P}_{\text{NURBS}}(\boldsymbol{\xi})$

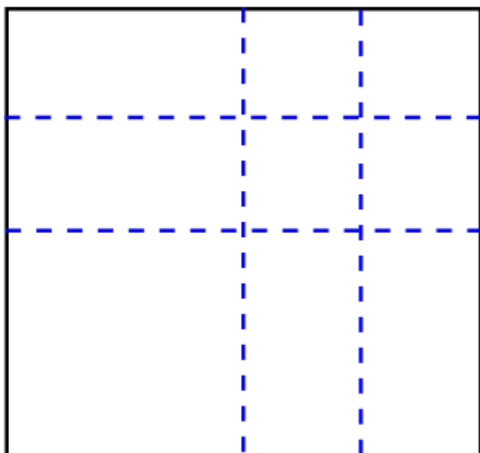


$$\boldsymbol{x} = \boldsymbol{F}(\boldsymbol{\xi}) = \sum N_{\boldsymbol{k}}^{\text{NURBS}}(\boldsymbol{\xi}) \boldsymbol{P}_{\boldsymbol{k}}$$

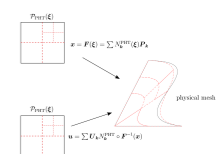


physical mesh

$\mathcal{P}_{\text{NURBS}}(\boldsymbol{\xi})$

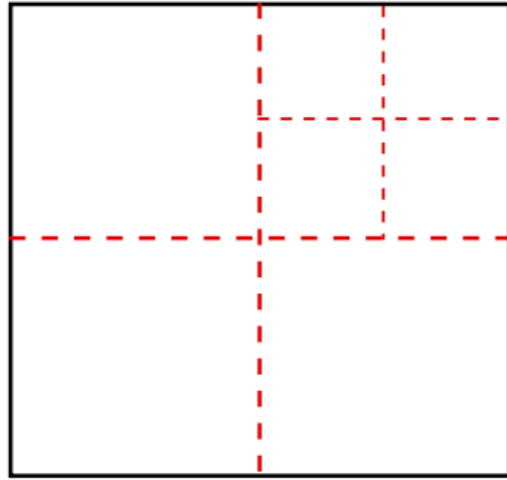


$$\boldsymbol{u} = \sum \boldsymbol{U}_{\boldsymbol{k}} N_{\boldsymbol{k}}^{\text{NURBS}} \circ \boldsymbol{F}^{-1}(\boldsymbol{x})$$

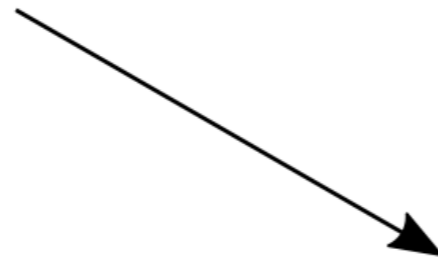


# Isogeometric analysis - PHT

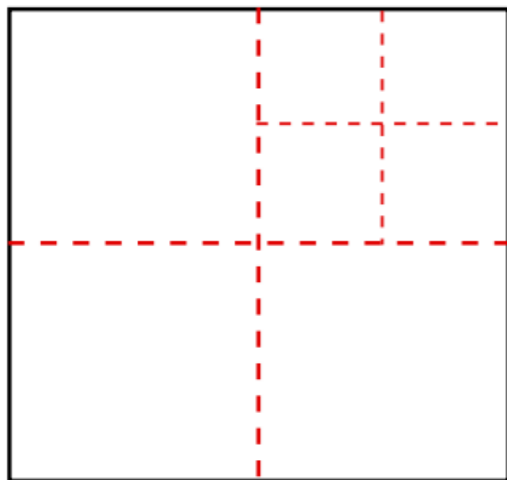
$\mathcal{P}_{\text{PHT}}(\xi)$



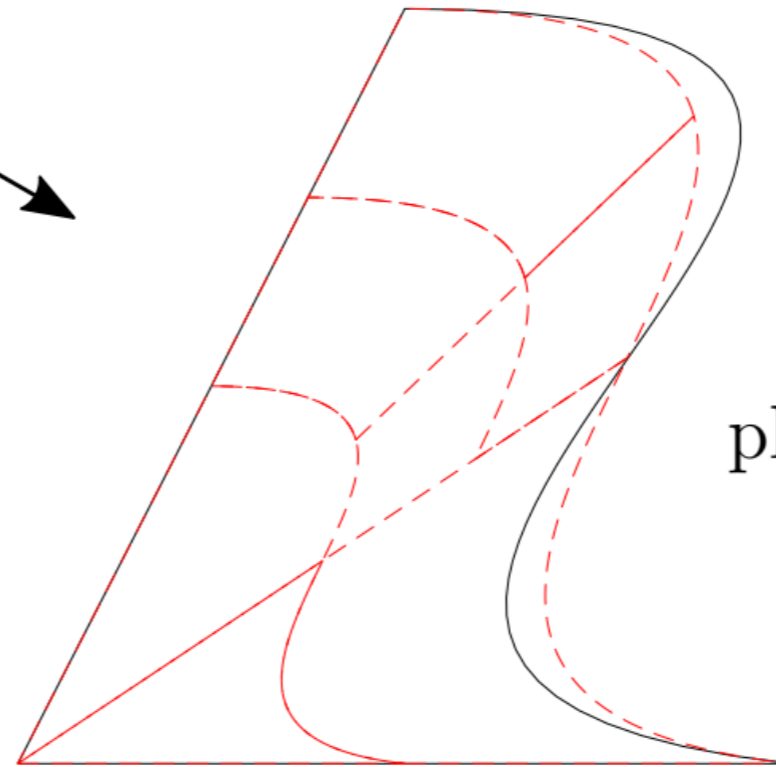
$$\mathbf{x} = \mathbf{F}(\xi) = \sum N_k^{\text{PHT}}(\xi) \mathbf{P}_k$$



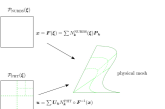
$\mathcal{P}_{\text{PHT}}(\xi)$



$$\mathbf{u} = \sum \mathbf{U}_k N_k^{\text{PHT}} \circ \mathbf{F}^{-1}(\mathbf{x})$$



physical mesh



# Geo-independent field approximation

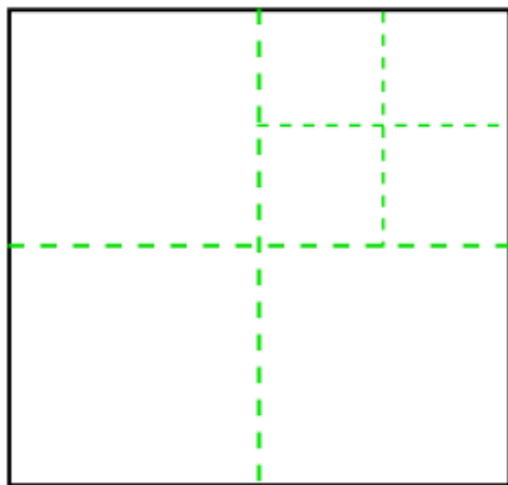
$\mathcal{P}_{\text{NURBS}}(\boldsymbol{\xi})$



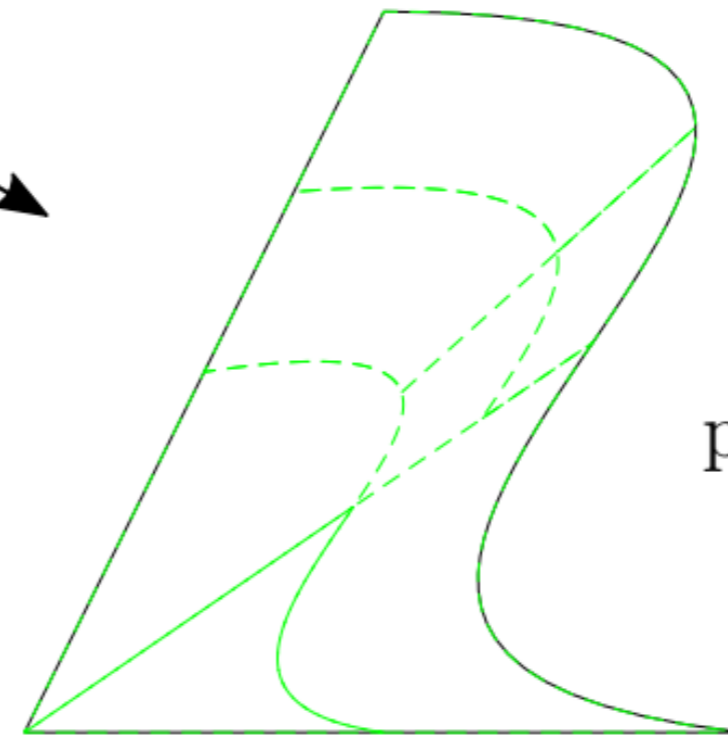
$$\boldsymbol{x} = \boldsymbol{F}(\boldsymbol{\xi}) = \sum N_k^{\text{NURBS}}(\boldsymbol{\xi}) \boldsymbol{P}_k$$



$\mathcal{P}_{\text{PHT}}(\boldsymbol{\xi})$

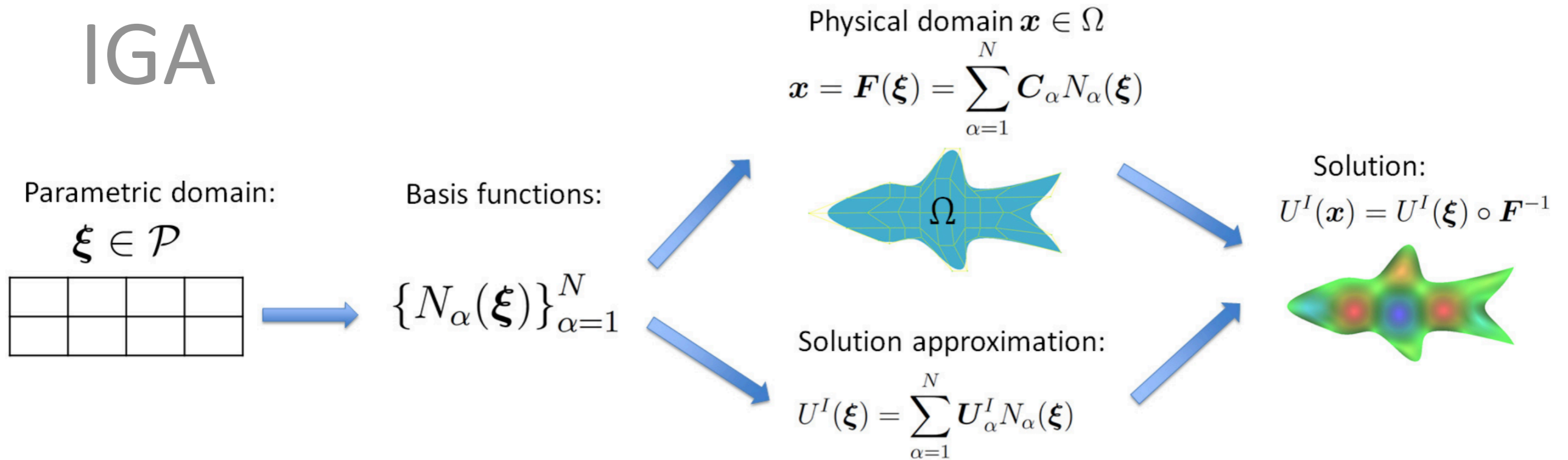


$$\boldsymbol{u} = \sum \boldsymbol{U}_k N_k^{\text{PHT}} \circ \boldsymbol{F}^{-1}(\boldsymbol{x})$$

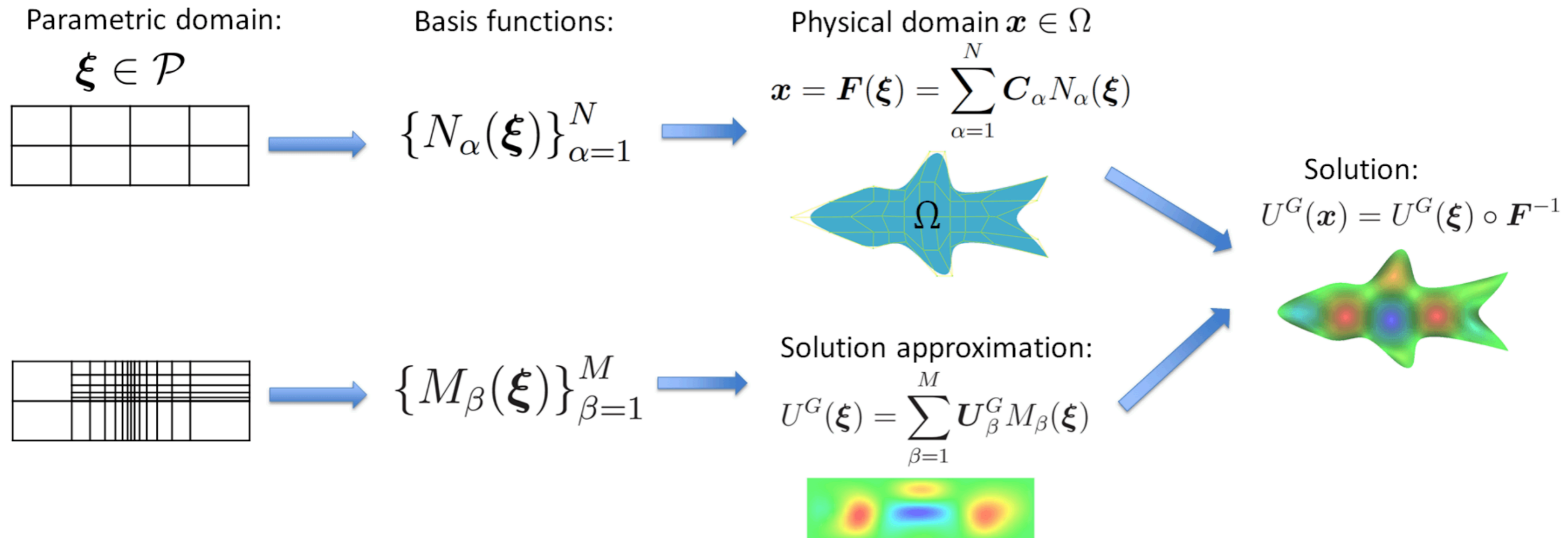


physical mesh

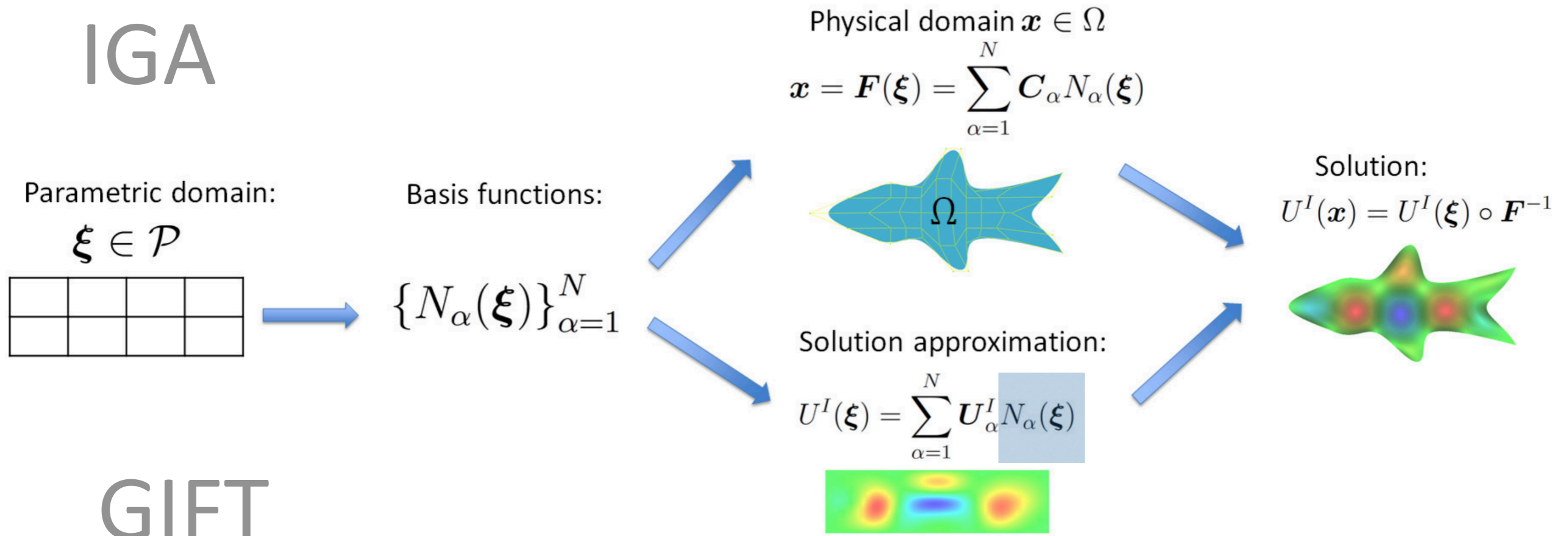
# IGA



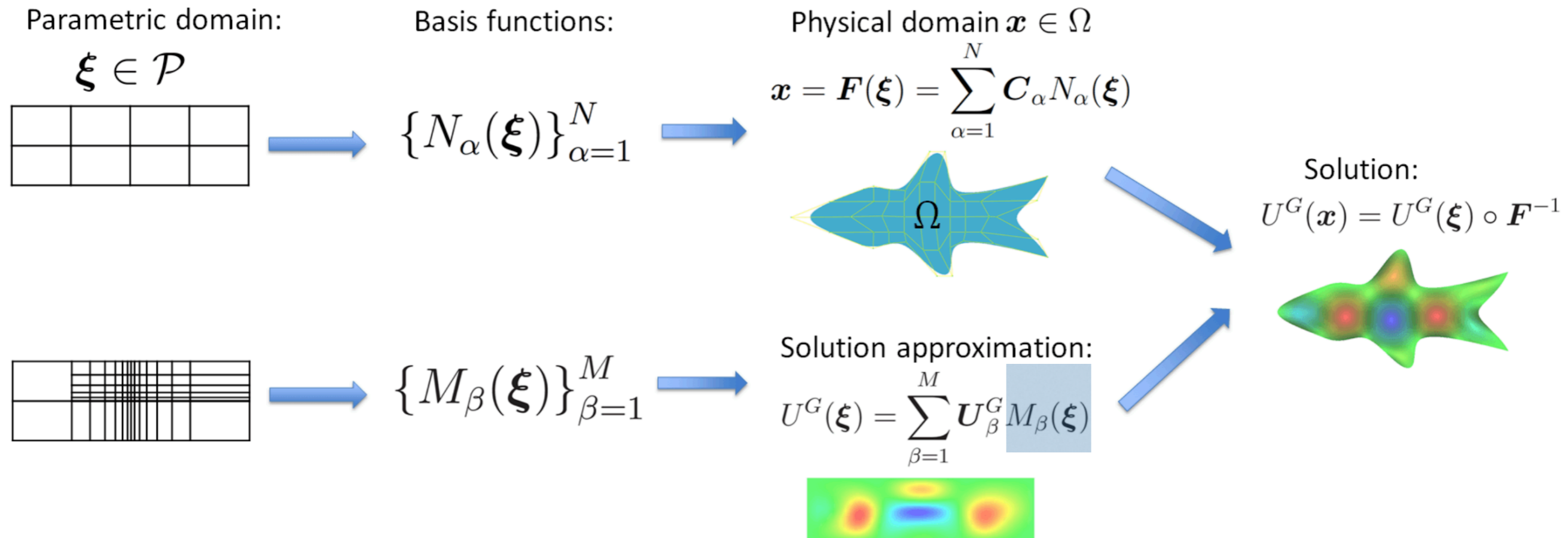
# GIFT



# IGA



# GIFT



Geometry parameterization	Solution basis	Degree parity	Patch test	Optimal convergence
$Q_0$	$A_1$	Iso-geometric	✓	✓
$Q_0$	$A_2$	Super-geometric	✓	✓
$Q_0$	$C_1$	Iso-geometric	×	✓
$Q_0$	$C_2$	Super-geometric	×	✓
$A_1$	$A_1$	Iso-geometric	✓	✓
$A_1$	$A_2$	Super-geometric	✓	✓
$A_2$	$A_1$	Sub-geometric	✓	✓
$B_1$	$A_1$	Iso-geometric	✓	✓
$B_1$	$A_2$	Super-geometric	✓	✓
$B_2$	$A_1$	Sub-geometric	✓	✓
$C_1$	$C_1$	Iso-geometric	✓	✓
$C_1$	$C_2$	Super-geometric	✓	✓
$C_2$	$C_1$	Sub-geometric	✓	✓
$C_1$	$A_1$	Iso-geometric	×	✓
$C_1$	$A_2$	Super-geometric	×	✓
$C_2$	$A_1$	Sub-geometric	×	✓
$A_1$	$D_1$	Iso-geometric	×	✓
$A_1$	$D_2$	Super-geometric	×	✓
$A_1$	$D_0$	Sub-geometric	×	✓

# GIFT - key features

- ☑ The same basis functions, which are used in CAD to represent the geometry, are used in the IGA as shape functions to approximate the unknown solution
- ☑ Geometry is exact at any stage of the solution refinement process
- ☑ Better accuracy per DOF in comparison with Lagrange Iso-parametric FEM but higher computational cost (bandwidth...)



# Numerical observations - no proof...

Together with the given (exact) geometry parametrization at the coarsest level, the convergence rate is entirely defined by the solution basis, and does not depend on the further refinement of the geometry parametrization:

- For a given geometry parameterization, the degree of the solution basis can be increased or decreased without changing the degree of the geometry (from iso-geometric to super-geometric and sub-geometric elements)
- For solution approximation, using same degree B-Splines or NURBS yields almost identical results

# Adaptive/enriched GIFT - FEM/BEM

## Adaptive GIFT for Helmholtz

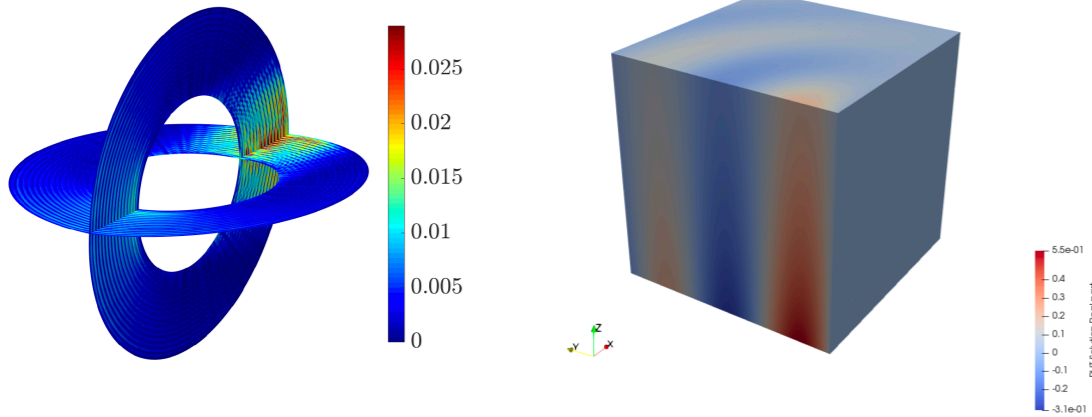
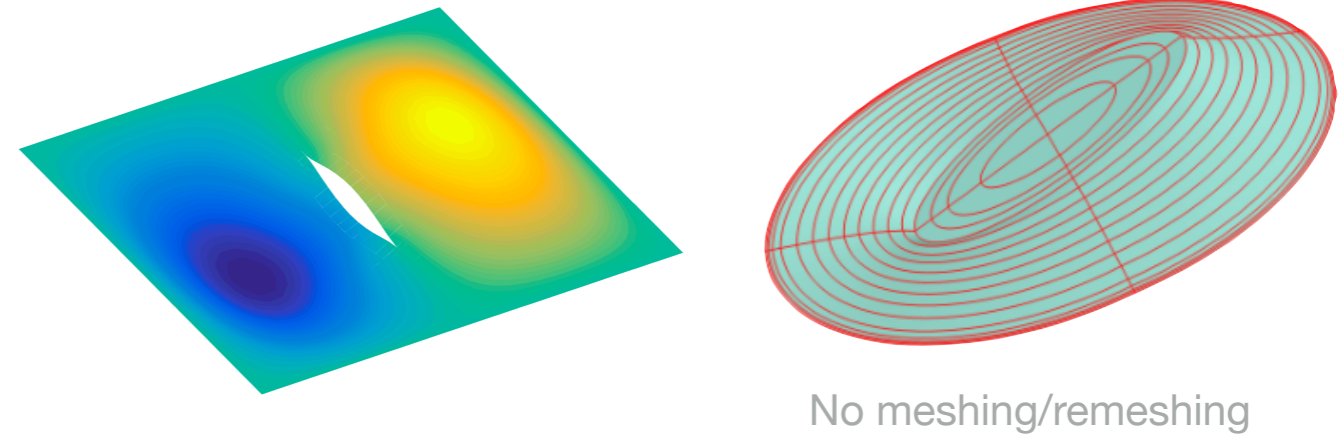


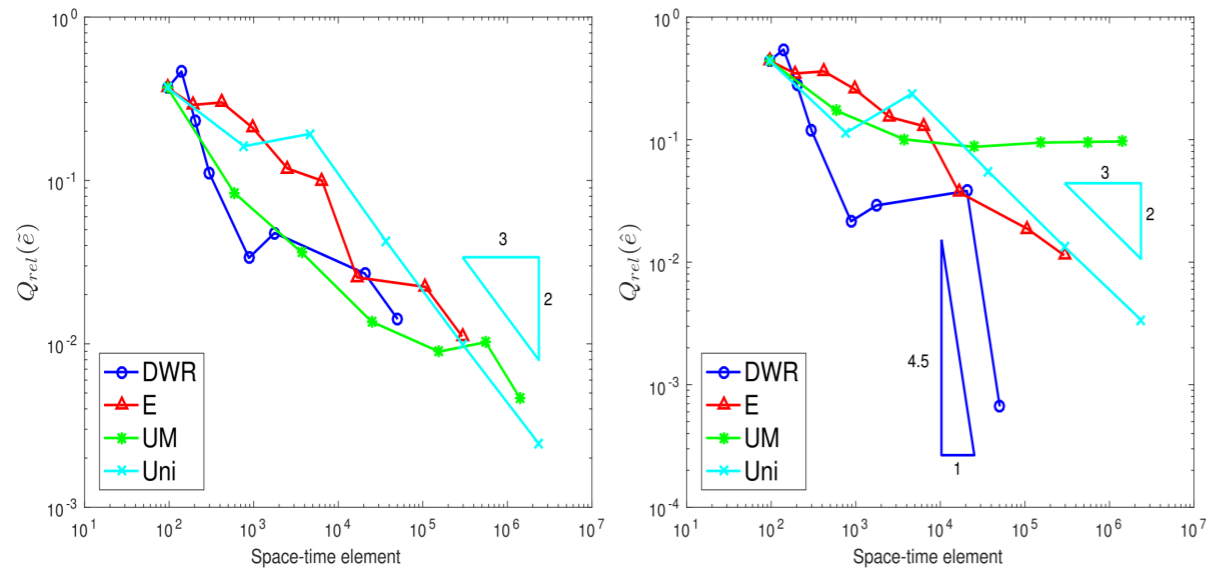
Figure 21: The unit cube problem: numerical solution for  $k = 10$  and  $n = 1$ .

## Crack growth

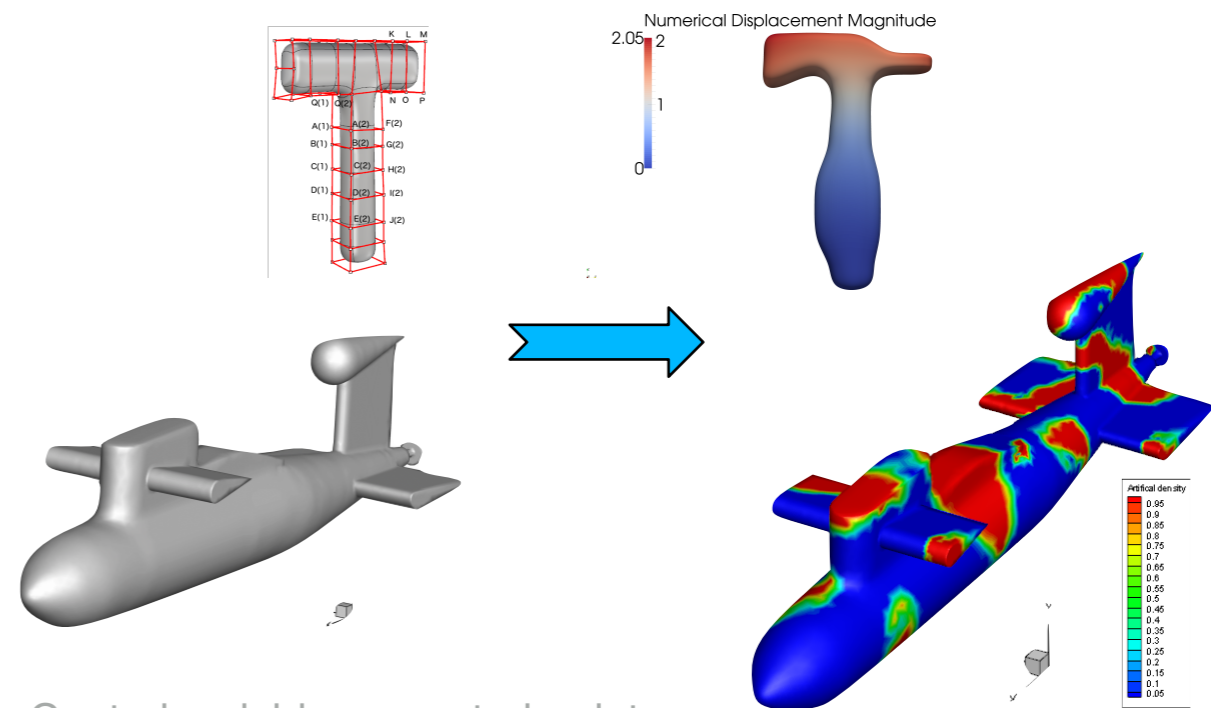


## Space-time adaptivity

$$Q(u) = \int_0^T \int_{\partial\Omega_Q} u e_x w(t) d\Gamma dt$$



## Shape optimisation elasticity & acoustics



Control variables = control points

# Adaptive GIFT-Helmholtz

- Residual-based and recovery based error indicators perform similarly
- Optimal convergence rates
- Adaptivity is effective only for “fine meshes”
- Failure in the “pre-asymptotic range” where effectivity per dof deteriorates

$$\frac{kh}{p} < 1$$



**h- and p- adaptivity driven by recovery and residual-based error estimators for PHT-splines applied to time-harmonic acoustics** Videla, Anitescu, Khajah, Bordas, Atroshchenko, 2019



**Plane-wave enriched Partition of Unity GIFT for Helmholtz equation with PHT-Splines** Videla, Tomar, Bordas, Atroshchenko, 2019

# Structural optimisation with IGABEM

Employ the same basis functions in CAD to discretize Boundary Integral Equations (BIE) :

$$C_{ij}(\mathbf{s})u_j(\mathbf{s}) + \int_S T_{ij}(\mathbf{s}, \mathbf{x})u_j(\mathbf{x})dS(\mathbf{x}) = \int_S U_{ij}(\mathbf{s}, \mathbf{x})t_j(\mathbf{x})dS(\mathbf{x})$$

1. Seamlessly compatible with CAD due to boundary representation.
2. CAD in and CAD out. (Fig. 2)
3. Infinite domain: acoustics, electro-magnetics.

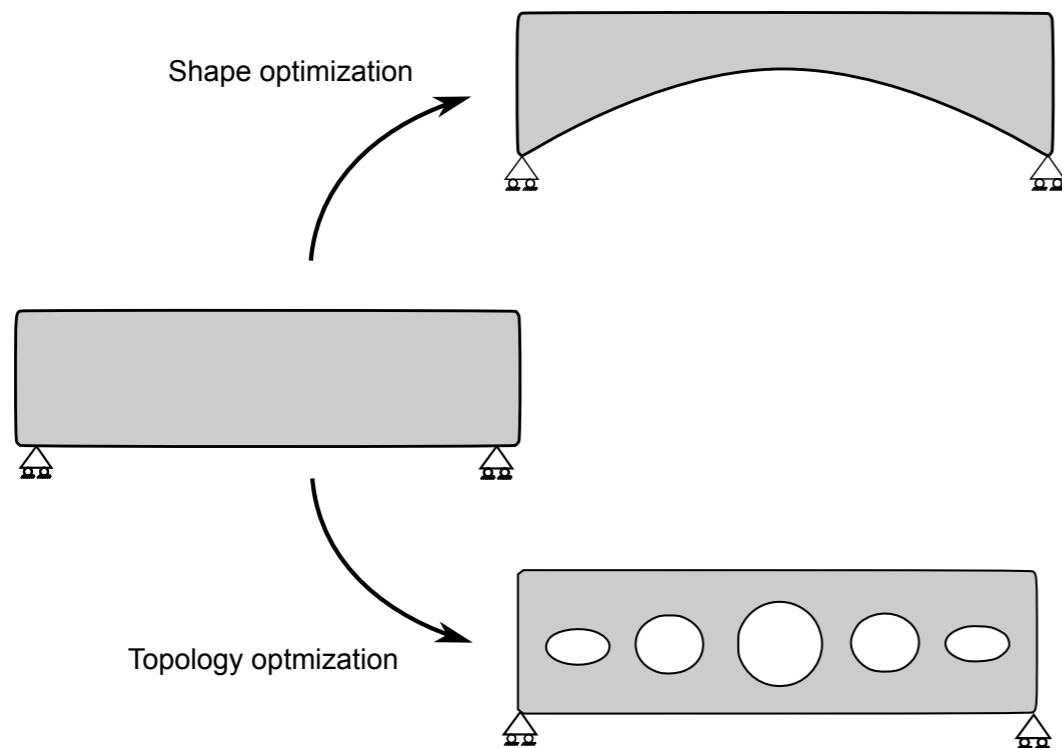


Fig. 1: Structural optimisation

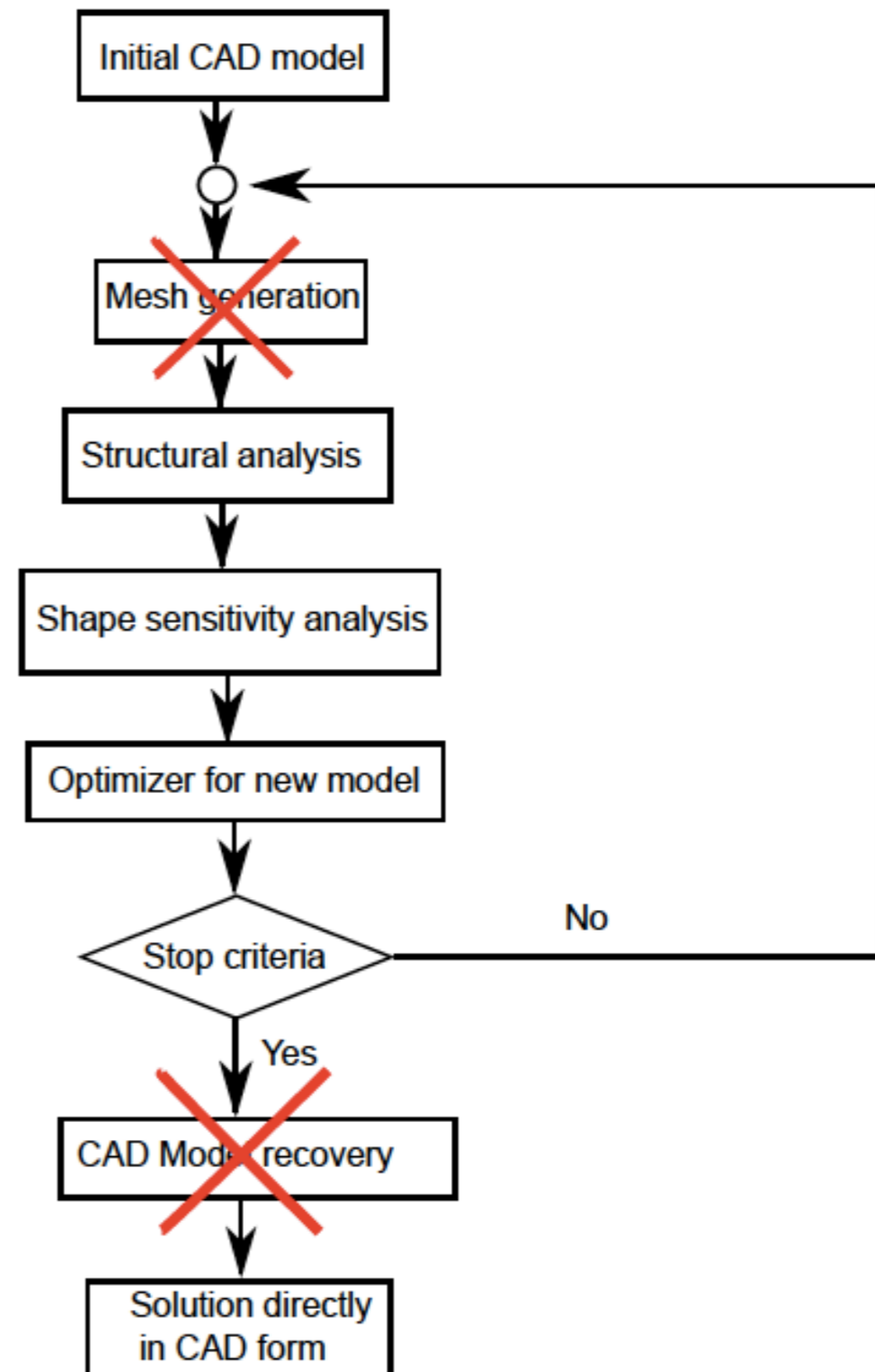


Fig. 2: IGABEM structural optimisation



# Elastic shape optimisation using IGABEM with T-splines

**CAD in & CAD out!**

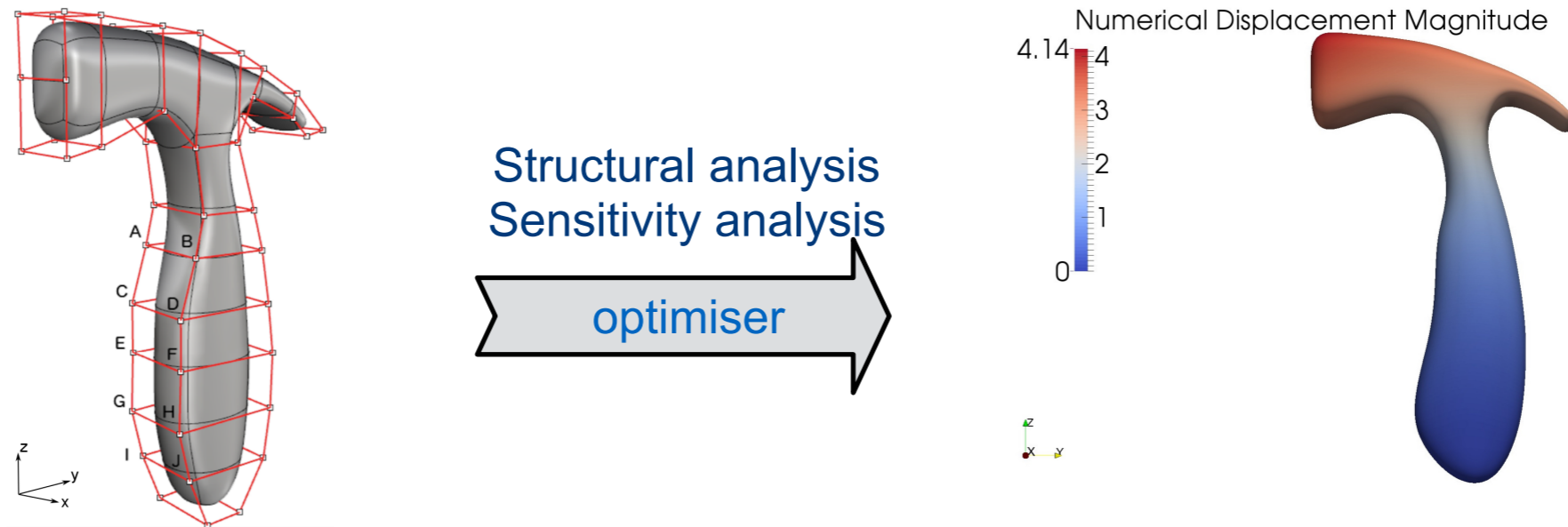


Fig. 1: T-spline model of a hammer (left: initial geometry; right: optimised geometry)

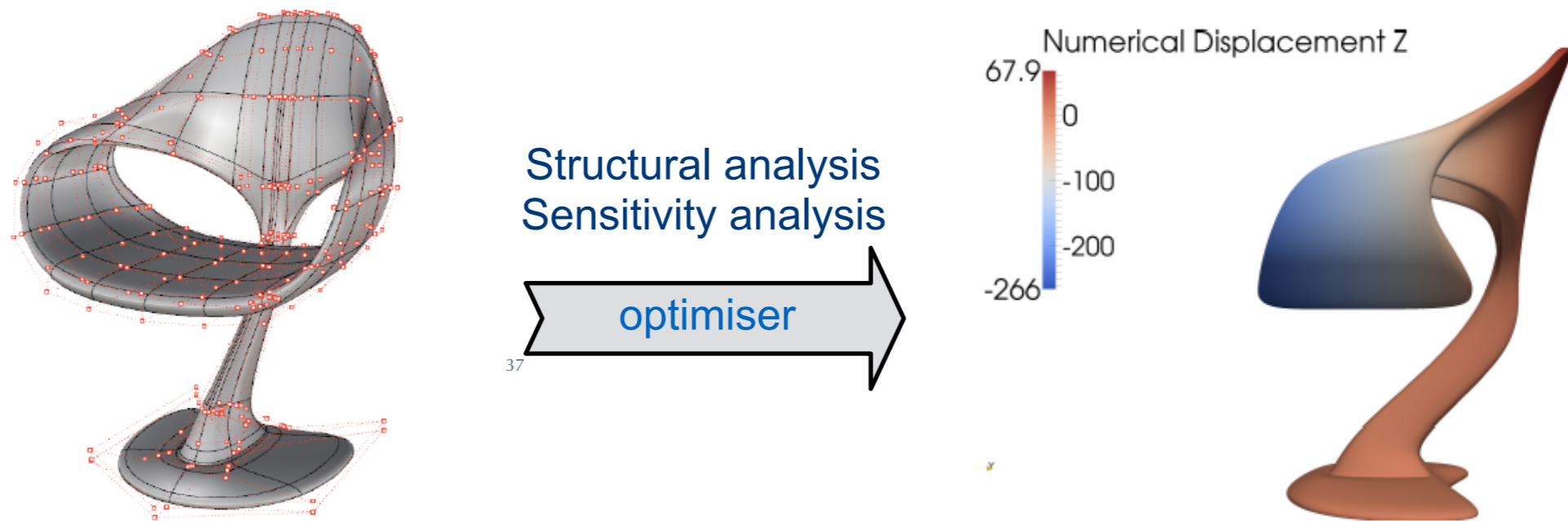


Fig. 2: T-spline model of a chair (left: initial geometry; right: optimised geometry)



# Acoustic shape optimisation using IGABEM with NURBS

*Exterior infinite domain problems!*

$$\begin{cases} \min & \Pi_\rho = \bar{\mathbf{P}}_f \mathbf{P}_f \\ \text{s.t.} & A(\mathbf{x}) - A_0 \leq 0 \\ & x^{i,l} \leq x^i \leq x^{i,u}, \quad i = 1, 2, \dots, N_x, \end{cases}$$

$$\frac{\partial \Pi_\rho}{\partial \vartheta} = \frac{\partial (\bar{\mathbf{P}}_f \mathbf{P}_f)}{\partial \vartheta} = 2\Re \left( \bar{\mathbf{P}}_f \frac{\partial \mathbf{P}_f}{\partial \vartheta} \right)$$

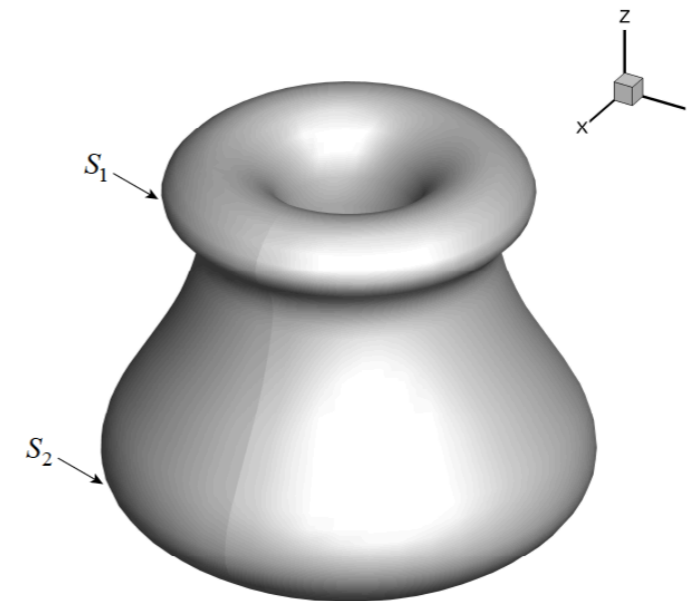
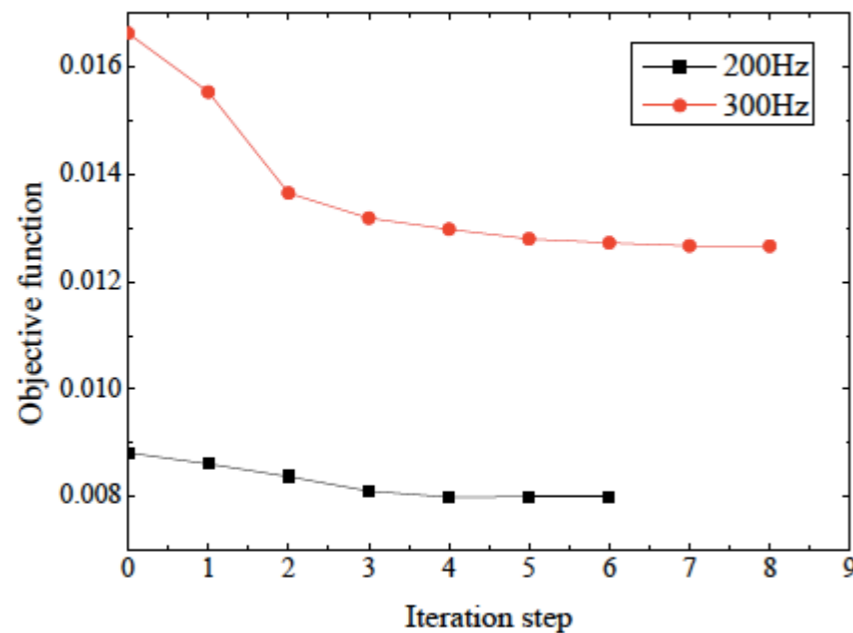


Fig. 1: Vase model with NURBS



38

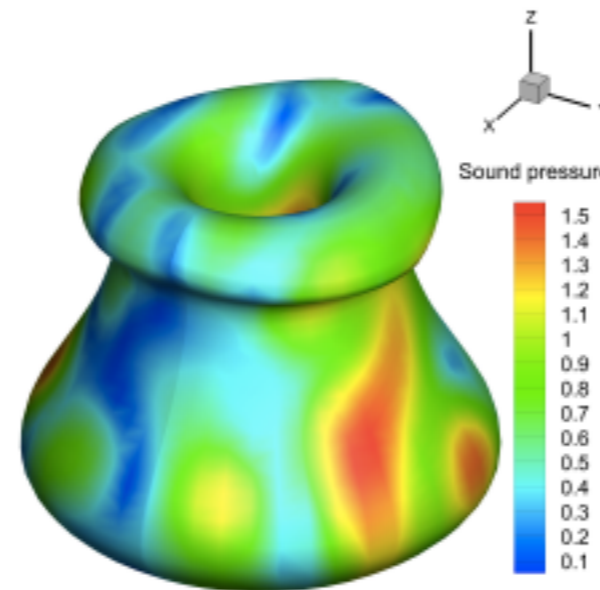


Fig. 2: Optimised model (f=100Hz)

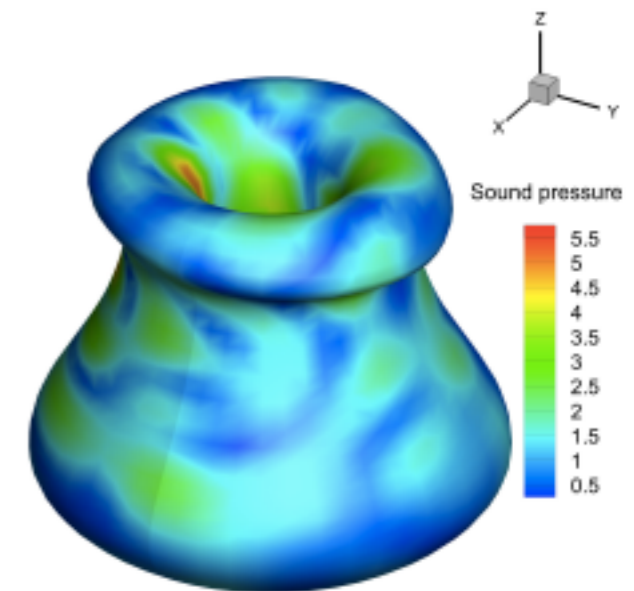


Fig. 3: Optimised model (f=300Hz)



# Acoustic topology optimisation using IGABEM with subdivision surfaces

*Exterior infinite domain problems!*

$$\begin{cases} \min & \Pi_\rho = \bar{\mathbf{P}}_f \mathbf{P}_f \\ \text{s.t.} & \sum_{e=1}^{N_e} \rho_e v_e - V_0 \leq 0 \\ & 0 \leq \rho_{\min} \leq \rho_e \leq 1 \end{cases}$$

$$\beta_i = \beta_0 \rho_i^n + \beta_1 (1 - \rho_i^n), \quad 0 \leq \rho_i \leq 1$$

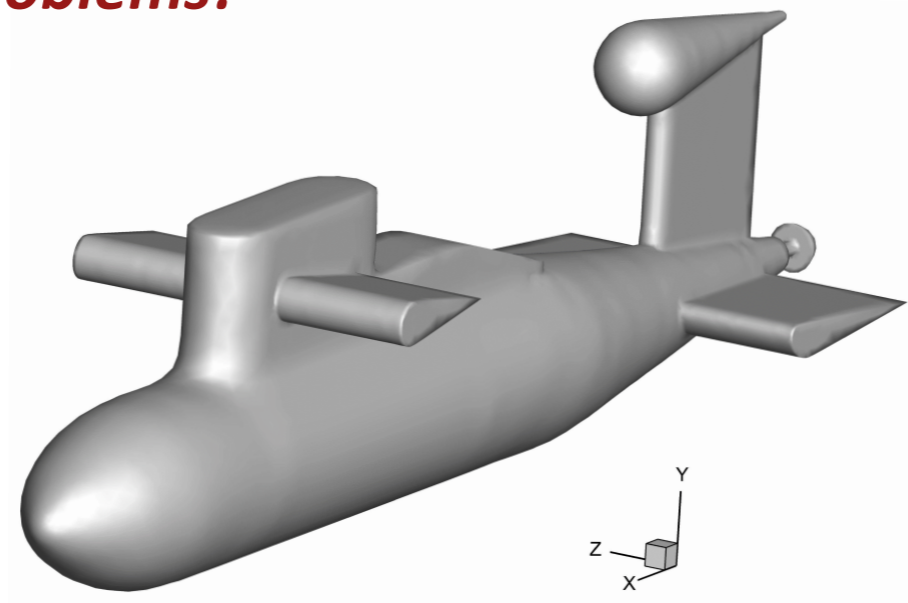


Fig. 1: Submarine model

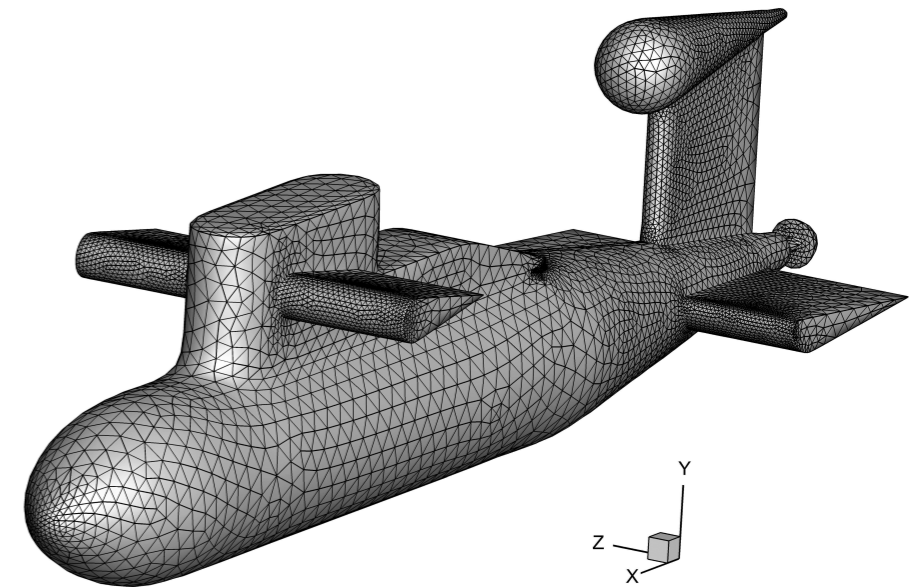


Fig. 2: Model with subdivision surfaces

... problems appear naturally

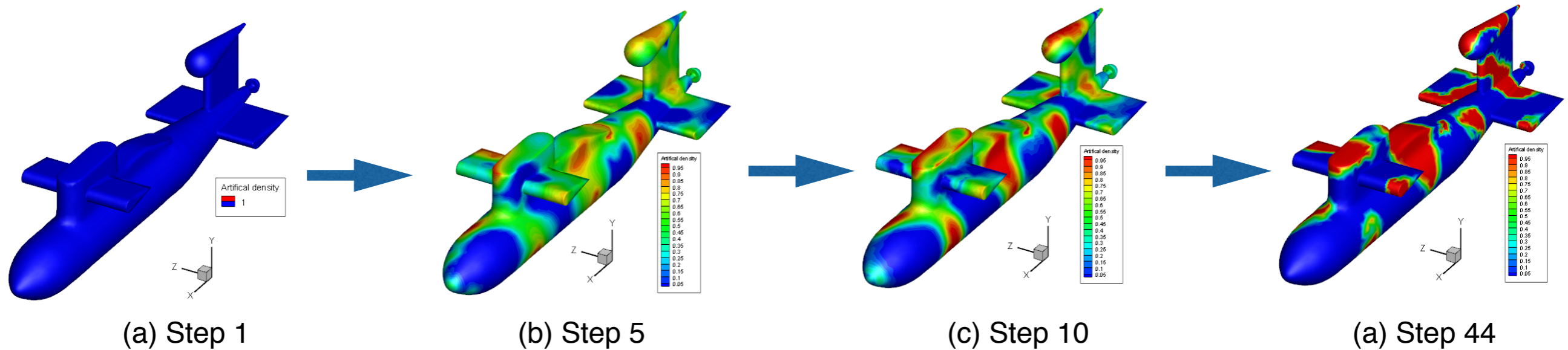


Fig. 3: Sound absorbing material distribution during iteration



# Acoustic topology optimisation using IGABEM with subdivision surfaces

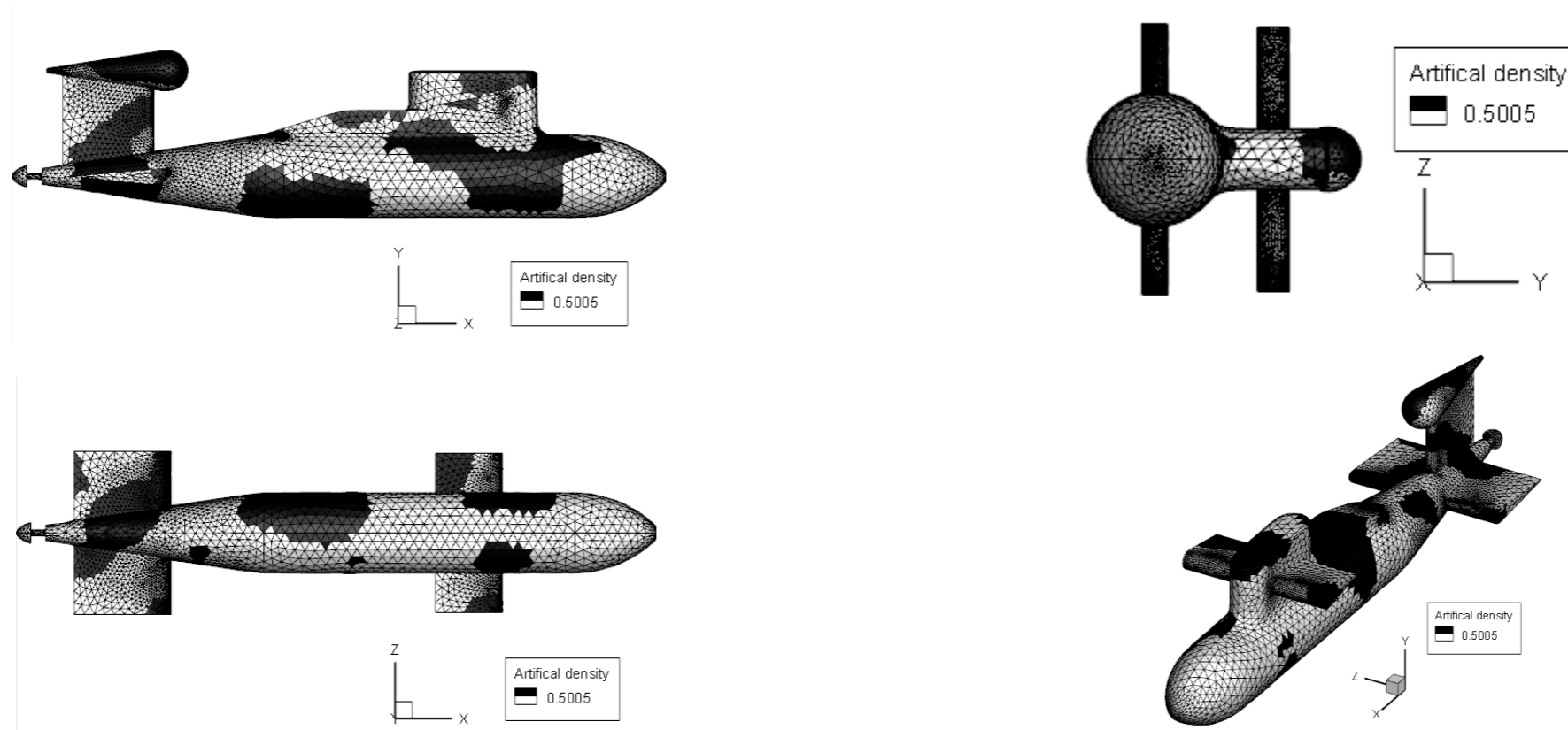


Fig. 1: The optimised sound adsorbing material distribution (black: adhesion to sound absorbing materials; white: no adhesion)

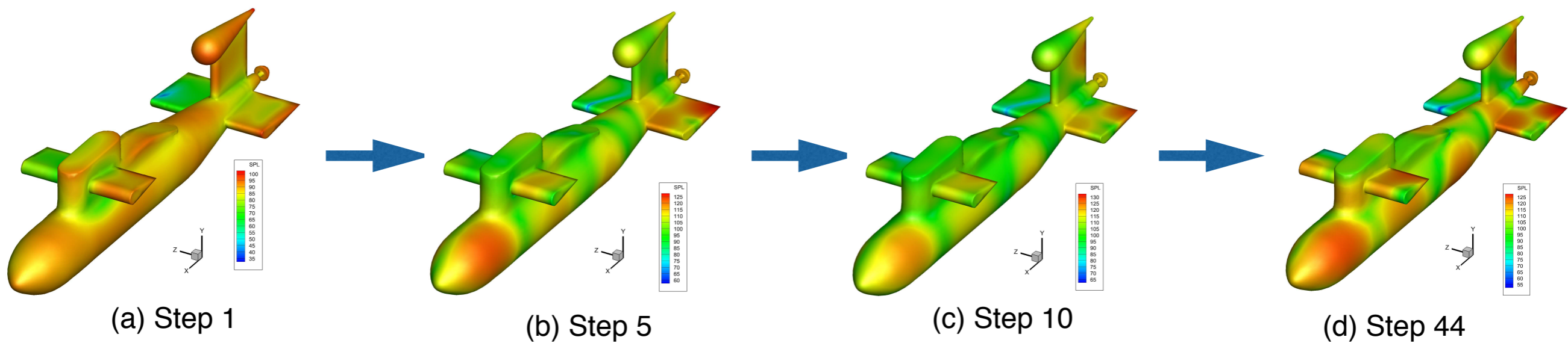


Fig. 2: The sound pressure distribution during Iteration





# GIFT - key features

- ☑ Retain advantages of IGA, but **decouple** the geometry and field approximation
- ☑ Standard patch tests may not always pass, yet the **convergence rates are optimal** as long as the geometry is exactly represented by the geometry basis
- ☑ With geometry exactly represented by NURBS, using same degree B-splines or NURBS for the approximation of the solution field yields almost identical results
- ☑ With geometry exactly represented by NURBS, using **PHT splines** for the approximation of the solution gives additional advantage of local adaptive refinement
- ☑ Any other approximation field can be used for the field variables

# Open source software

**Bauhaus Universität**

**<https://github.com/canitevc/IGAPack>**

**University of Luxembourg**

**<http://www.legato-team.eu>**

**FEniCS**



FENICS  
PROJECT

```
mu = Constant(0.3)
```

```
...
```

```
S = (t**3/12) * (2.0*mu*K + \  
    (2.0*mu*lmbda) / (2.0*mu + lmbda) *tr(K) *Identity(2))
```

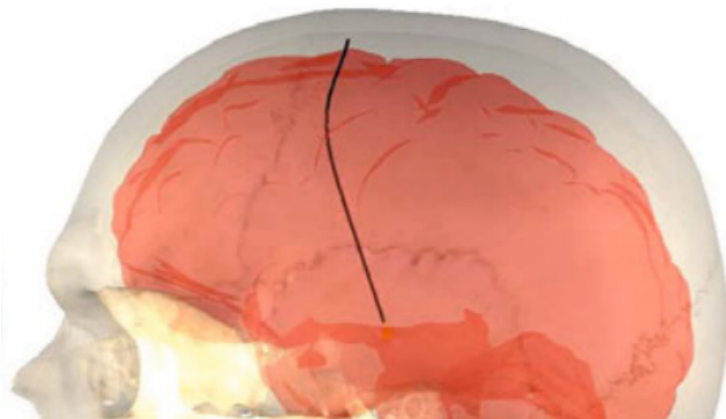


DON'T YOU  
THINK  
IF I WERE  
WRONG  
I'D KNOW IT?

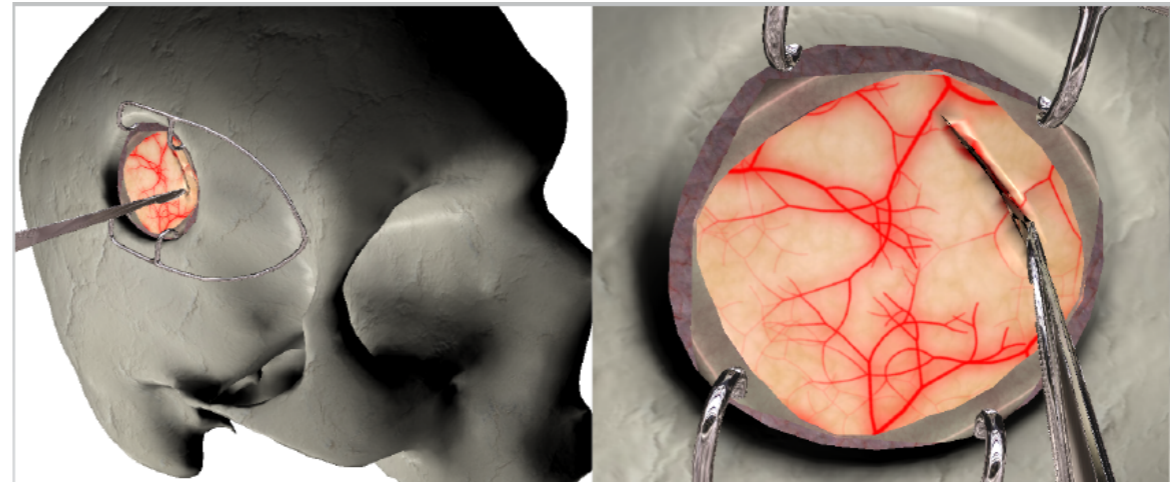
*In coupling geometry and field variables*

-DR. SHELDON LEE COOPER  
B.S., M.S., M.A., PH.D., SC.D.

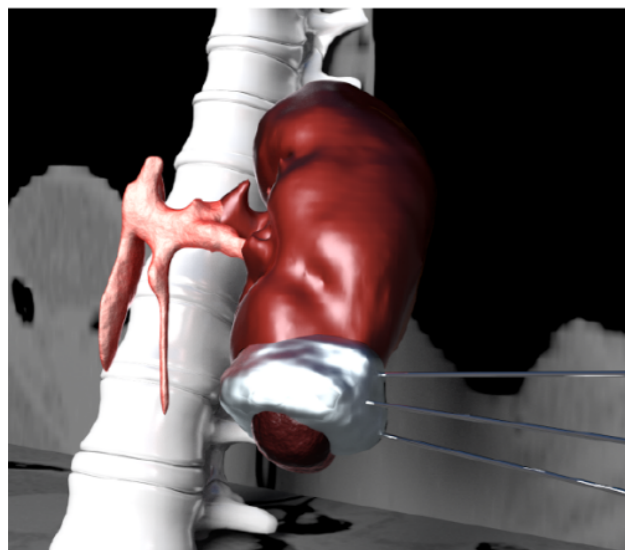
# Real-time simulations with XFEM



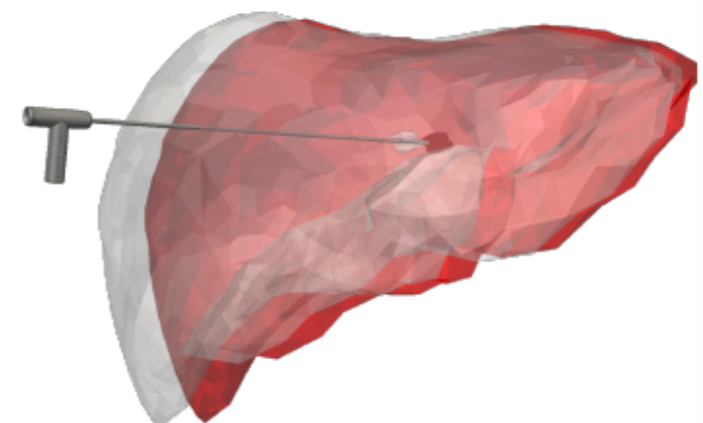
Bilger et al, MICCAI, 2011



Courtecuisse et al, Med. Image Anal., 2014



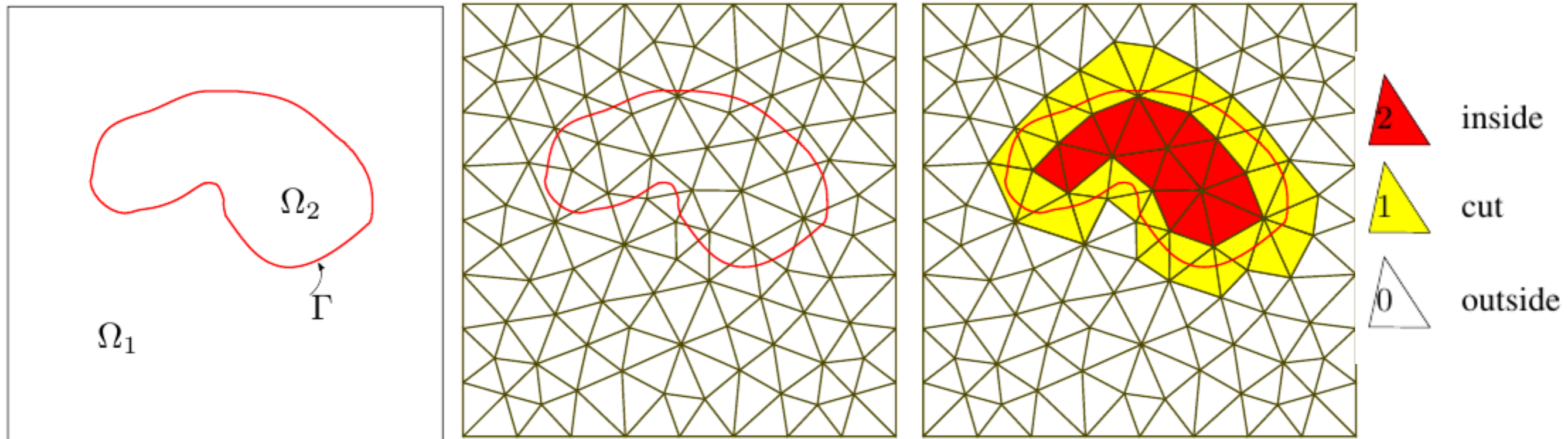
Talbot et al, SIGGRAPH, 2015



Hamzé et al, Comput. Med. Imag. Grap. 2015



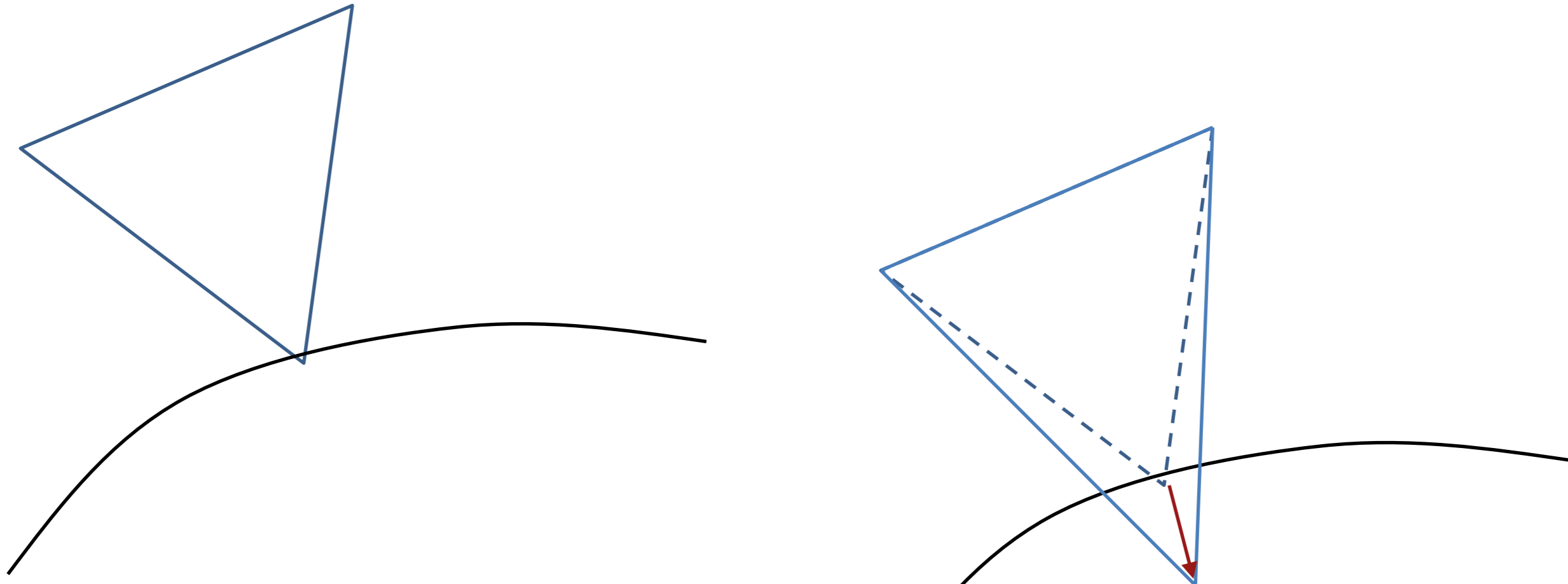
# Cut FE method/XFEM



Fictitious **Boundary Method**

Implicit (unfitted) **Interface**: interface problems, moving boundary problems

# Limit cases



Alternative stabilization approaches:

- ❑ Agathos, Chatzi, Bordas, 2017, 2018, 2019
- ❑ Ghost penalty, Burman *et al*, 2015
- ❑ Stable XFEM, Gupta, Banerjee, Babuška, Duarte, 2013
- ❑ Neighboring gradient, Haslinger *et al*, 2009
- ❑ Menk, Bordas, 2009, Béchet, Moës, 2008

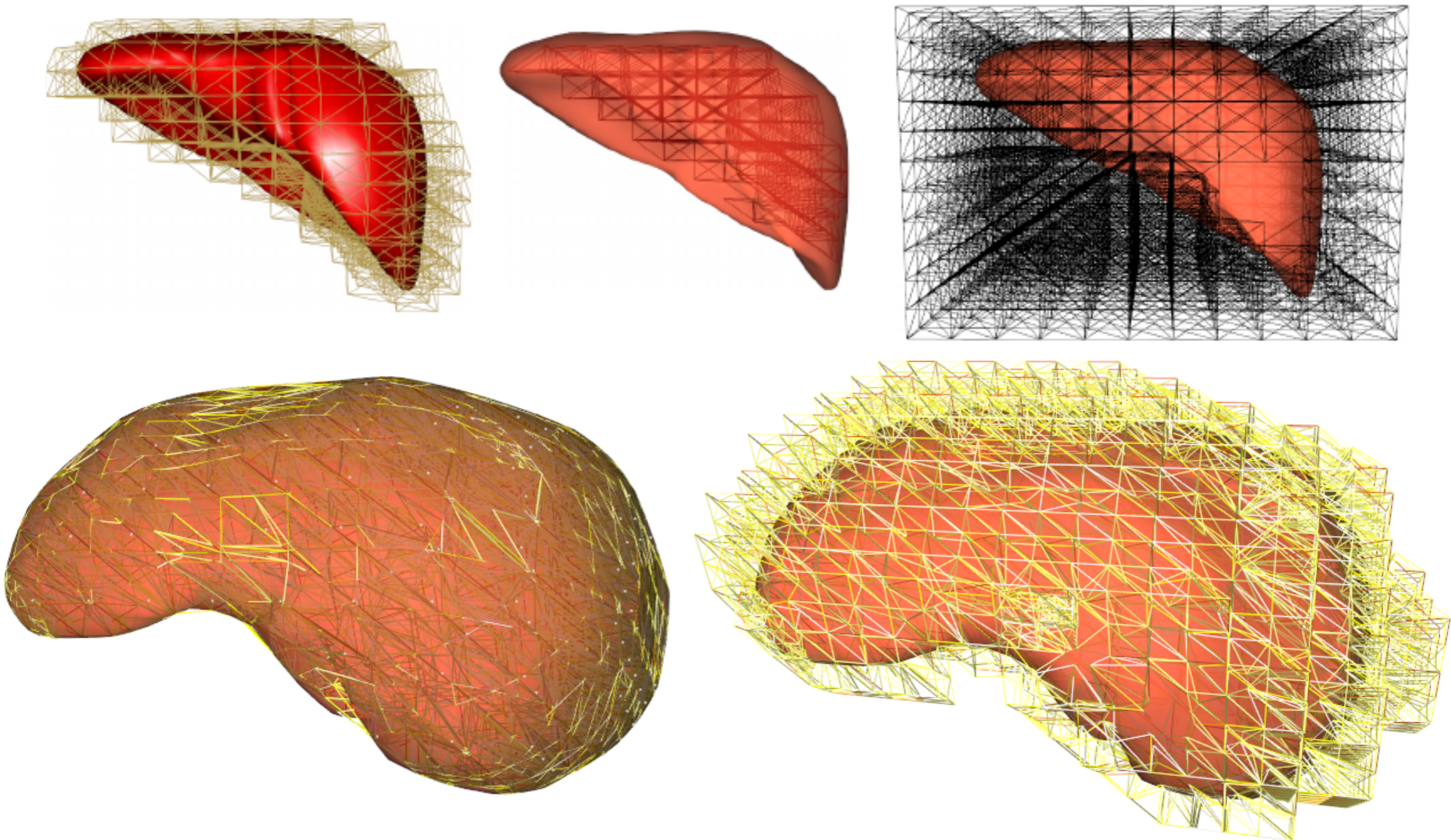
Real-time Error Control for Surgical Simulation, HP Bui et al, IEEE Trans. Biomed. Eng., 2016.

Stéphane Pierre Alain BORDAS, Department of Computational Engineering & Sciences University of Luxembourg



RealTCut

# Implicit boundaries for liver



Real-time Error Control for Surgical Simulation, HP Bui et al, IEEE Trans. Biomed. Eng., 2016.

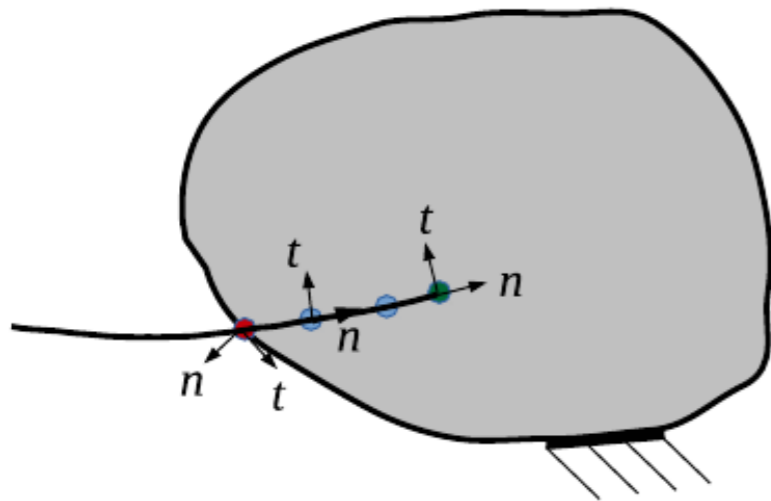
Stéphane Pierre Alain BORDAS, Department of Computational Engineering & Sciences University of Luxembourg



RealTCut

# Models: Needle insertion simulation

Needle insertion problem



$$\operatorname{div} \boldsymbol{\sigma} + \mathbf{b} + \boldsymbol{\lambda} = \rho \ddot{\mathbf{u}} \quad \text{in } \Omega$$

$$\boldsymbol{\epsilon} = \frac{1}{2} (\operatorname{grad} \mathbf{u} + (\operatorname{grad} \mathbf{u})^T)$$

$$\boldsymbol{\sigma} = f(\boldsymbol{\epsilon}, \nu)$$

$$\boldsymbol{\sigma} \cdot \mathbf{n} = \bar{\mathbf{t}} \text{ on } \Gamma_t;$$

$$\mathbf{u} = \bar{\mathbf{u}} \text{ on } \Gamma_u,$$

$$\lambda_n^{ts} > \lambda_{p0}, \quad (\text{penetrate})$$

$$\lambda_t^{ts} < \mu \lambda_n^{ts} \quad (\text{stick}); \quad \lambda_t^{ts} = \mu \lambda_n^{ts} \quad (\text{slip}),$$

$$\mathbf{b} = \bar{\mathbf{b}} - c \dot{\mathbf{u}}$$

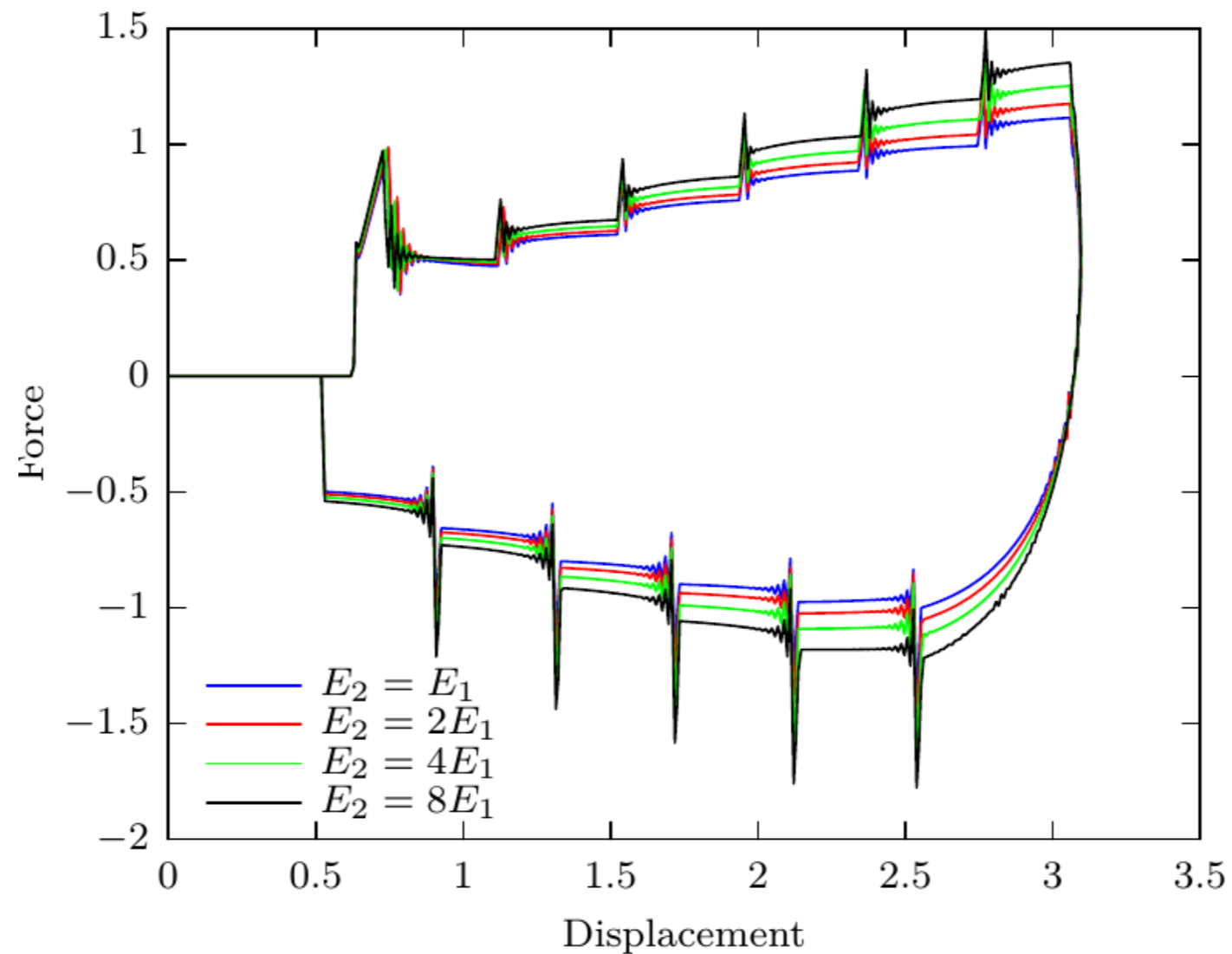
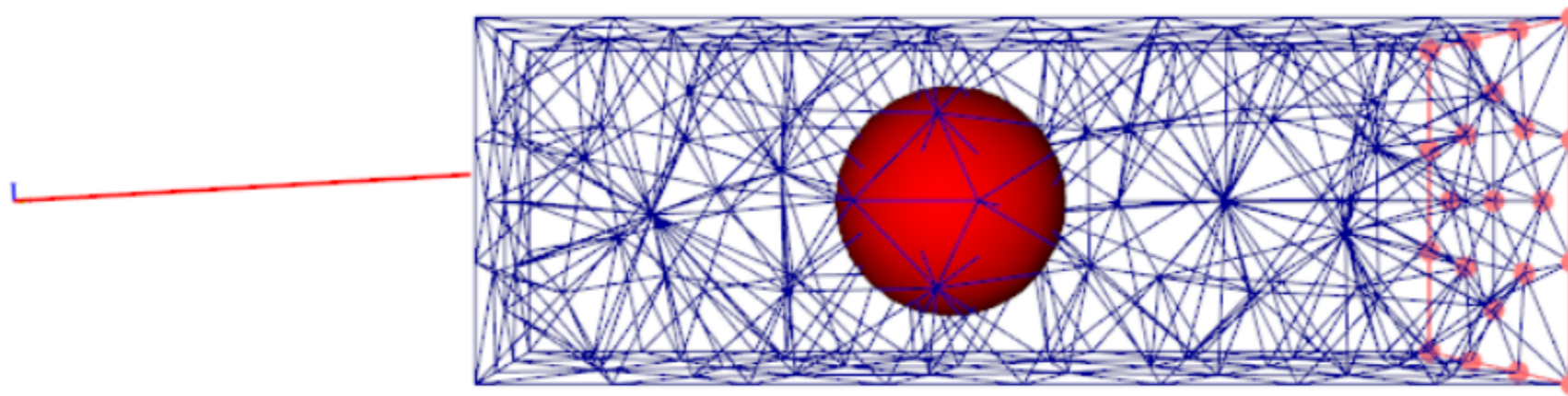
$$\lambda_n^{nt} < \mu \lambda_t^{nt} + \lambda_{c0} \quad (\text{stick}); \quad \lambda_n^{nt} \geq \mu \lambda_t^{nt} + \lambda_{c0} \quad (\text{cut and slip}),$$

$$\lambda_n^{ns} < \mu \lambda_t^{ns} \quad (\text{stick}); \quad \lambda_n^{ns} = \mu \lambda_t^{ns} \quad (\text{slide}).$$

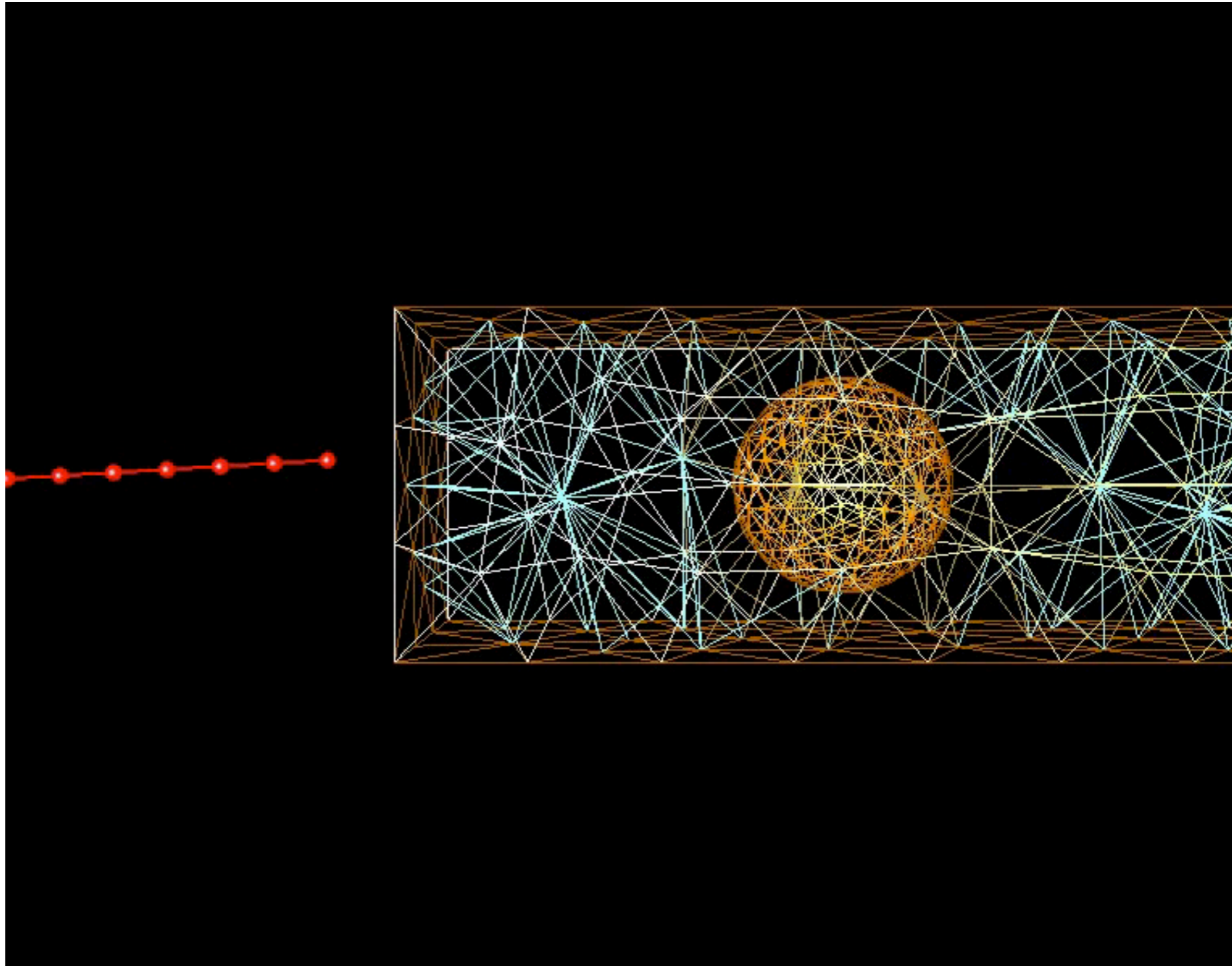




# Needle Insertion with Implicit Interface



# Needle Insertion with Implicit Boundary



Real-time Error Control for Surgical Simulation, HP Bui et al, IEEE Trans. Biomed. Eng., 2016.

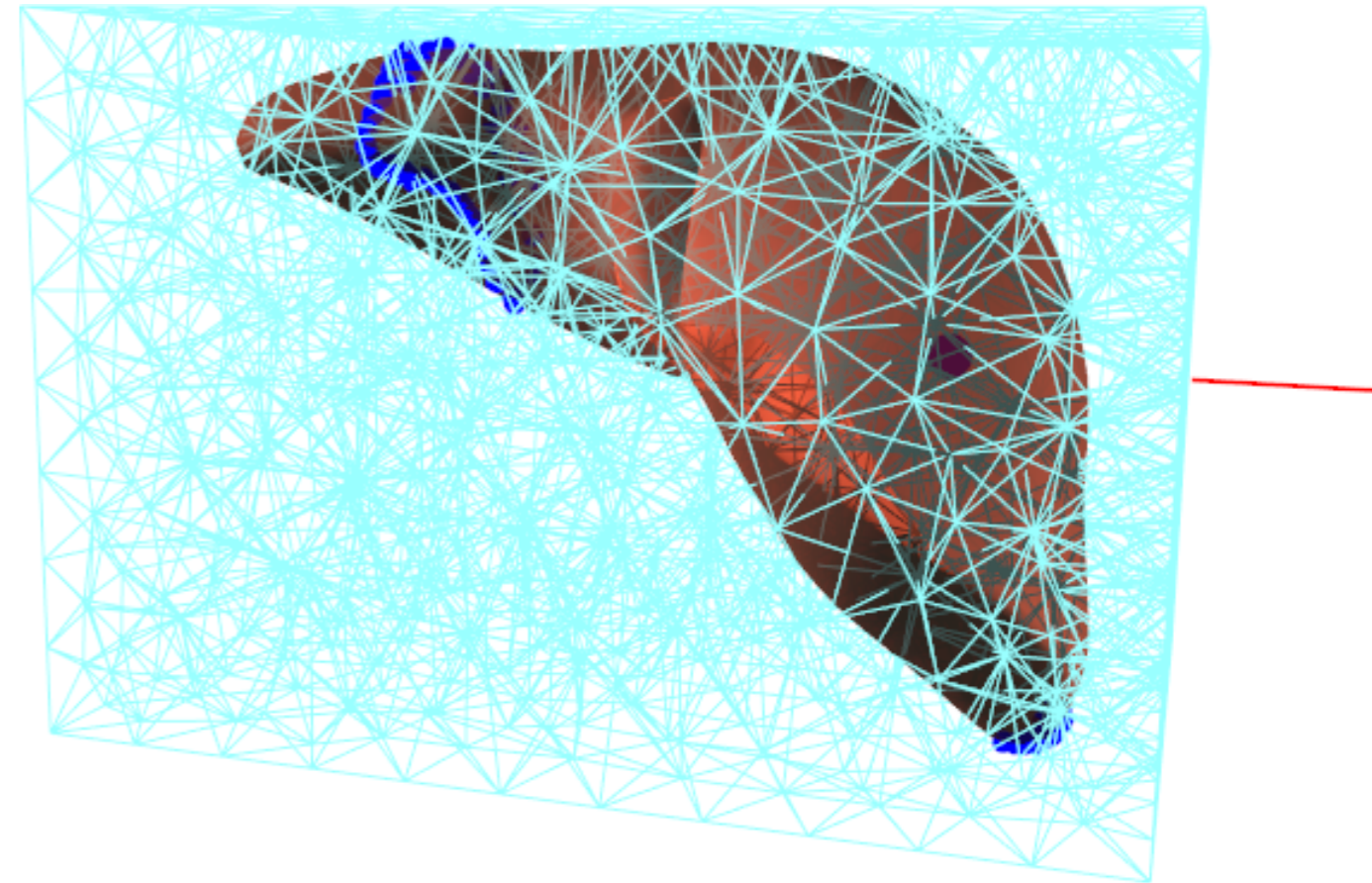
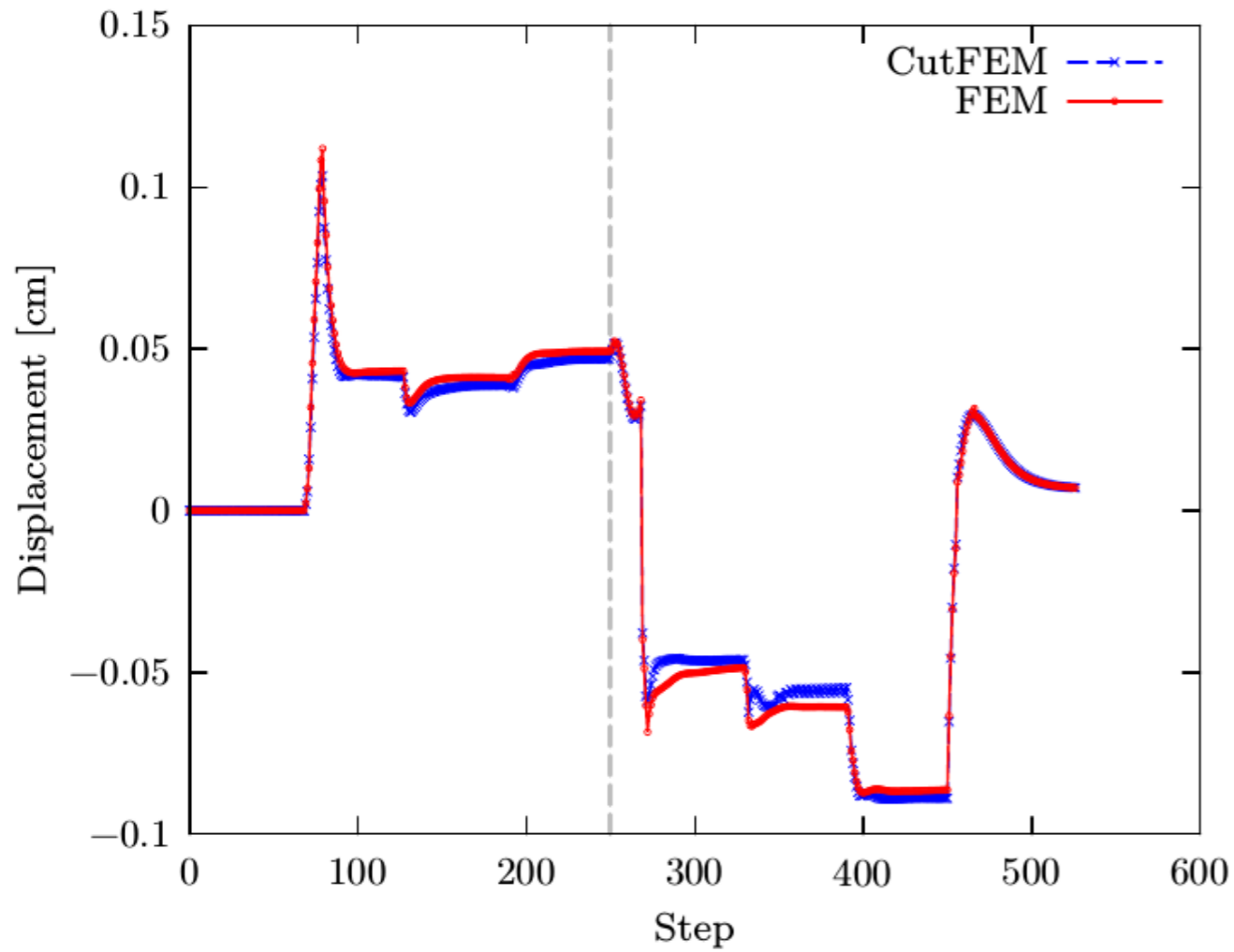
Stéphane Pierre Alain BORDAS, Department of Computational Engineering & Sciences University of Luxembourg



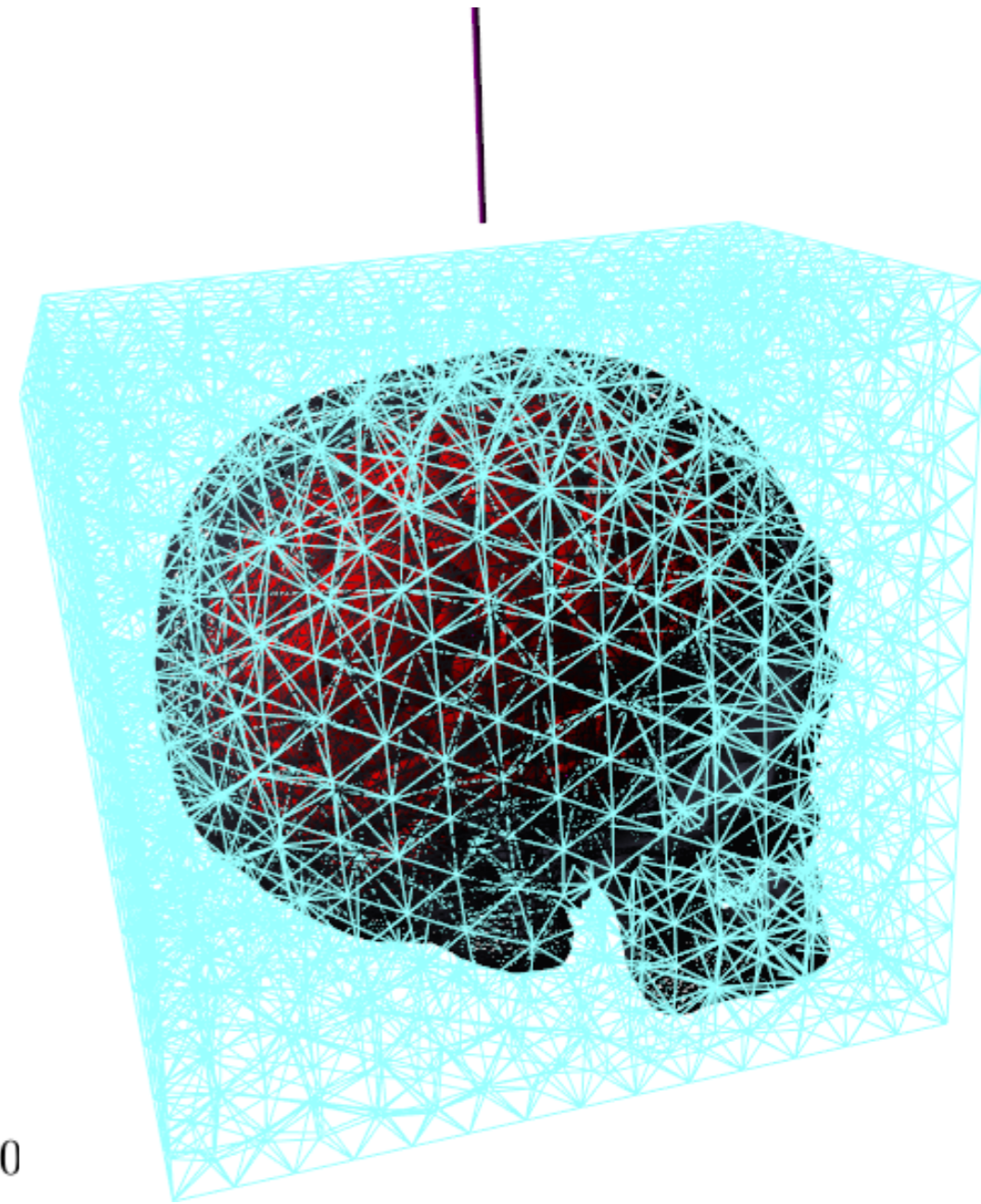
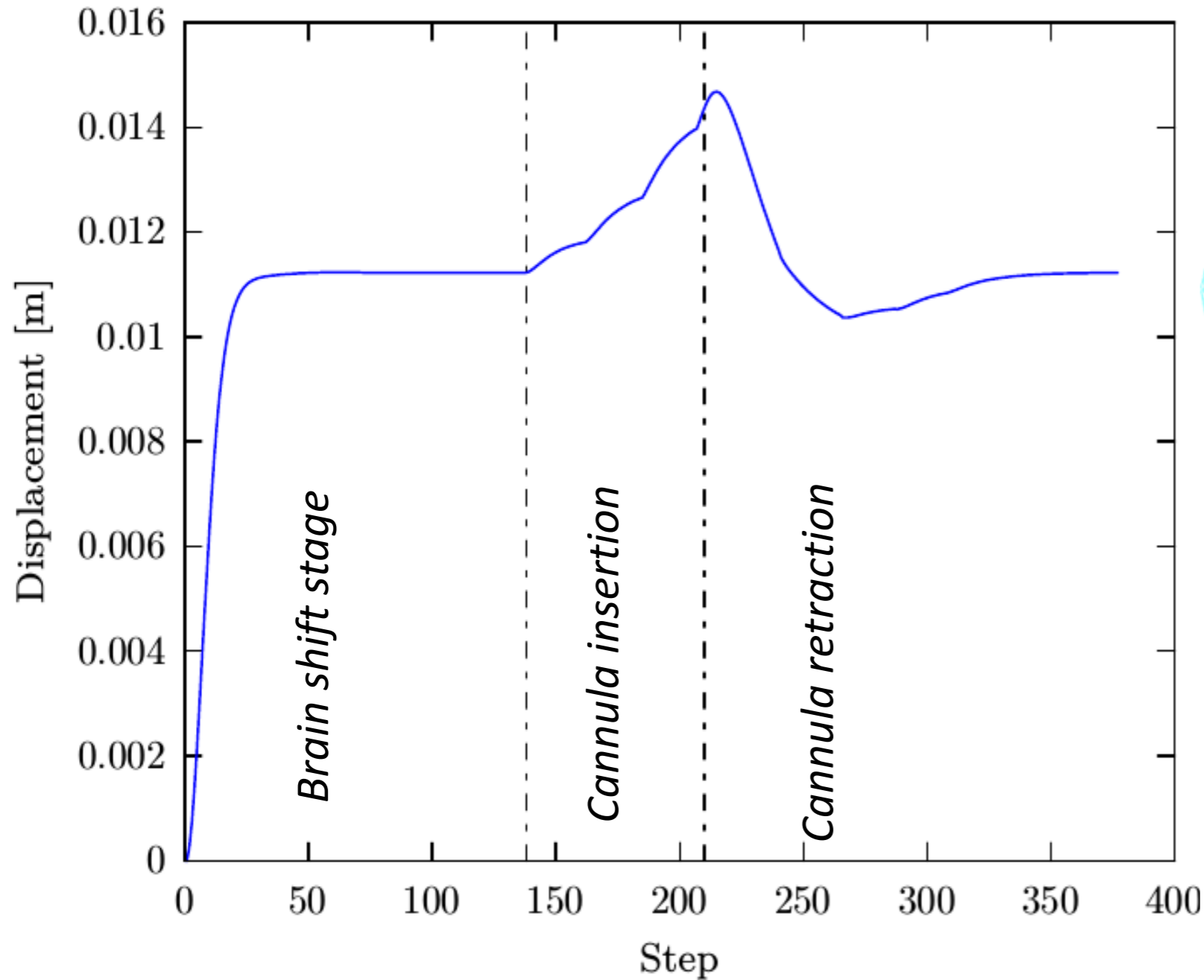
erc

RealTCut

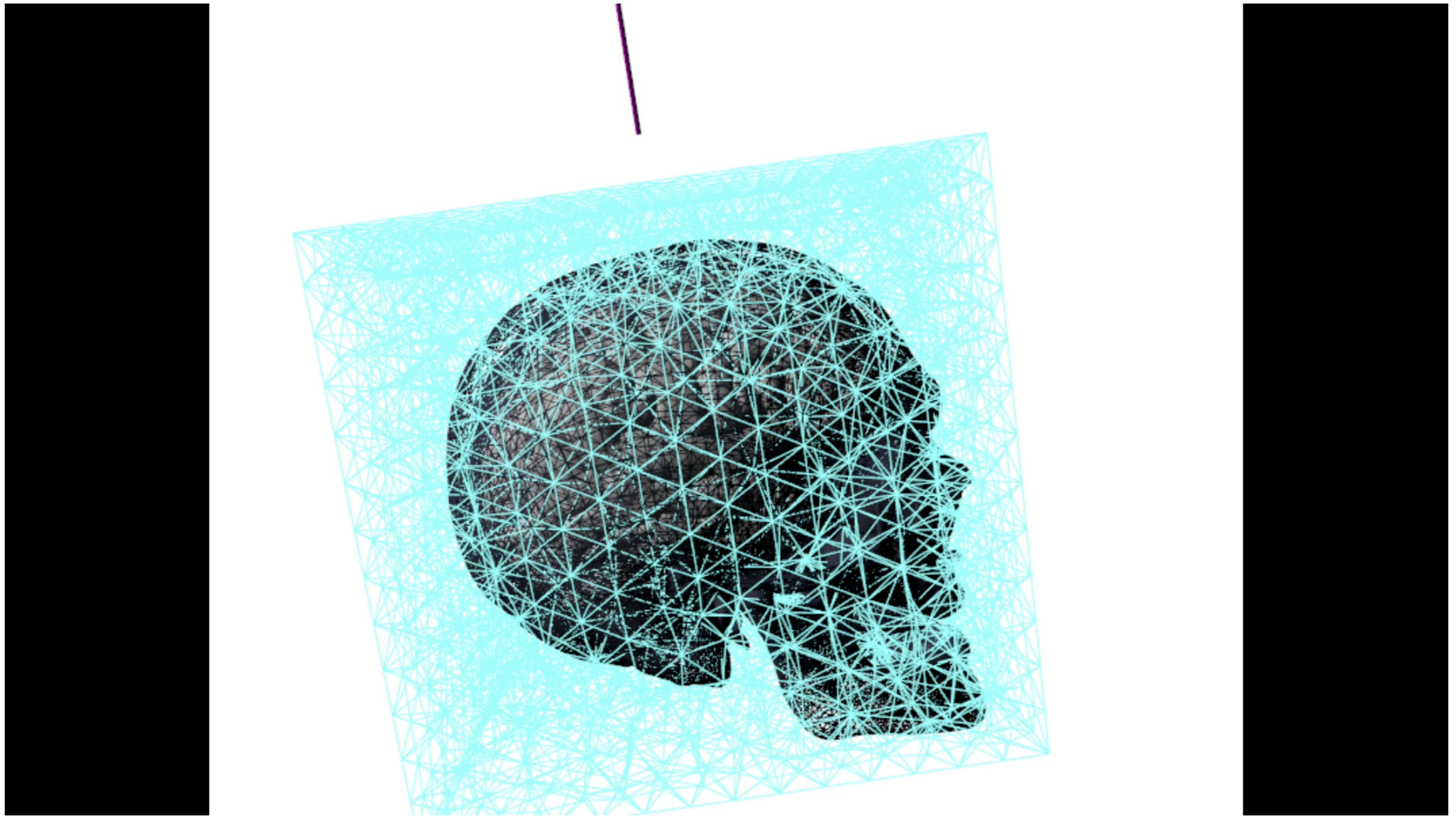
# Needle insertion into liver



# Brain shift and electrode implantation



# Brain shift and electrode implantation



Controlling the Error on Target Motion through Real-time Mesh Adaptation: Applications to Deep Brain Stimulation, HP Bui et al, Int J Numer Meth Bio, 2017.



RealTCut

# Error estimation and adaptivity

Controlling the Error on Target Motion through Real-time Mesh Adaptation: Applications to Deep Brain Stimulation, HP Bui et al, Int J Numer Meth Bio, 2017.



RealTCut

# Superconvergence recovery

## Controlling the Error on Target Motion through Real-time Mesh Adaptation: Application to Deep Brain Stimulation

H. P. Bui, S. Tomar, H. Courtecuisse, M. Audette, S. Cotin and S. P. A. Bordas

Controlling the Error on Target Motion through Real-time Mesh Adaptation: Applications to Deep Brain Stimulation, HP Bui et al, Int J Numer Meth Bio, 2017.



RealTCut



# References

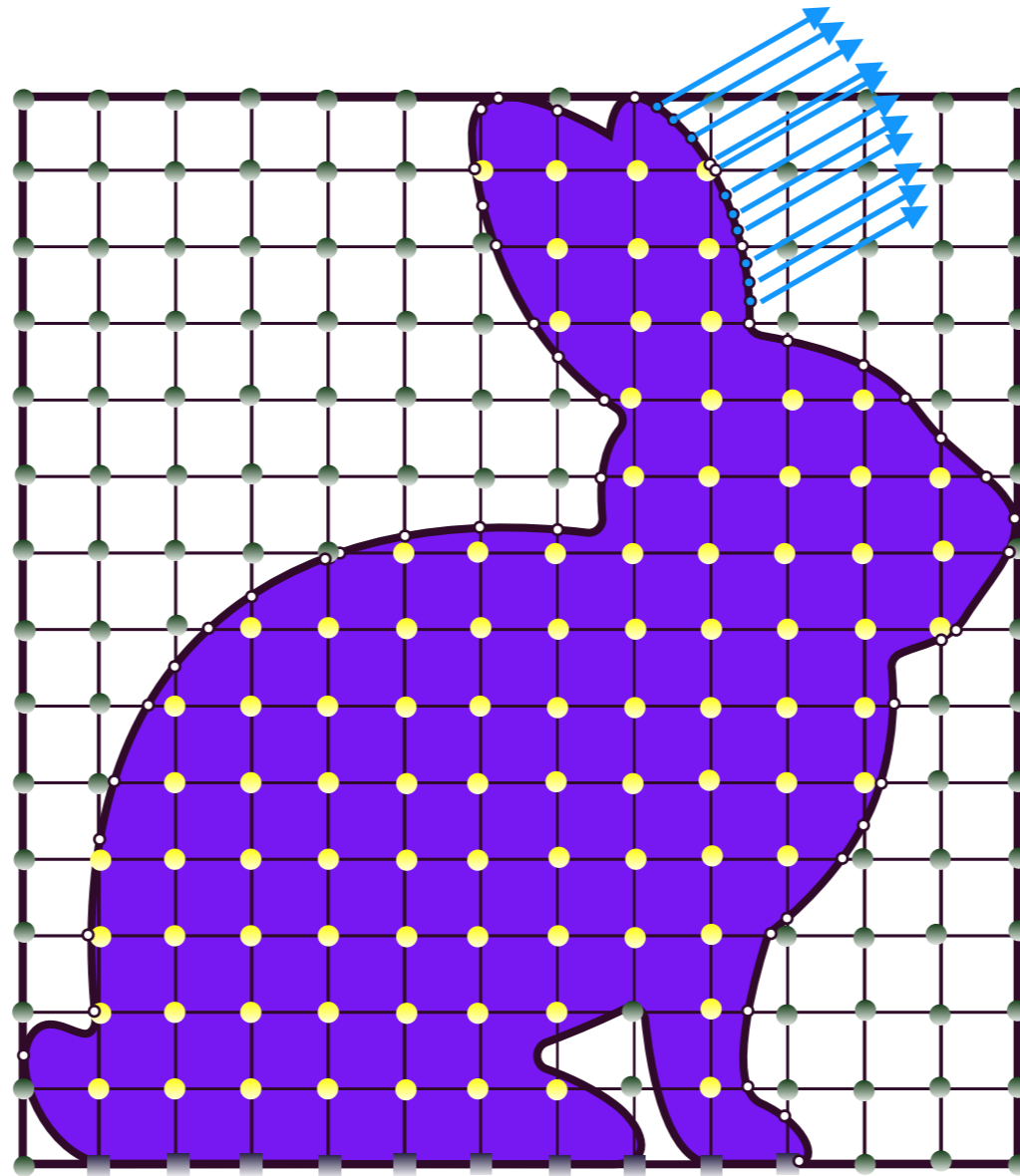
Real-time Error Control for Surgical Simulation, **HP Bui et al**, *IEEE Trans. Biomed. Eng.*, **2016**.

Controlling the Error on Target Motion through Real-time Mesh Adaptation: Applications to Deep Brain Stimulation, **HP Bui et al**, *Int J Numer Meth Bio*, **2017**.

Corotational Cut Finite Element Method for real-time surgical simulation: application to needle insertion simulation, **HP Bui et al**, *arXiv:1712.03052[cs.CE]*



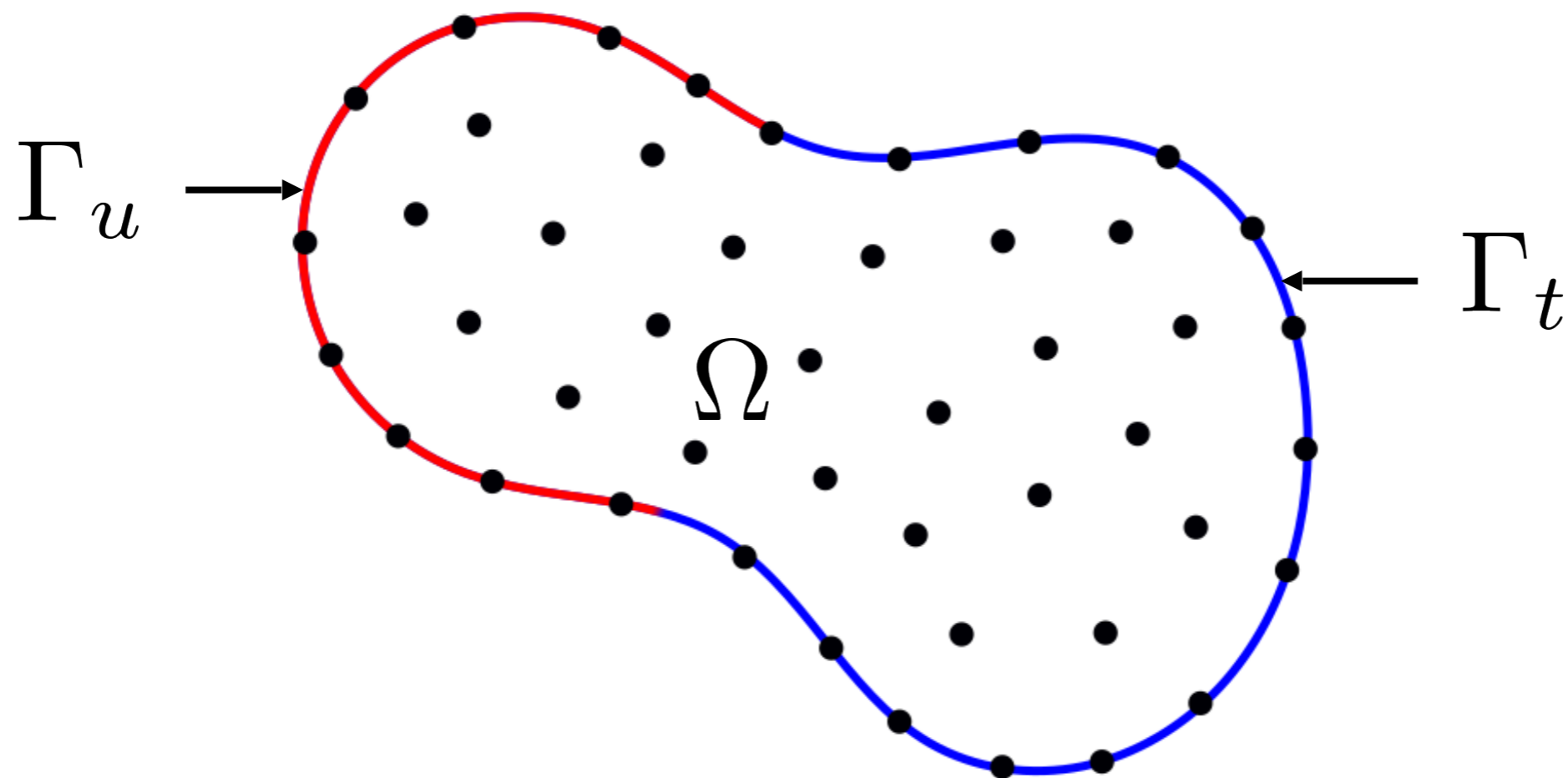
# Point Collocation Methods



Thibault Jacquemin - Satyendra Tomar  
Kostas Agathos - Shoya Mohsenimofidi

# Collocation Methods

- Collocation methods belong to the family of meshless methods
- Nodes are arbitrarily distributed over the domain  $\Omega$ , and the boundaries  $\Gamma_u$  and  $\Gamma_t$



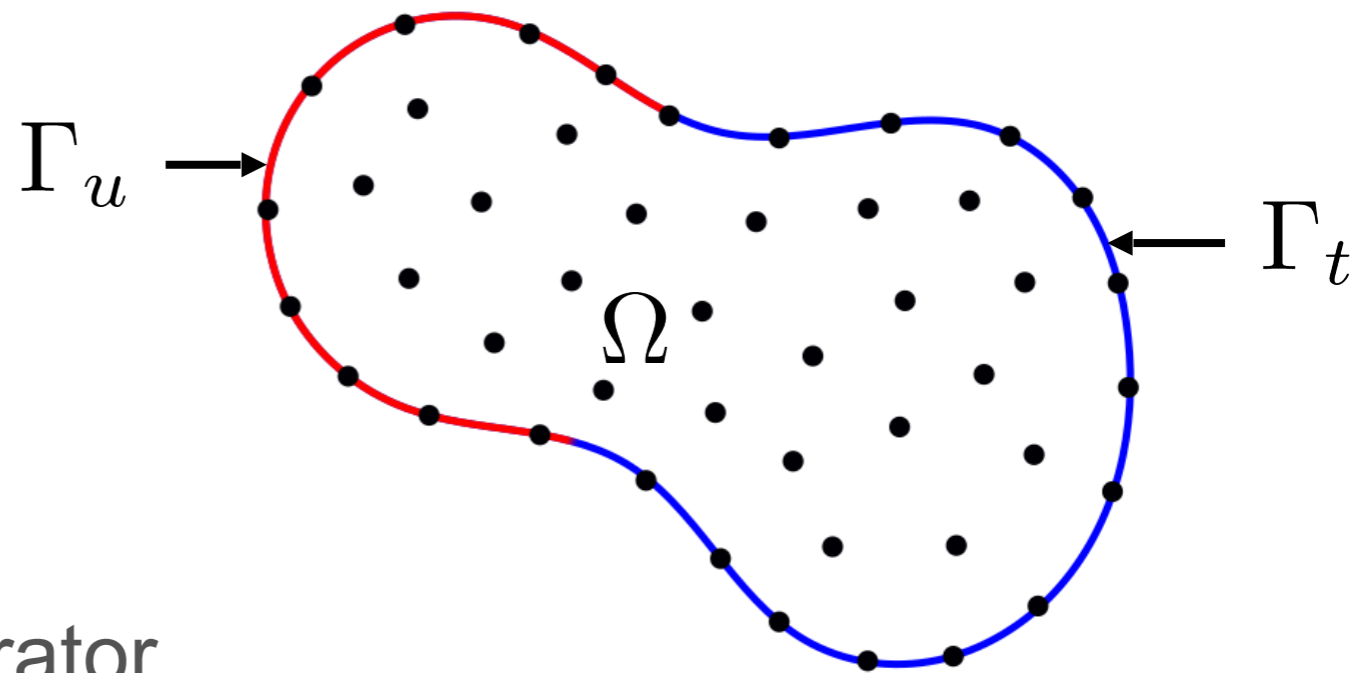
# Strong form of PDE

- Typical PDE for a field  $f$ :

$$\mathcal{A}(f) = 0 \quad \text{in} \quad \Omega$$

$$f - \bar{f} = 0 \quad \text{on} \quad \Gamma_u$$

$$\mathcal{B}(f) = 0 \quad \text{on} \quad \Gamma_t$$



- $\mathcal{A}$  is the PDE differential operator  
The known field values  $f$  are applied to  $\Gamma_u$
- $\mathcal{B}$  is a differential operator applied to  $\Gamma_t$
- The PDE is solved only at collocation points
- The collocation points are typically the nodes of the domain

# Advantages and Drawbacks of Collocation Methods

- Advantages:
  - ✓ Fewer constraints than element based methods with regards to point placement
  - ✓ Easy adaptation of the approximation to reduce the error
  - ✓ Low observed error for many problems
- Drawbacks:
  - ✗ Slower to solve than the Finite Element Method as the system matrix is non-symmetric, and denser due to larger support

# Various Types of Collocation Methods

- Based on an approximation of the unknown field
  - Moving Least Square Approximation (MLS)
  - Isogeometric Analysis (IGA)
  - Radial Basis Functions (RBF)
  - ...
- Based on an approximation of the differential operator
  - Finite Difference (FD)
  - Generalized Finite Difference (GFD)
  - Radial Basis Function Finite Difference (RBF-FD)
  - Discretization-Corrected Particle Strength Exchange (DCPSE)

# Methods of Particular Interest

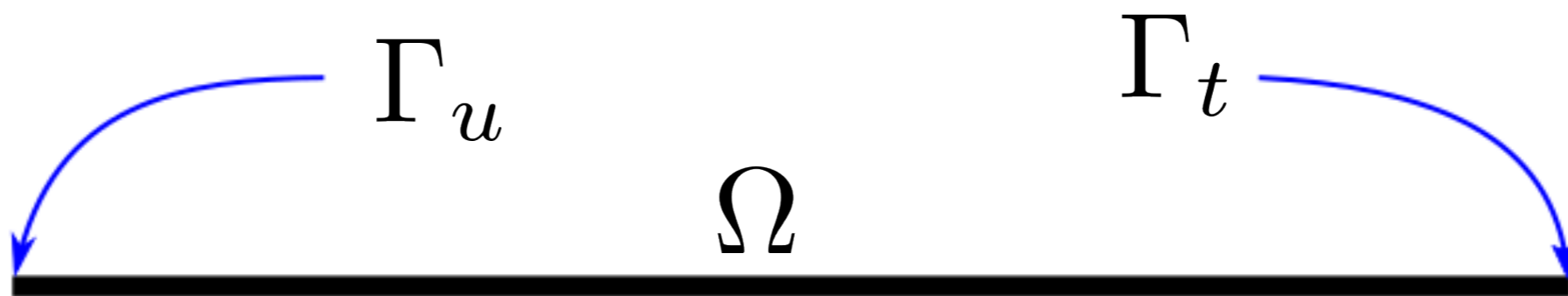
- Two methods:
  - The Generalized Finite Difference
  - The Discretization-Corrected Particle Strength Exchange
- These methods are based on a Taylor's series expansion to approximate the differential operator
- As compared to other collocation methods:
  - They lead to relatively low error
  - Fast to compute the solution

# The GFD Method: Step by Step

- Step 1: Discretization & Support Node Selection
- Step 2: Taylor's Series Approximation
- Step 3: Derivatives' Approximation
- Step 4: Assembly of the Linear System
- Step 5: Solution of the Linear System

# The GFD Method: Step by Step

- Consider the domain  $\Omega$  in 1D:



- The PDE to be solved:

$$\mathcal{A}(f) = 0 \quad \text{in} \quad \Omega$$

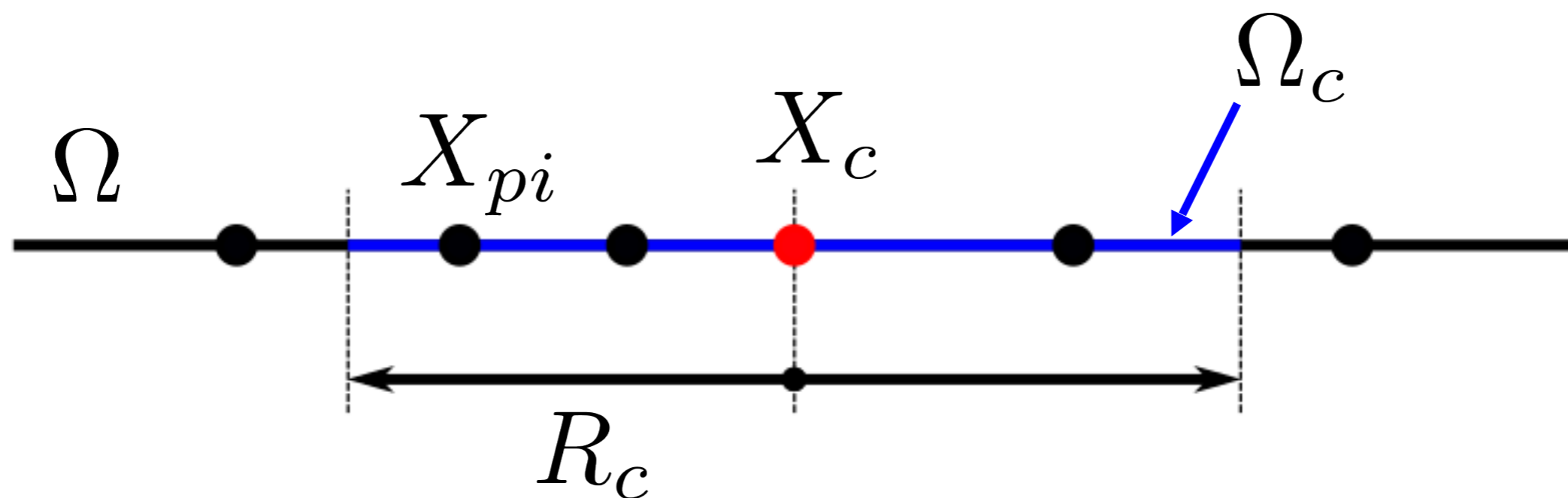
$$f - \bar{f} = 0 \quad \text{on} \quad \Gamma_u$$

$$\mathcal{B}(f) = 0 \quad \text{on} \quad \Gamma_t$$



# Discretization/Support Node Selection

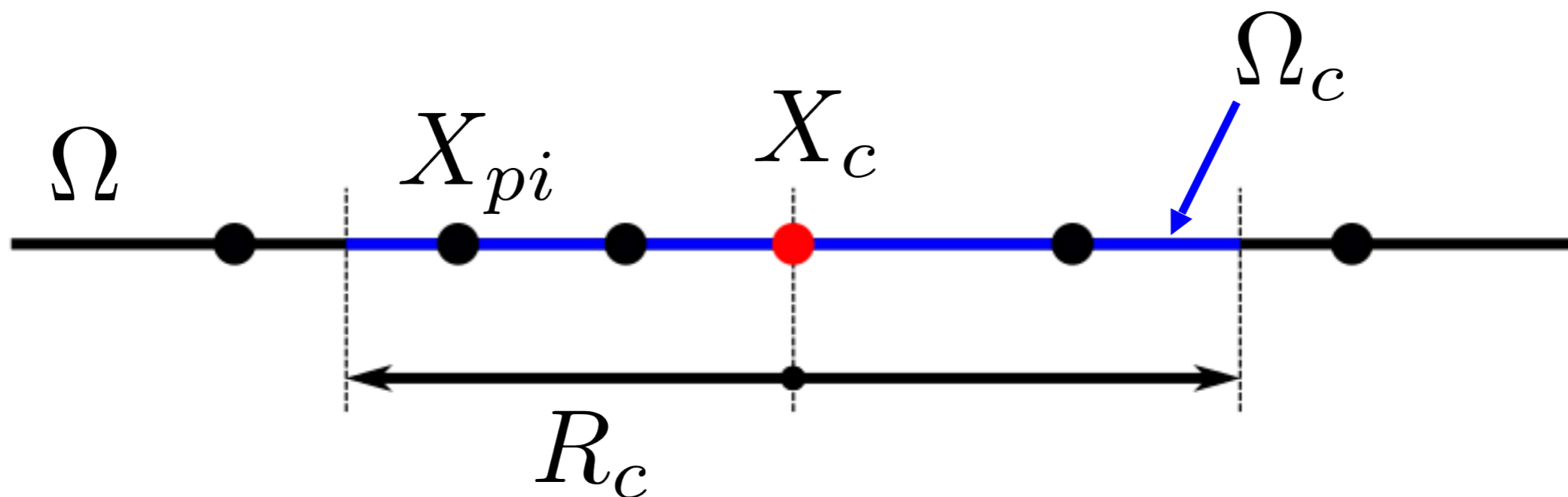
- Nodes are distributed over the domain  $\Omega$
- To each collocation node  $X_c$ , a radius  $R_c$  is associated which defines a sub-domain  $\Omega_c$  as the support of  $X_c$
- The nodes  $X_{pi}$  in  $\Omega_c$  are called the support nodes of  $X_c$
- The size of  $\Omega_c$  depends on the order of the differential operator



# Taylor's Series Approximation

- Based on a Taylor's Series Expansion, the nodes  $X_{pi}$  are used to approximate the derivatives in  $X_c$
- Since  $X_{pi}$  is located in the vicinity of  $X_c$  we can write:

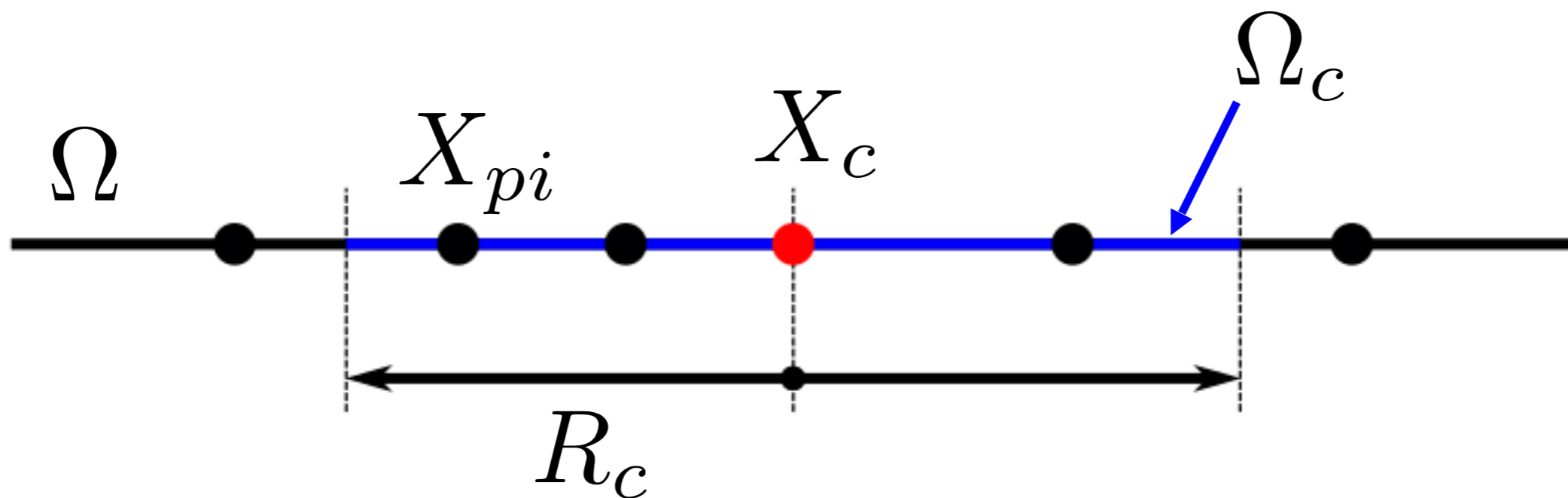
$$f(X_{pi}) = f(X_c) + \sum_{i=1}^{\infty} \frac{(X_{pi} - X_c)^i}{i!} \frac{\partial^i f(X_c)}{\partial x^i}$$



# Taylor's Series Approximation

- Considering a second order PDE, the second order approximation of the Taylor's series expansion is:

$$f_h(X_{pi}) = f(X_c) + (x_{pi} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{pi} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2}$$

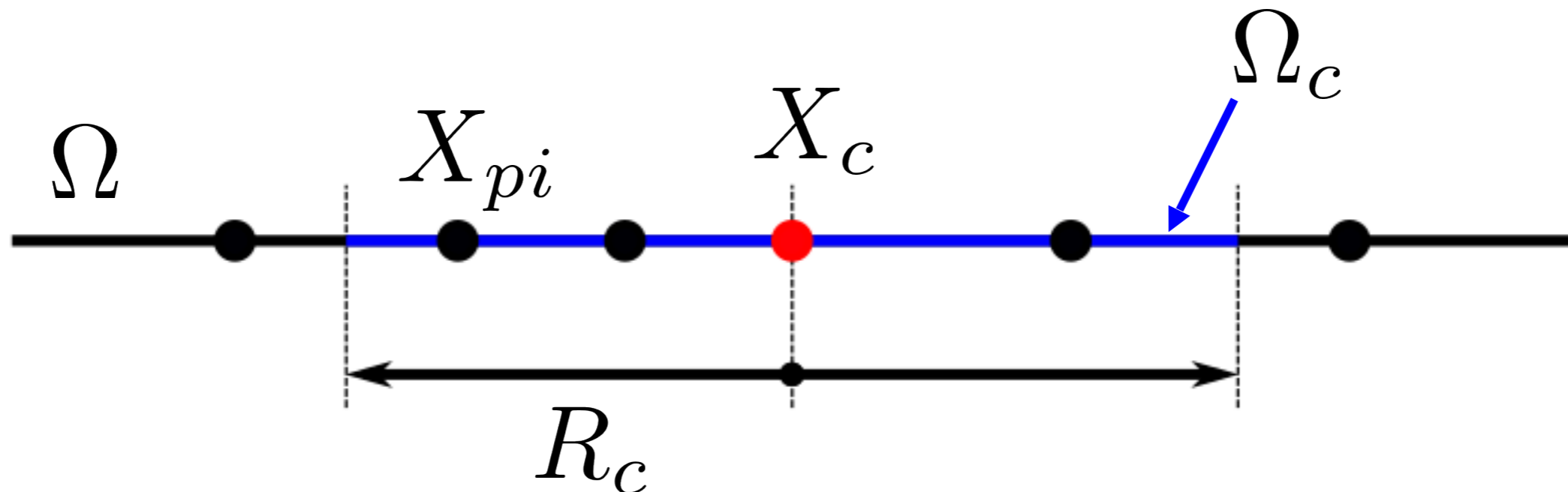


# Derivatives' Approximation

- Taylor's series expansion can be written for all nodes  $X_{pi}$  (3 for the considered example) in  $\Omega_c$

$$\begin{cases} f_h(X_{p1}) = f(X_c) + (x_{p1} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{p1} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2} \\ f_h(X_{p2}) = f(X_c) + (x_{p2} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{p2} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2} \\ f_h(X_{p3}) = f(X_c) + (x_{p3} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{p3} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2} \end{cases}$$

- 2 unknowns  $\frac{\partial f}{\partial x}(X_c)$  and  $\frac{\partial^2 f}{\partial x^2}(X_c)$ , 3 equations in this example



# Derivatives' Approximation

- What are the values of derivatives  $\frac{\partial f}{\partial x}$  and  $\frac{\partial^2 f}{\partial x^2}$  that allow to best reproduce the field values  $f(X_{pi}), i = 1,2,3$  ?
- Minimization problem described by the functional B:

$$B(X_c) = \sum_{i=1}^3 w(X_{pi} - X_c) \left[ f(X_c) - f(X_{pi}) + (x_{pi} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{pi} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2} \right]^2$$

- $w$  is a weight function centred in  $X_c$
- $w$  balances the contribution of each support node as a function of its distance to  $X_c$

# Derivatives' Approximation

- What are the derivatives  $\frac{\partial f}{\partial x}$  and  $\frac{\partial^2 f}{\partial x^2}$  that allow to best reproduce the field values  $f(X_{pi}), i = 1,2,3$  ?

$$\left. \frac{\partial B(X)}{\partial Df(X)} \right|_{X=X_c} = 0$$

where

$$Df(X) = \left[ \frac{\partial f(X)}{\partial x}, \frac{\partial^2 f(X)}{\partial x^2} \right]^T$$

# Derivatives' Approximation

- Derivatives of  $B(X)$  w.r.t.  $\frac{\partial f}{\partial x}$  and  $\frac{\partial^2 f}{\partial x^2}$  at  $X_c$  are

$$\sum_{i=1}^3 w(X_{pi} - X_c)(x_{pi} - x_c) \left[ f(X_c) - f(X_{pi}) + (x_{pi} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{pi} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2} \right]$$

$$\sum_{i=1}^3 w(X_{pi} - X_c) \frac{(x_{pi} - x_c)^2}{2!} \left[ f(X_c) - f(X_{pi}) + (x_{pi} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{pi} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2} \right]$$

- Equating these terms to zero, this can be written in a matrix form as:

$$\begin{bmatrix} m_{11} & m_{12} \\ m_{21} & m_{22} \end{bmatrix} \begin{bmatrix} \frac{\partial f(X_c)}{\partial x} \\ \frac{\partial^2 f(X_c)}{\partial x^2} \end{bmatrix} = \begin{bmatrix} -m_{01} & m_{01,1} & m_{01,2} & m_{01,3} \\ -m_{02} & m_{02,1} & m_{02,2} & m_{02,3} \end{bmatrix} \begin{bmatrix} f(X_c) \\ f(X_{p1}) \\ f(X_{p2}) \\ f(X_{p3}) \end{bmatrix}$$

# Derivatives' Approximation

$$\begin{bmatrix} m_{11} & m_{12} \\ m_{21} & m_{22} \end{bmatrix} \begin{bmatrix} \frac{\partial f(X_c)}{\partial x} \\ \frac{\partial^2 f(X_c)}{\partial x^2} \end{bmatrix} = \begin{bmatrix} -m_{01} & m_{01,1} & m_{01,2} & m_{01,3} \\ -m_{02} & m_{02,1} & m_{02,2} & m_{02,3} \end{bmatrix} \begin{bmatrix} f(X_c) \\ f(X_{p1}) \\ f(X_{p2}) \\ f(X_{p3}) \end{bmatrix}$$

- The coefficients of this system are:

$$m_{ij,k} = w(X_{pk} - X_c) P_{(i+1)k}(X_c) P_{(j+1)k}(X_c)$$

$$m_{ij} = \sum_{k=1}^3 m_{ij,k}$$

$$P(X_c) = \begin{bmatrix} 1 & 1 & 1 \\ (x_{p1} - x_c) & (x_{p2} - x_c) & (x_{p3} - x_c) \\ \frac{(x_{p1} - x_c)^2}{2!} & \frac{(x_{p2} - x_c)^2}{2!} & \frac{(x_{p3} - x_c)^2}{2!} \end{bmatrix}$$



# Derivatives' Approximation

- The derivatives  $\frac{\partial f}{\partial x}$  and  $\frac{\partial^2 f}{\partial x^2}$  can then be approximated as a function of the field values  $F(X_c)$

$$\underbrace{\begin{bmatrix} m_{11} & m_{12} \\ m_{21} & m_{22} \end{bmatrix}}_{A(X_c)} \underbrace{\begin{bmatrix} \frac{\partial f(X_c)}{\partial x} \\ \frac{\partial^2 f(X_c)}{\partial x^2} \end{bmatrix}}_{Df(X_c)} = \underbrace{\begin{bmatrix} -m_{01} & m_{01,1} & m_{01,2} & m_{01,3} \\ -m_{02} & m_{02,1} & m_{02,2} & m_{02,3} \end{bmatrix}}_{E(X_c)} \underbrace{\begin{bmatrix} f(X_c) \\ f(X_{p1}) \\ f(X_{p2}) \\ f(X_{p3}) \end{bmatrix}}_{F(X_c)}$$

- $Df(X_c) = A^{-1}(X_c)E(X_c)F(X_c)$

# Derivatives' Approximation

- Different types of weight functions:
  - 3<sup>rd</sup> Order Splines

$$w(s) = \begin{cases} \frac{2}{3} - 4s^2 + 4s^3 & \text{if } s \leq 0.5 \\ \frac{4}{3} - 4s + 4s^2 - \frac{4}{3}s^3 & \text{if } 0.5 < s \leq 1 \\ 0 & \text{if } s > 1 \end{cases}$$

- 4<sup>th</sup> Order Splines

$$w(s) = \begin{cases} 1 - 6s^2 + 8s^3 - 3s^4 & \text{if } s \leq 1 \\ 0 & \text{if } s > 1 \end{cases}$$

# Assembly of Linear System

$$\begin{matrix} 1 \\ \vdots \\ i \\ \vdots \\ n \end{matrix} \left[ \begin{array}{c} \phantom{A} \\ \phantom{A} \\ \phantom{A} \\ \phantom{A} \\ \phantom{A} \end{array} \right] \begin{matrix} \\ \\ \\ \\ \\ \end{matrix} \begin{matrix} \\ \\ \\ \\ \\ \end{matrix} \left[ \begin{array}{c} \phantom{F} \\ \phantom{F} \\ \phantom{F} \\ \phantom{F} \\ \phantom{F} \end{array} \right] = \begin{matrix} \\ \\ \\ \\ \\ \end{matrix} \left[ \begin{array}{c} \phantom{B} \\ \phantom{B} \\ \phantom{B} \\ \phantom{B} \\ \phantom{B} \end{array} \right]$$

$A \quad F \quad = \quad B$

Equation for Collocation Node "i"

# The DCPSE Method: Step by Step

- Step 1: Discretization & Support Node Selection
  - Step 2: Taylor's Series Approximation
  - Step 3: Derivatives Approximation
  - Step 4: Assembly of the Linear System
  - Step 5: Resolution of the Linear System
- The operations performed during the steps 1, 2, 4 and 5 are the same as for the GFD method

# DCPSE: Derivatives' Approximation

- The Taylor's series expansion is convoluted by a function  $\eta$  over the domain  $\Omega_c$

$$\begin{aligned} \int_{\Omega_c} f_h(X_p) \eta(X_p - X_c) dX_p &= \int_{\Omega_c} f(X_c) \eta(X_p - X_c) dX_p \\ &+ \int_{\Omega_c} \frac{\partial f(X_c)}{\partial x} (x_p - x_c) \eta(X_p - X_c) dX_p \\ &+ \int_{\Omega_c} \frac{\partial^2 f(X_c)}{\partial x^2} \frac{(x_p - x_c)^2}{2!} \eta(X_p - X_c) dX_p \end{aligned}$$

- Compare with the GFD terms

$$\begin{cases} f_h(X_{p1}) = f(X_c) + (x_{p1} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{p1} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2} \\ f_h(X_{p2}) = f(X_c) + (x_{p2} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{p2} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2} \\ f_h(X_{p3}) = f(X_c) + (x_{p3} - x_c) \frac{\partial f(X_c)}{\partial x} + \frac{(x_{p3} - x_c)^2}{2!} \frac{\partial^2 f(X_c)}{\partial x^2} \end{cases}$$

# DCPSE: Derivatives' Approximation

- The coefficient vector  $\mathbf{a}$  of the correction function is selected in order to satisfy the moment condition
- For instance, for a second order derivative approximation :

$$\left\{ \begin{array}{l} M_0(X_c) = 0 \quad \Leftrightarrow \quad \sum_{i=1}^3 P(X_{pi} - X_c)^T \mathbf{a} w(X_{pi} - X_c) = 0 \\ M_1(X_c) = 0 \quad \Leftrightarrow \quad \sum_{i=1}^3 (x_{pi} - x_c) P(X_{pi} - X_c)^T \mathbf{a} w(X_{pi} - X_c) = 0 \\ M_2(X_c) = 1 \quad \Leftrightarrow \quad \sum_{i=1}^3 \frac{(x_{pi} - x_c)^2}{2!} P(X_{pi} - X_c)^T \mathbf{a} w(X_{pi} - X_c) = 1 \end{array} \right.$$

# DCPSE: Derivatives' Approximation

- This problem can be put in a matrix form:

$$\begin{bmatrix} A_{11} & A_{12} & A_{13} \\ A_{21} & A_{22} & A_{23} \\ A_{31} & A_{32} & A_{33} \end{bmatrix} \begin{bmatrix} a_1 \\ a_2 \\ a_3 \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$

- The coefficients are

$$A_{i,j}(X_c) = \sum_{i=1}^3 Q_i(X_c, X_{pi}) P_j(X_{pi} - X_c) w(X_{pi} - X_c)$$

$$Q(X_c, X_p) = \left[ 1, (x_p - x_c), \frac{(x_p - x_c)^2}{2!} \right]^T$$

# DCPSE: Derivatives' Approximation

- The vector  $a$  can be calculated for the derivative

$$D^{n_x}(f(X_c))$$

- The derivative can then be approximated as:

$$D^{n_x} f(X_c) = \sum_{i=1}^3 f_h(X_{pi}) P(X_p - X_c)^T a w(X_p - X_c)$$



# DCPSE vs. GFD

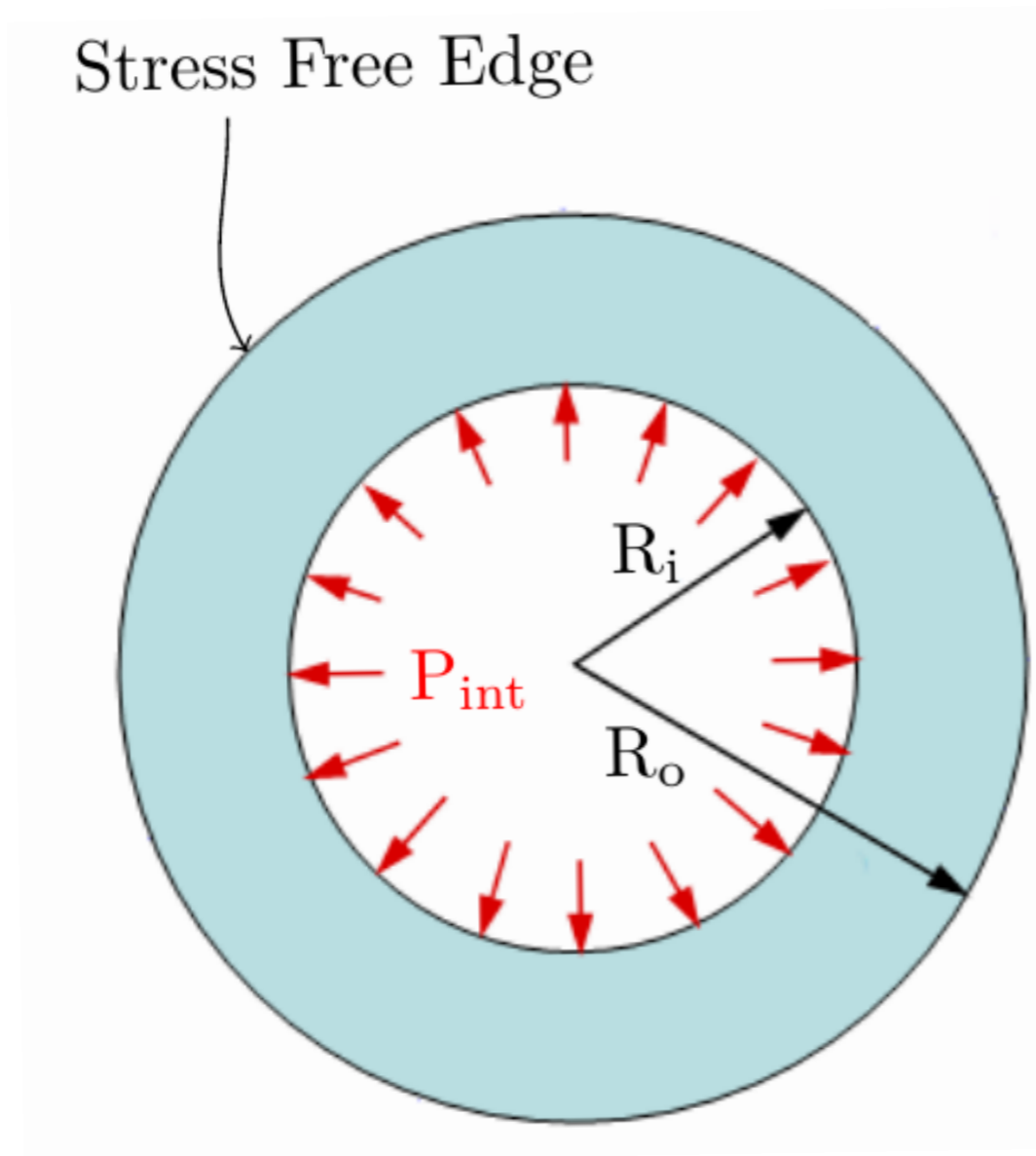
- Means of approximating the derivatives
- Derivatives approximated by GFD reproduce the field values at the support nodes using Taylor's series expansion
- DCPSE also uses Taylor's series expansion to approximate the derivatives but it uses a convolution function to cancel selected terms

# Sensitivity to various parameters

- Based on a sensitivity study, the following parameters are chosen for the case of 2D and 3D linear elastic problems

Parameter	GFD	DC PSE
Weight Function Type	4 <sup>th</sup> Order Spline	Exponential
Correction Function	N/A	Polynomial
• Size of Inner Nodes Support (2D/3D)	11/37	13/37
Size of Boundary Nodes Support (2D/3D)	19/75	17/75

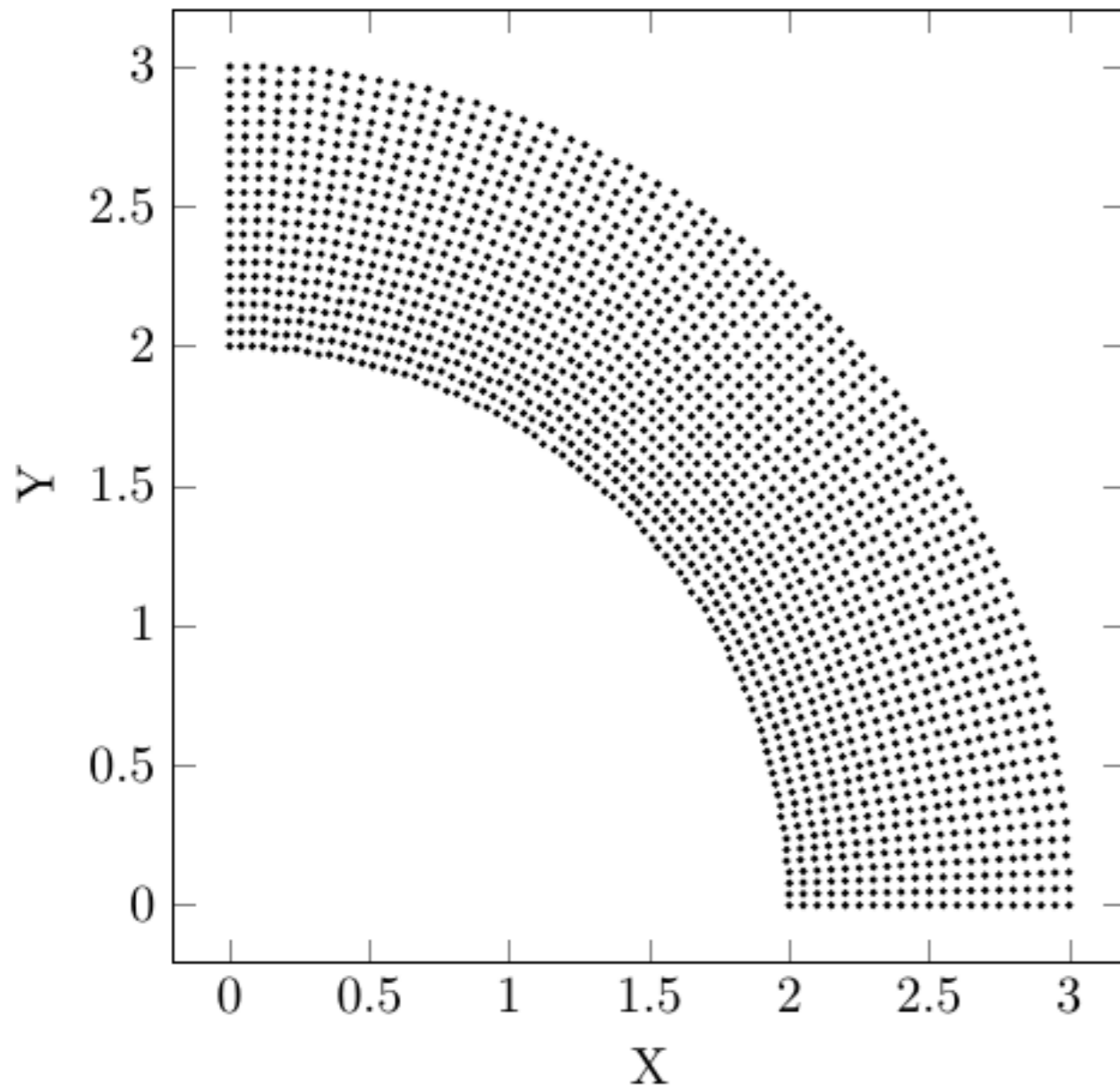
# Model problem



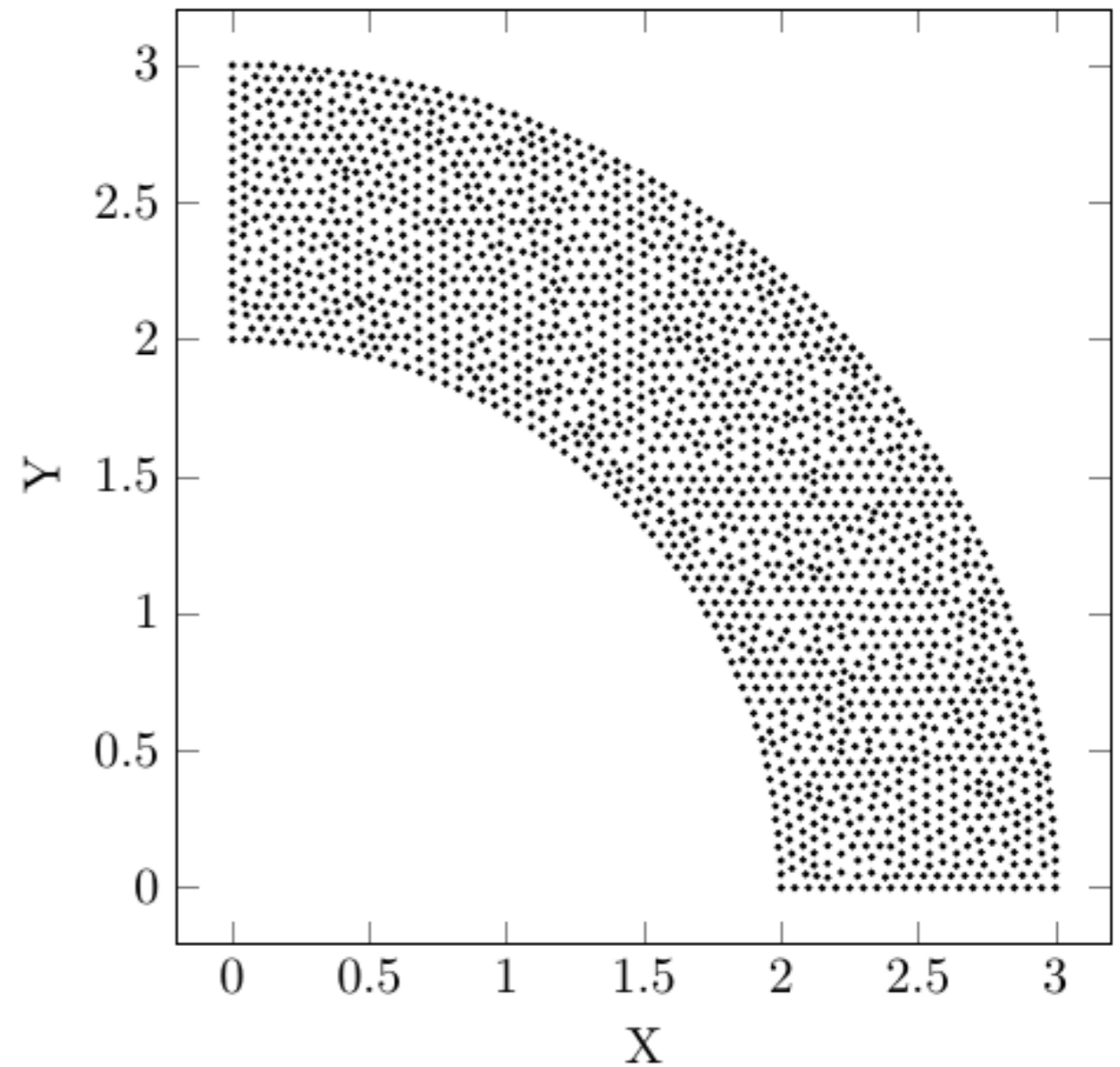
Plane Stress Cylinder  
under Internal Pressure

- Only a quarter of the cylinder modelled due to the symmetries in the cartesian coordinate system

# Nodes arrangements, structured vs. Delaunay triangulation



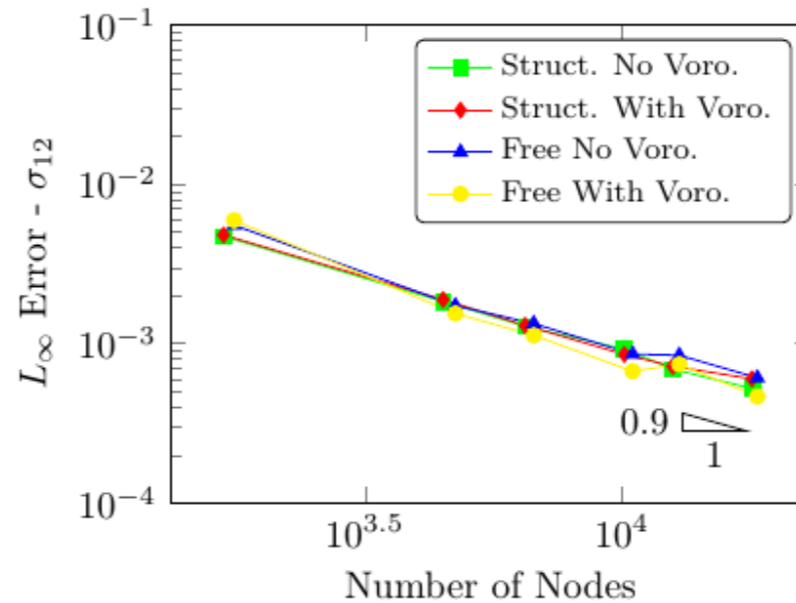
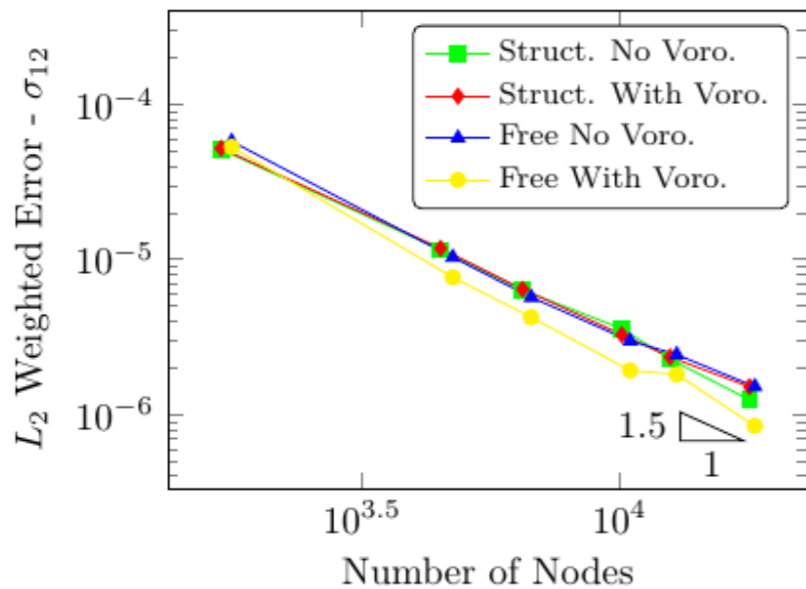
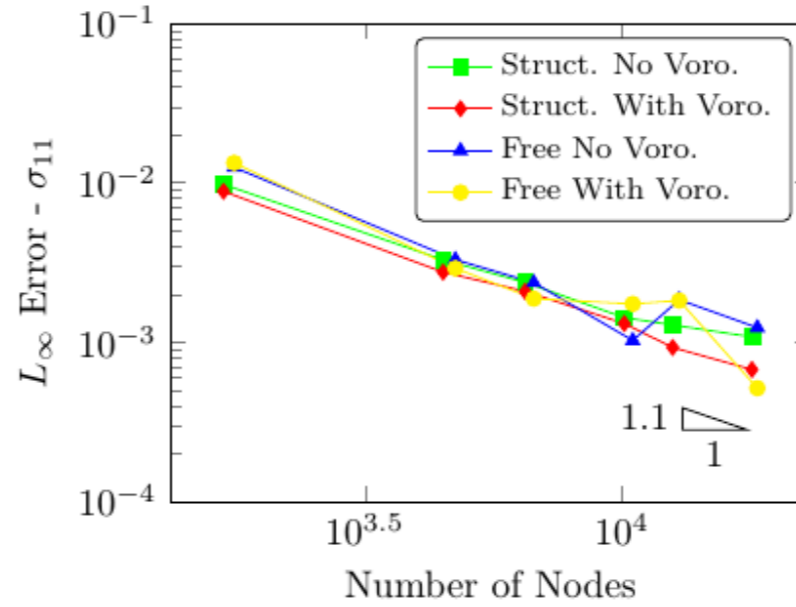
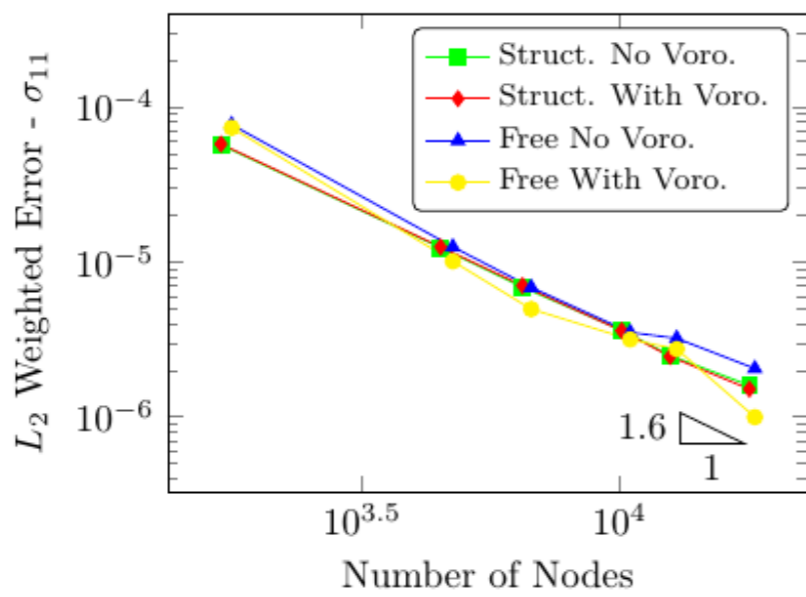
Structured Node Arrangement



Delaunay Triangulation Node Arrangement (Free)

No significant impact on accuracy/convergence

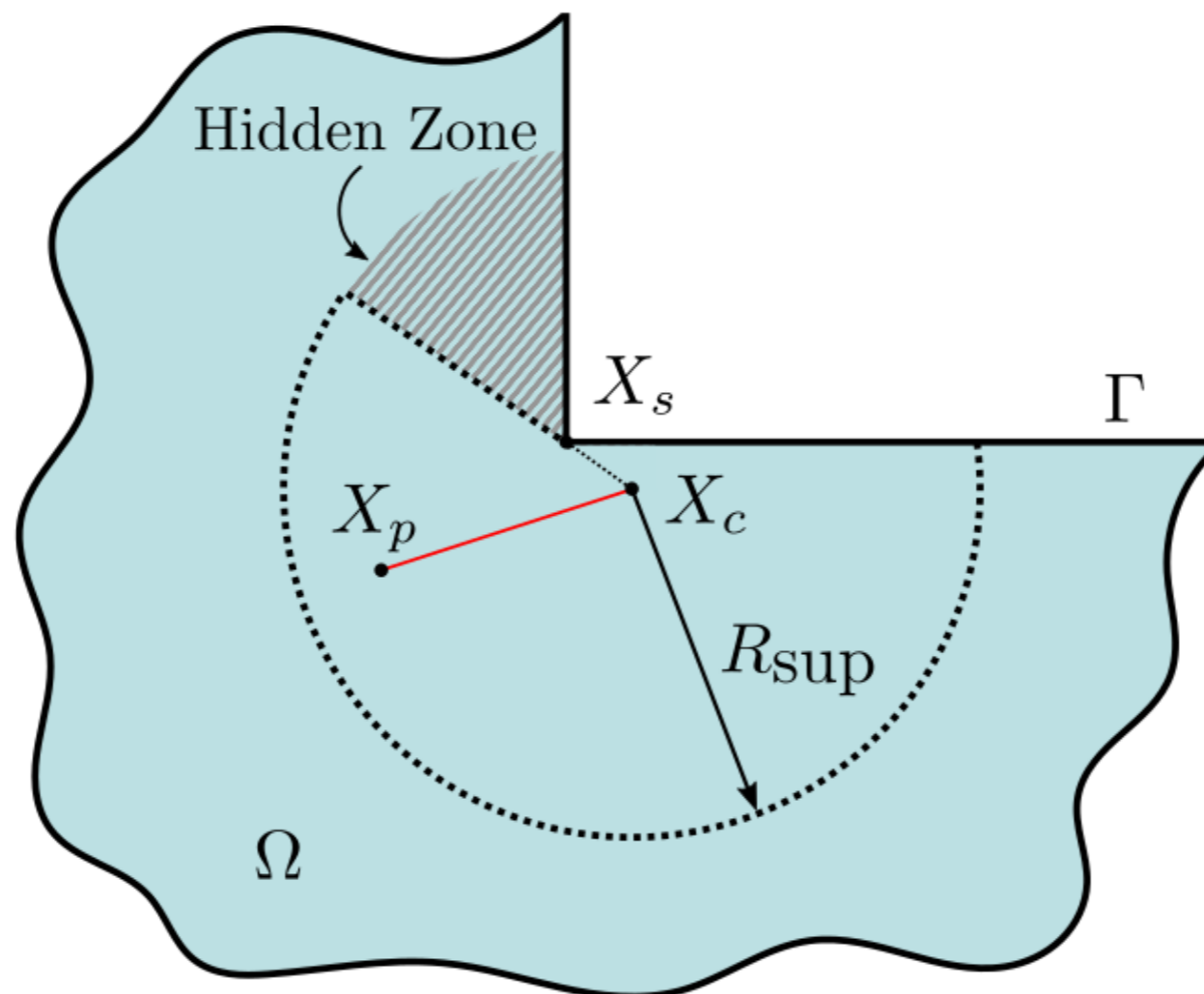
# Voronoi Based Weights: GFD



- No improvement or worse results for the structured node arrangement
- Small improvement for the node arrangement based on a Delaunay Triangulation

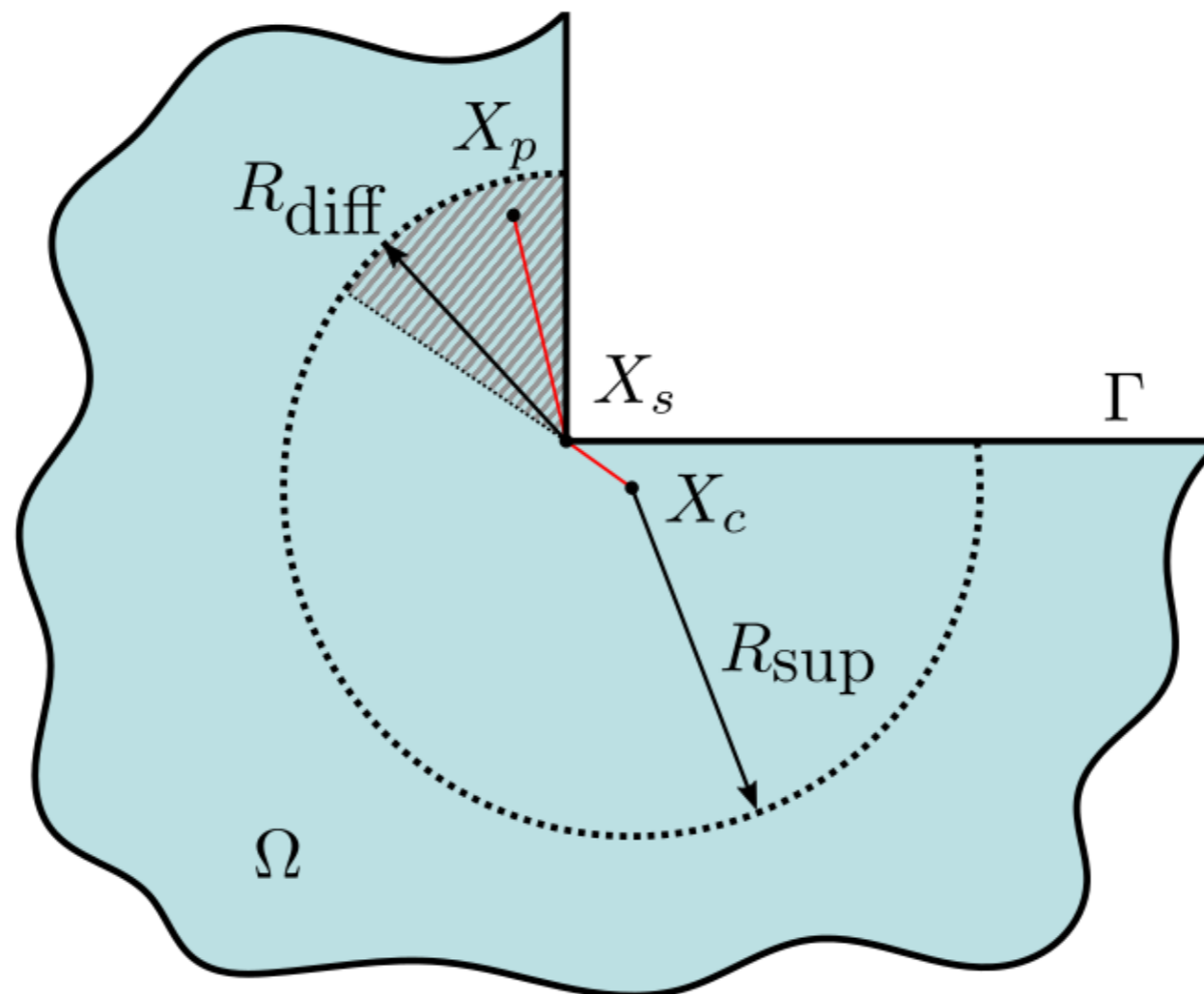
# Node Selection Around Singularities

- The visibility criterion:  
Only the nodes visible from the collocation nodes are selected in the support of the collocation node



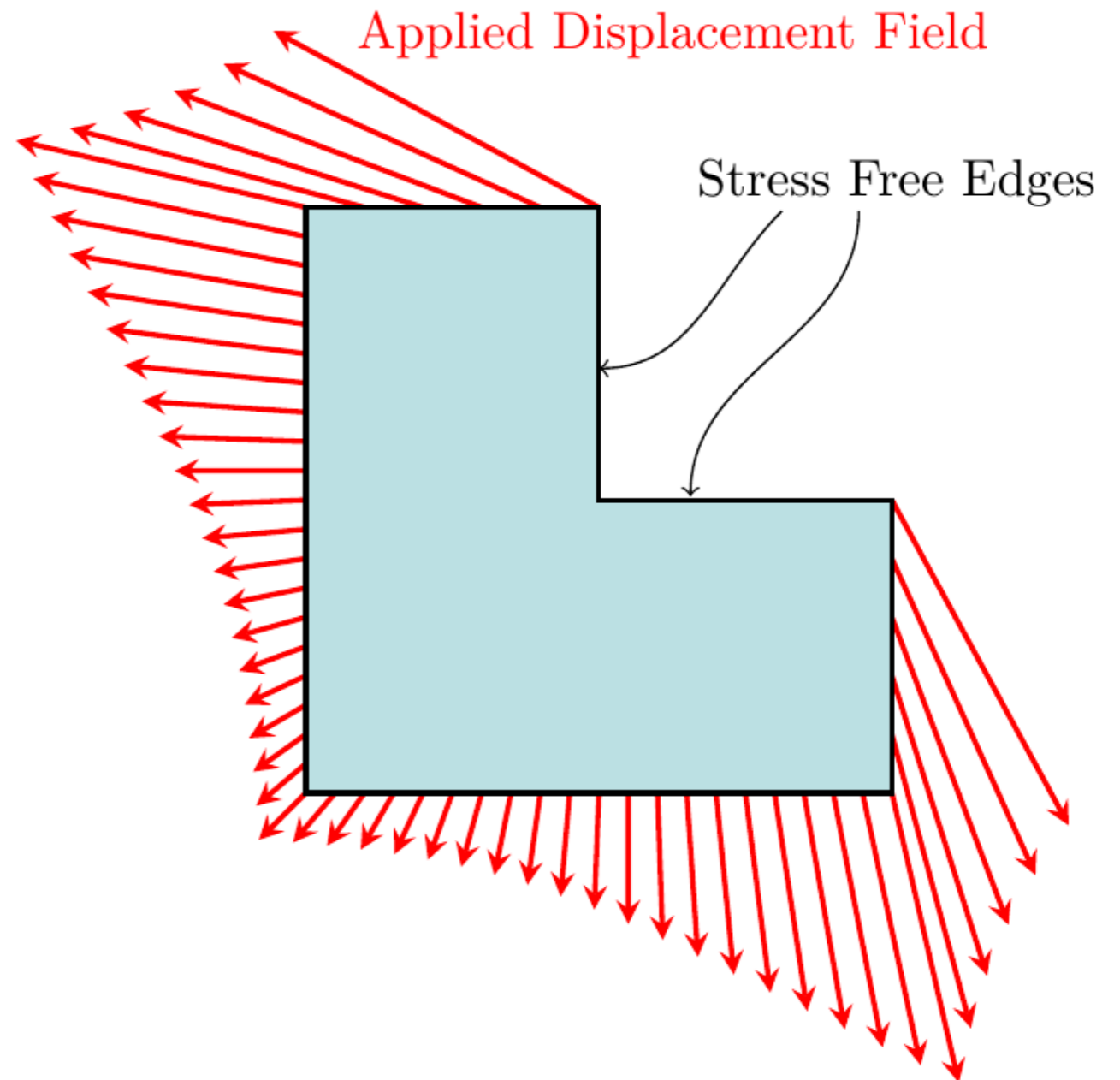
# Node Selection Around Singularities

- The diffraction criterion:  
The “hidden” nodes are included in the support of the collocation node only if the “diffracted” length is smaller than the support radius



# Node Selection Around Singularities: Model Problem

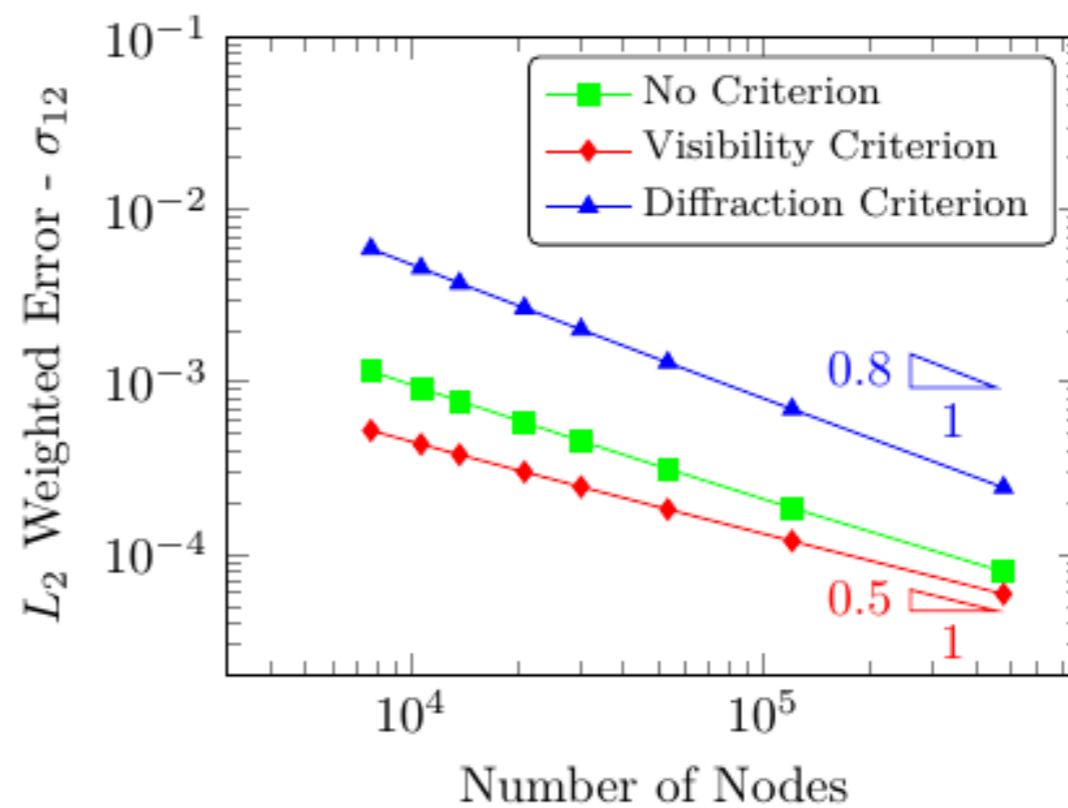
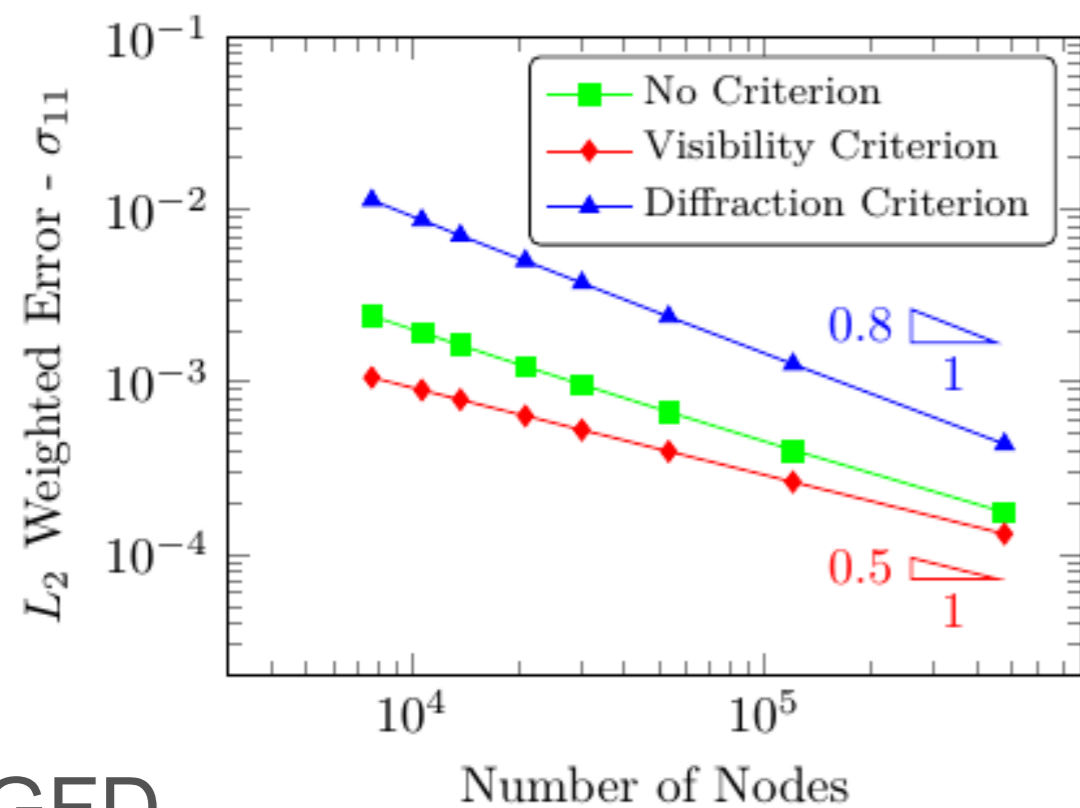
## L-Shape in Mode I Loading



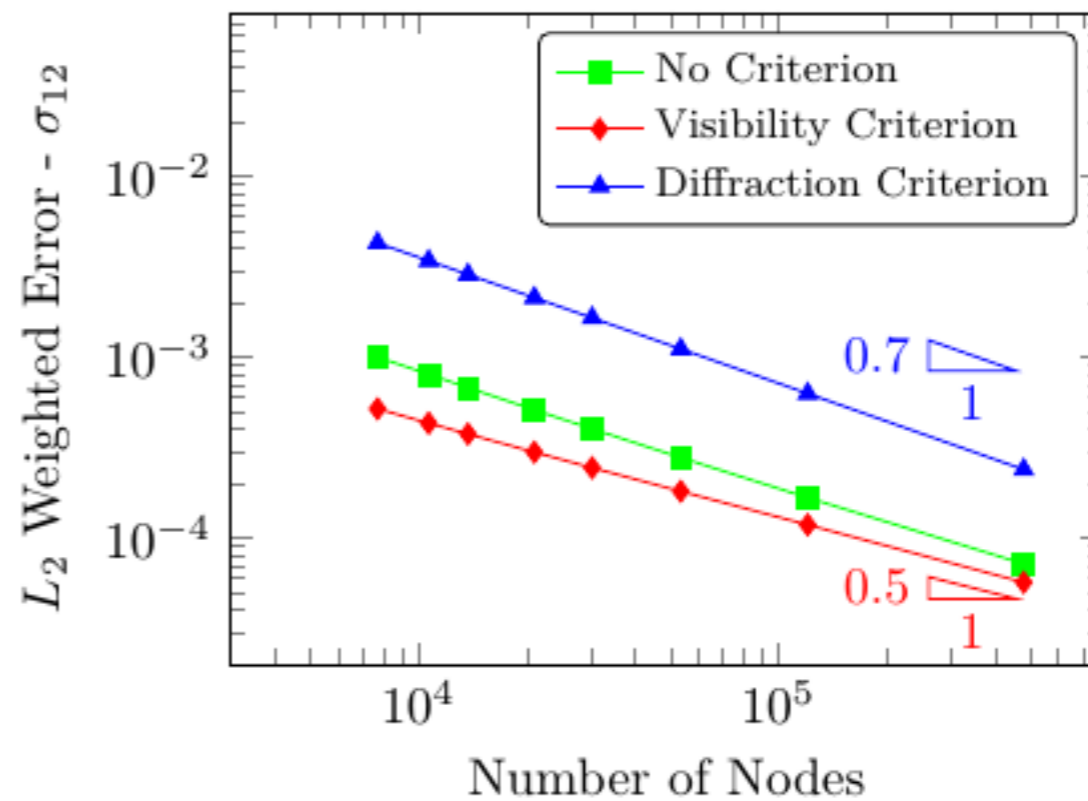
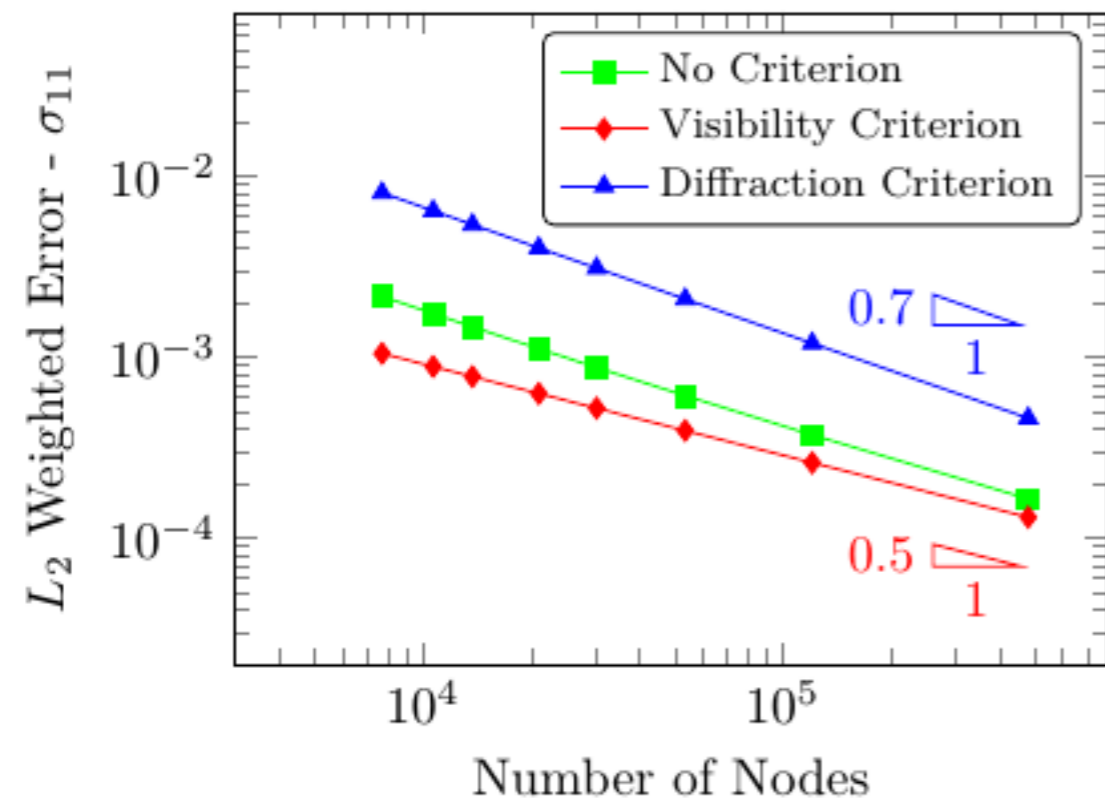


# Visibility superior to diffraction

DCPSE

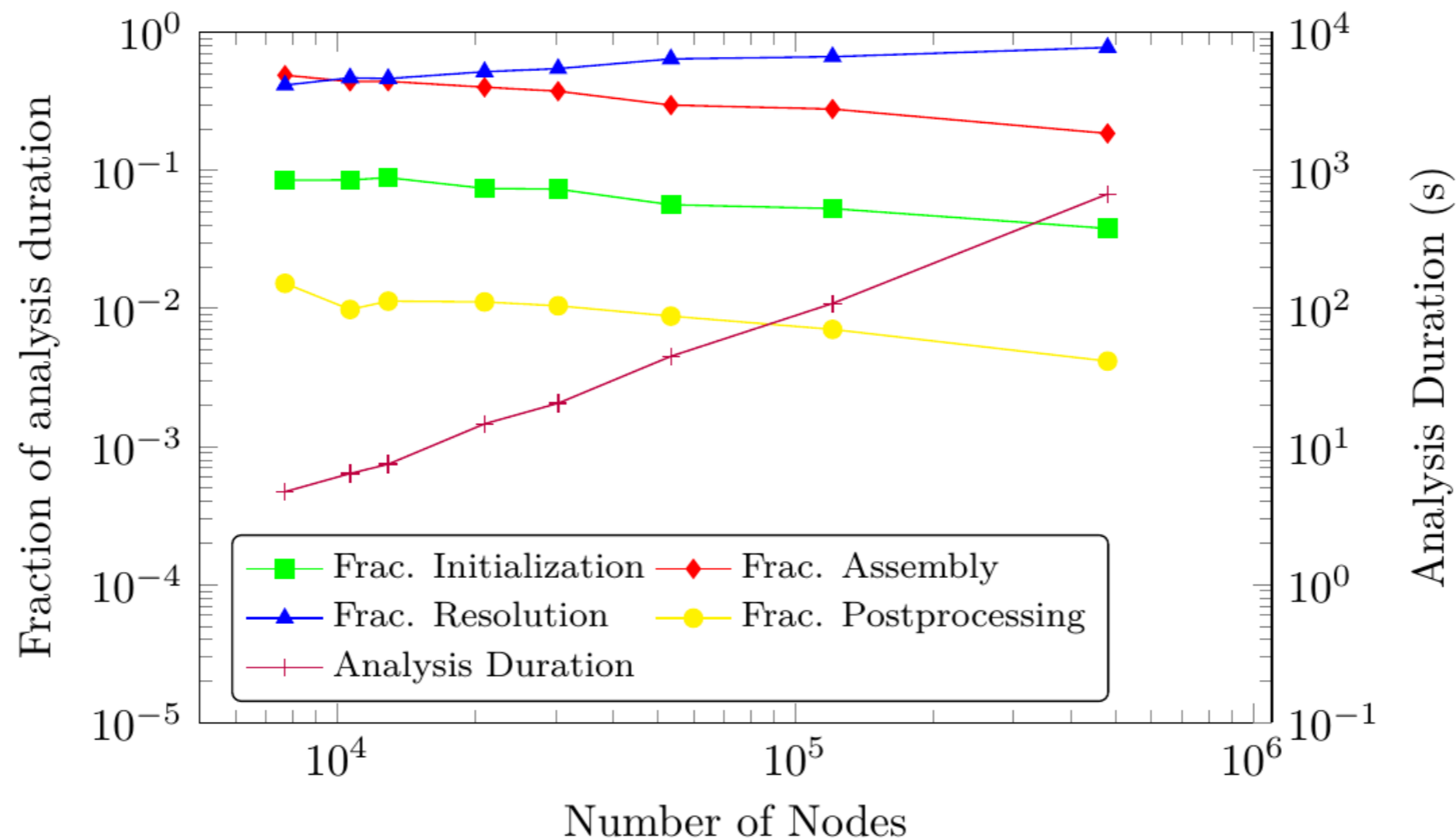


GFD

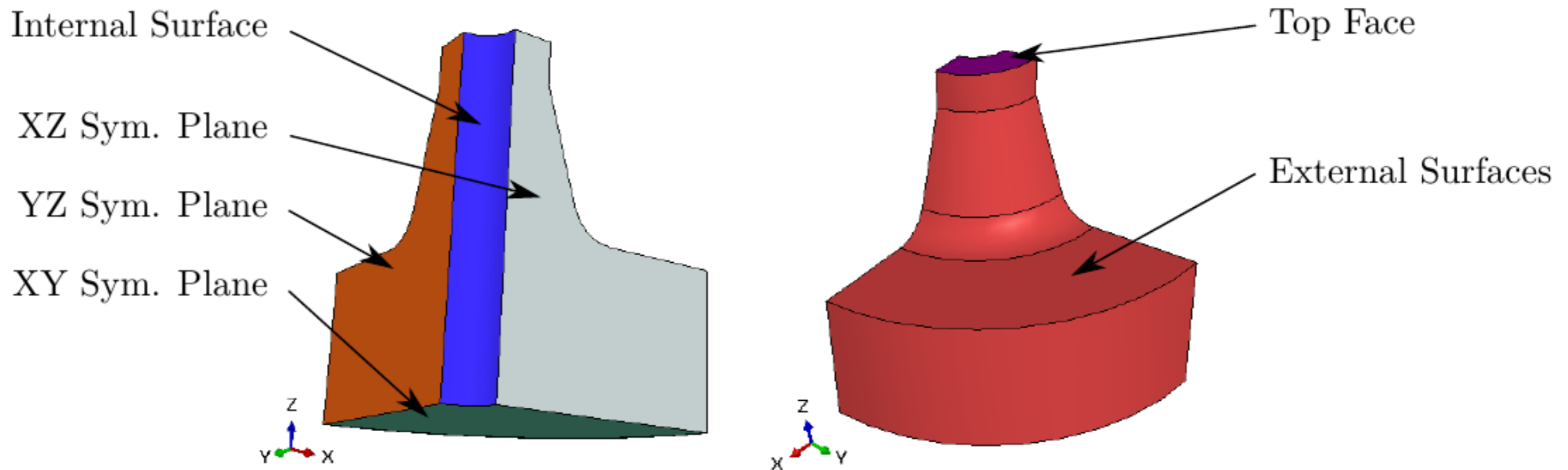


# Timings Analysis – L-Shaped Problem

- Similar analysis durations and time split are observed for the GFD and the DCPSE methods (right axes for total time)
- For problems of large dimension, the largest fraction of the analysis is spent in the solution step

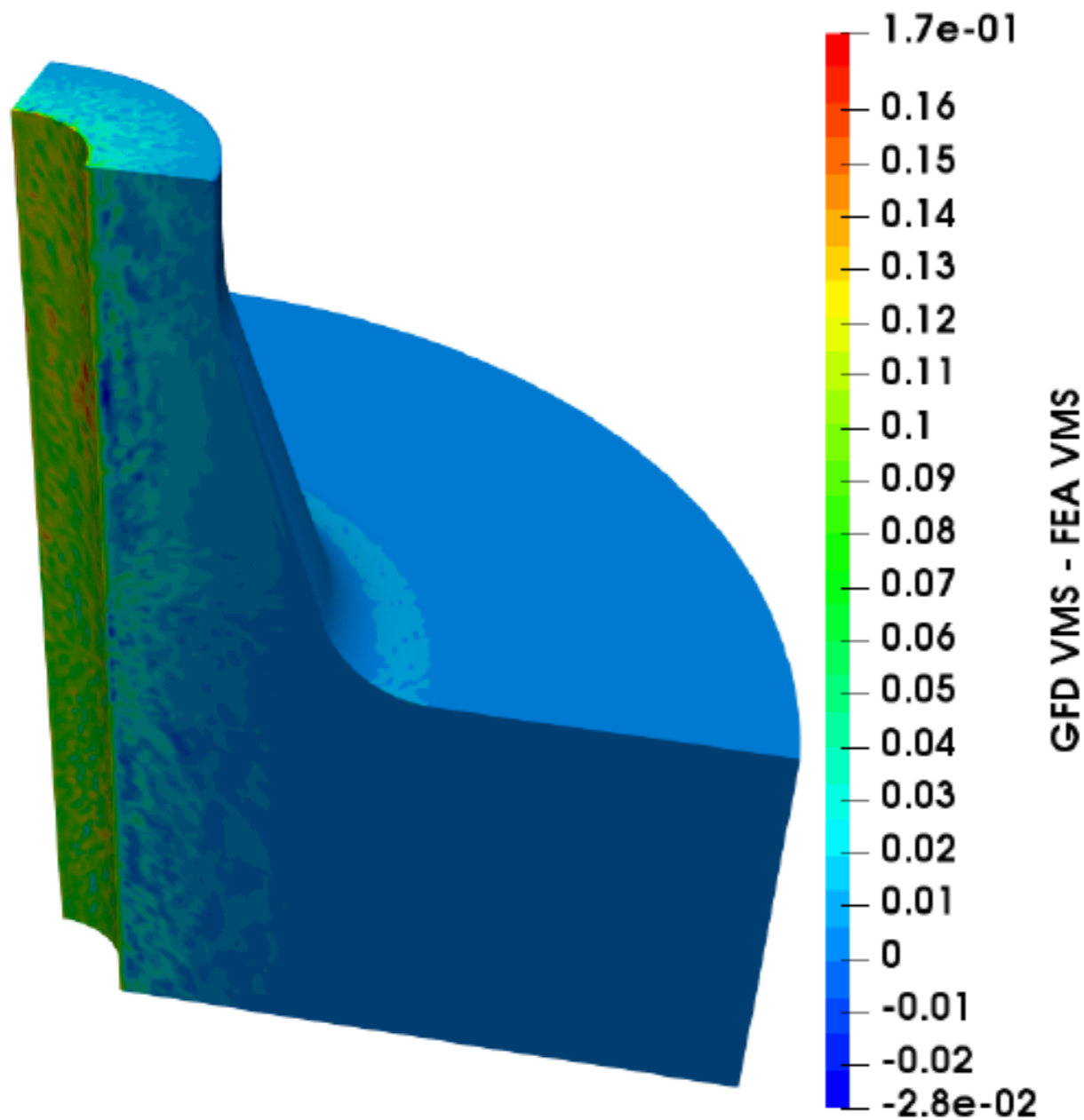


# 3D Results – ISO Flange: Model and Boundary Conditions

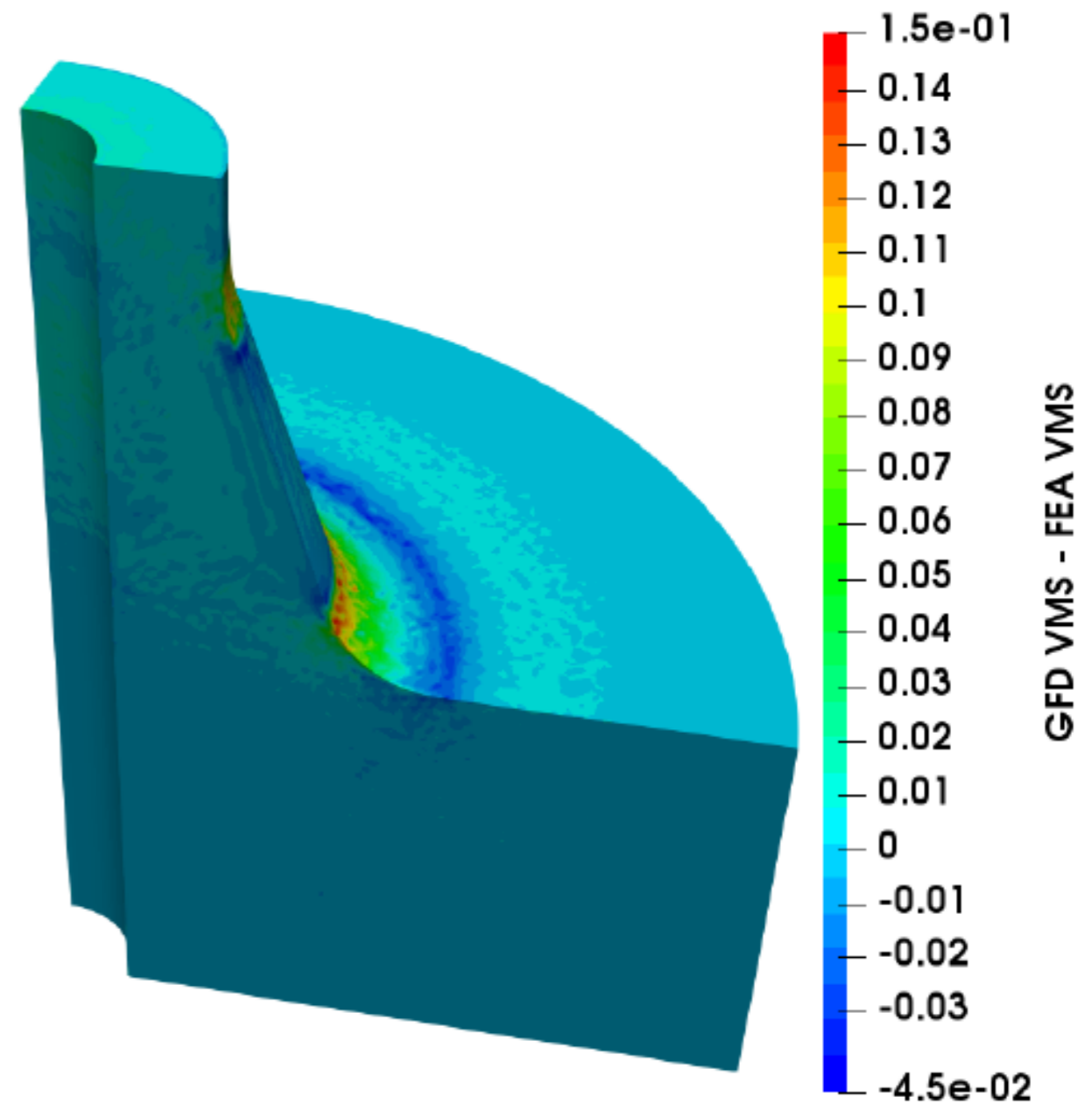


Surface(s)	Pressure Loading	Displacement Loading
XZ, YZ and XY Sym. Planes	Resp. constrained in the Y, X and Z directions	
External Surface	Stress free	
Top Face	Constrained in Z	Applied displacement in Z
Internal Surface	Constant pressure	Stress free

# 3D Results – ISO Flange: Difference (GFD - FEA)

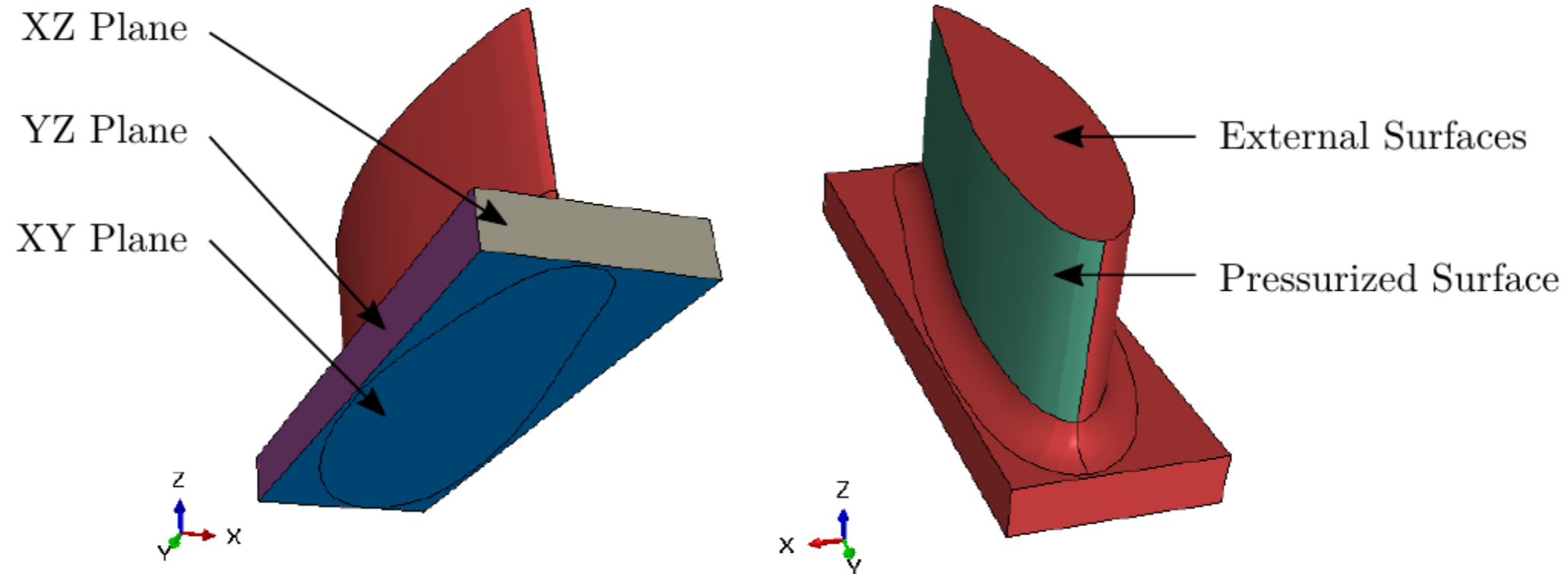


Pressure Loading



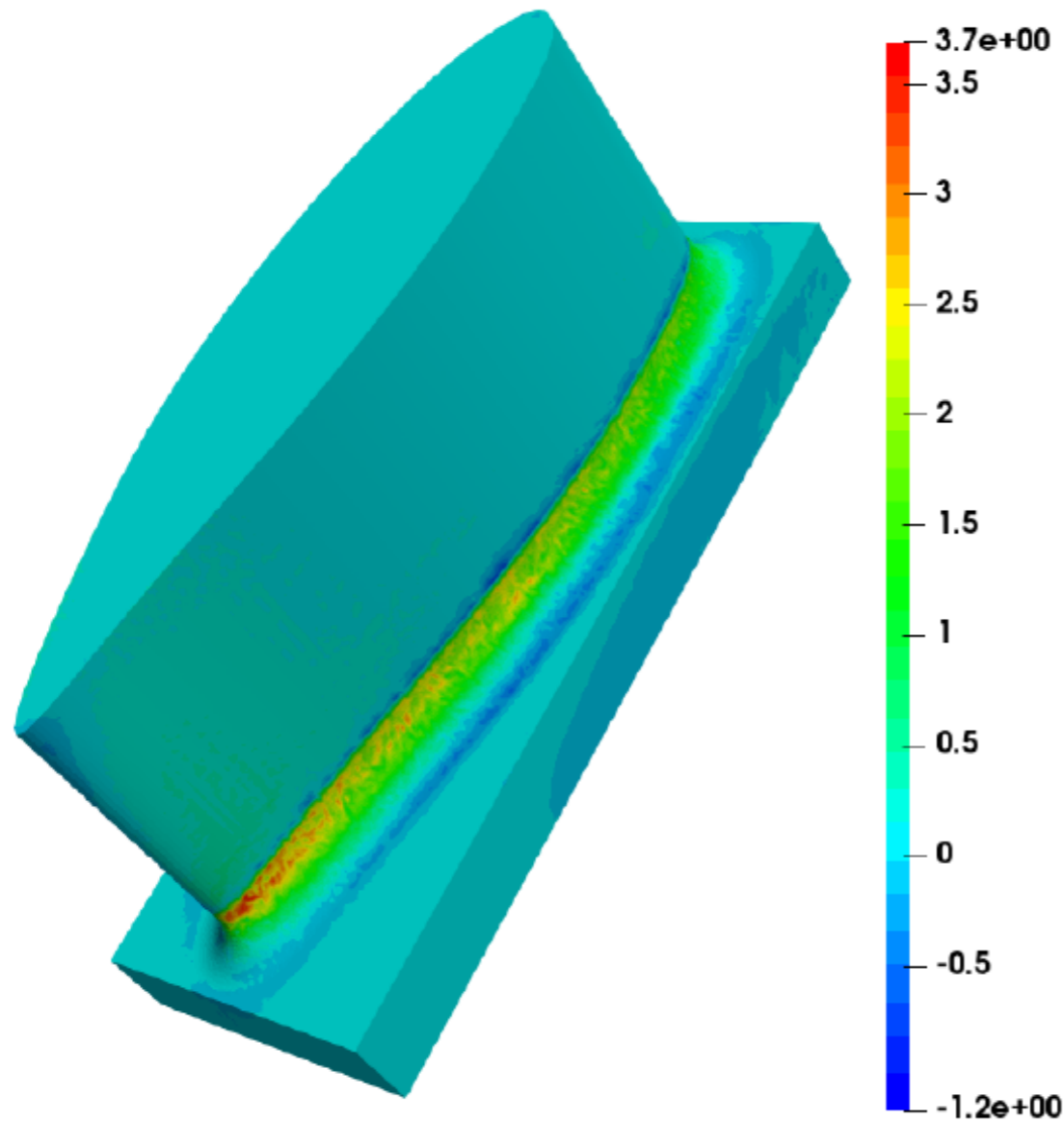
Displacement Loading

# 3D Results – Simplified Blade: Model and Boundary Conditions

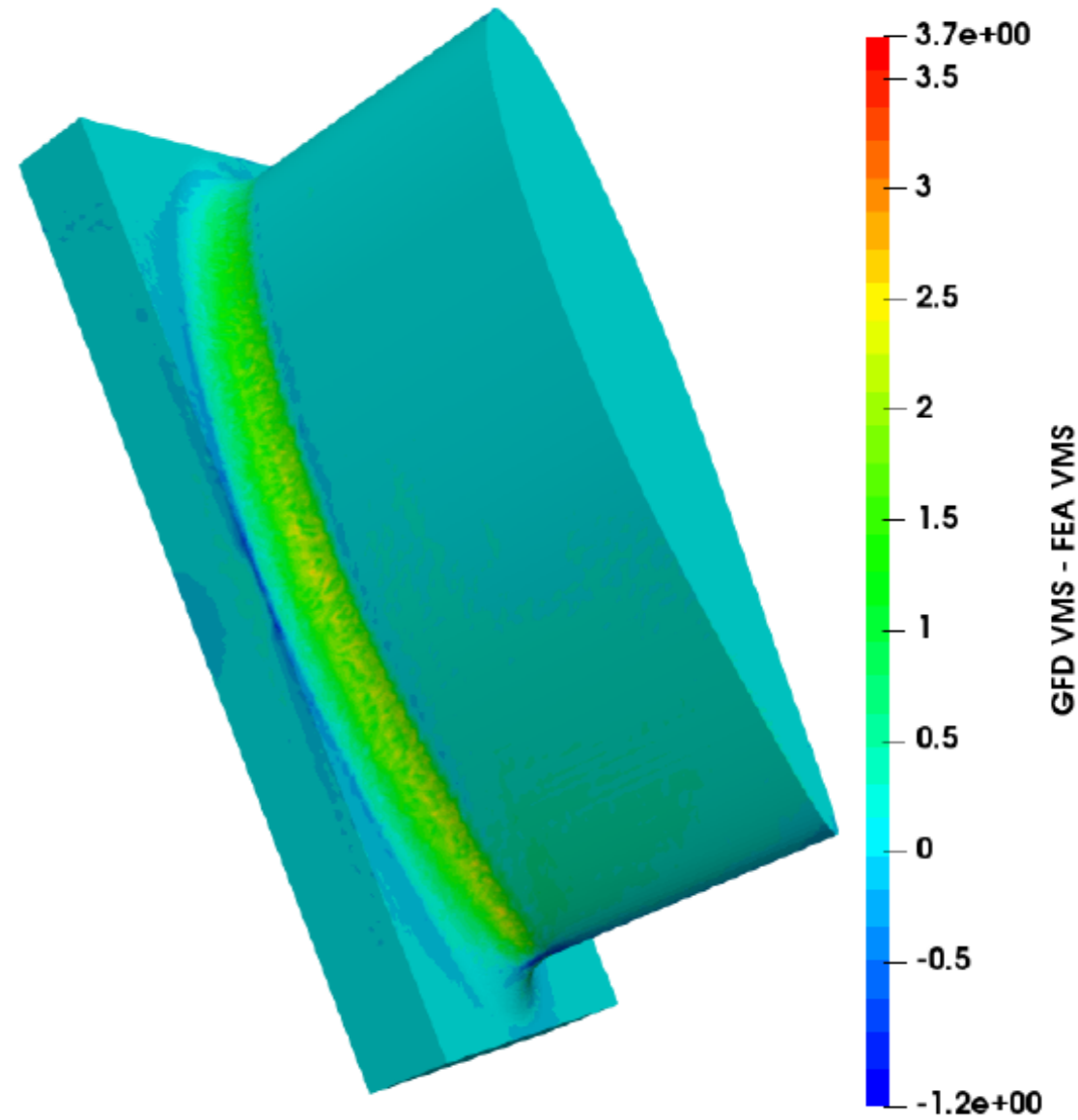


Surface(s)	Pressure Loading
XZ, YZ and XY Sym. Planes	Resp. constrained in the Y, X and Z directions
External Surface	Stress free
Pressurized Surface	Constant pressure

# 3D Results – Simplified Blade: Difference (GFD - FEA)



(a)



(b)

# Analysis of 3D Results

- Large number of nodes
  - 548,648 nodes for the ISO flange
  - 484,238 nodes for the blade
- The results obtained with GFD and FEA are very close
- The observed stress concentrations are larger with the collocation method

# Sensitivity to various parameters

- Both the GFD and the DCPSE methods are sensitive to the parameters on which they depend
- Some of the parameters are:
  - The weight function considered (e.g. spline, exponential...)
  - The correction function for DCPSE (e.g. polynomial, exponential)
  - The number of nodes considered in the inner nodes support
  - The number of nodes considered in the boundary nodes support



# Programs used

- CGAL: Node neighbour search and geometry of the boundary
- Voro++: Voronoi diagram for 2D and 3D problems
- Eigen: Matrix and vector classes for the assembly step
- OpenMP: Multithreading
- MPICH: Parallel multi-node computing
- PETSc: Matrix preconditioners (algebraic multigrid), parallel iterative solver (GMRES),
- MUMPS: MULTifrontal Massively Parallel Sparse direct solver

# Legato-team

University of Luxembourg